

Connecting System Architecture to Model-Based Design

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Application Engineer
MathWorks
Plano, TX

Agenda

- MathWorks Overview
- System Architecting and Model-Based Design
 - User Needs
 - Current workflows
 - MathWorks solution
 - Example
- Summary
- Resources

Background

- University of Illinois at Urbana-Champaign
 - B.S, M.S. Aerospace Engineering
- SpaceX Rocket Development Facility
 - Test Engineer
 - Lead Engineer, Integration & Test
- Eaton Aerospace, Fuel and Motion Controls
 - Lead Aerospace Systems Engineer
- MathWorks, Application Engineering Group
 - Lead Engineer, Aerospace applications



MathWorks at a Glance

- Privately held
- 4500 employees worldwide
- More than 4 million users in 185 countries

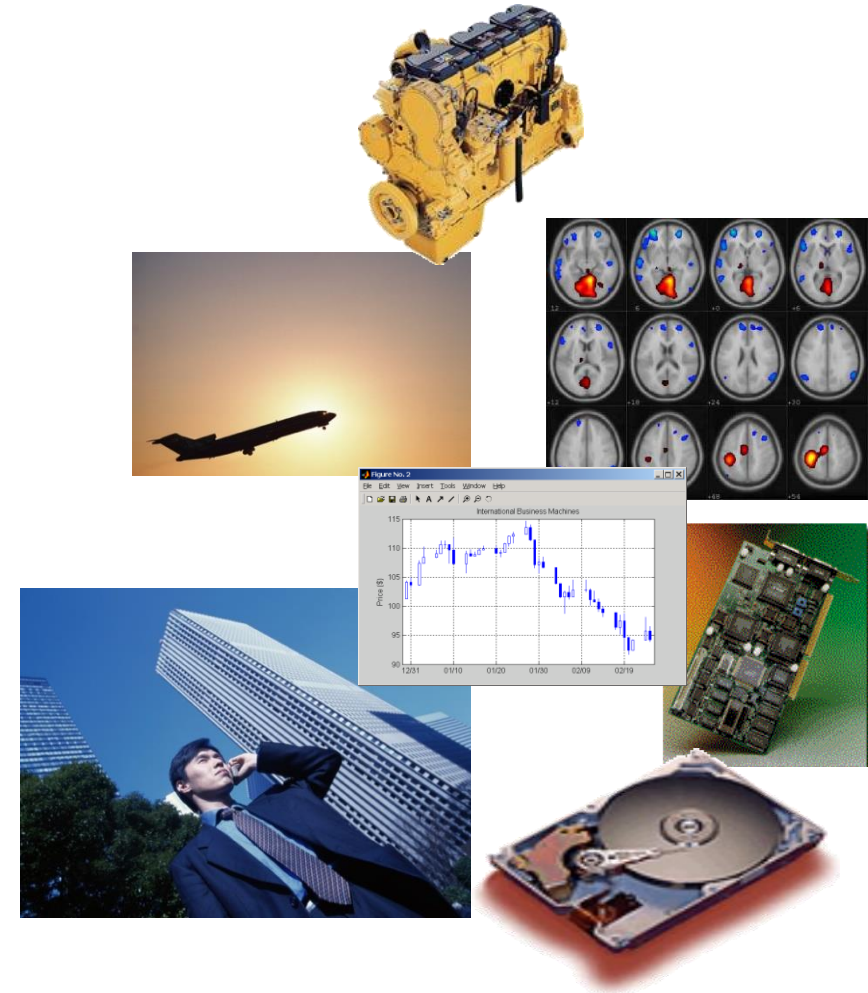


● Office locations

● Distributors serving 16 countries

Key Industries

- Aerospace and defense
- Automotive
- Biological sciences
- Biotech and pharmaceutical
- Communications
- Electronics
- Energy production
- Financial services
- Industrial automation and machinery
- Medical devices
- Metals, materials, and mining
- Neuroscience
- Railway systems
- Semiconductors
- Software and internet

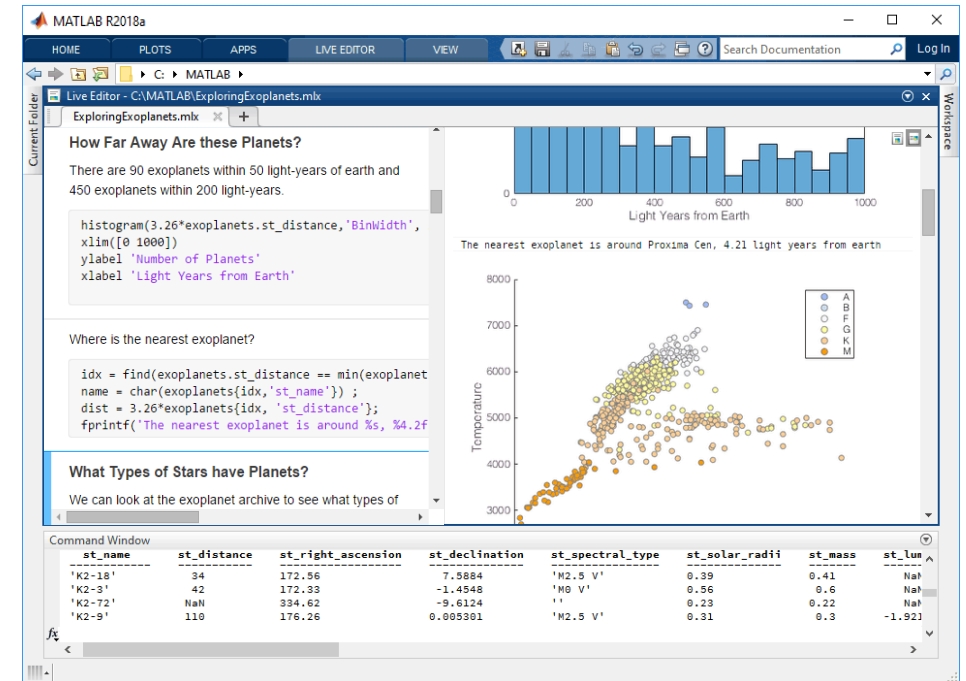


Core MathWorks Products

MATLAB®

Math. Graphics. Programming.

- Designed for engineers and scientists
- Professionally developed, tested, and documented
- Toolboxes for:
 - Machine learning, data analytics, deep learning, image processing and computer vision, signal processing and communications, computational finance, robotics and control systems
- Interactive apps that automatically generate programs
- Easily scales to clusters, GPUs, and clouds
- Direct deployment to production enterprise applications
- Automatic conversion to embeddable C and CUDA code
- Integrates with Simulink to support Model-Based Design



Core MathWorks Products

SIMULINK®

Simulation and Model-Based Design

Model and simulate your system

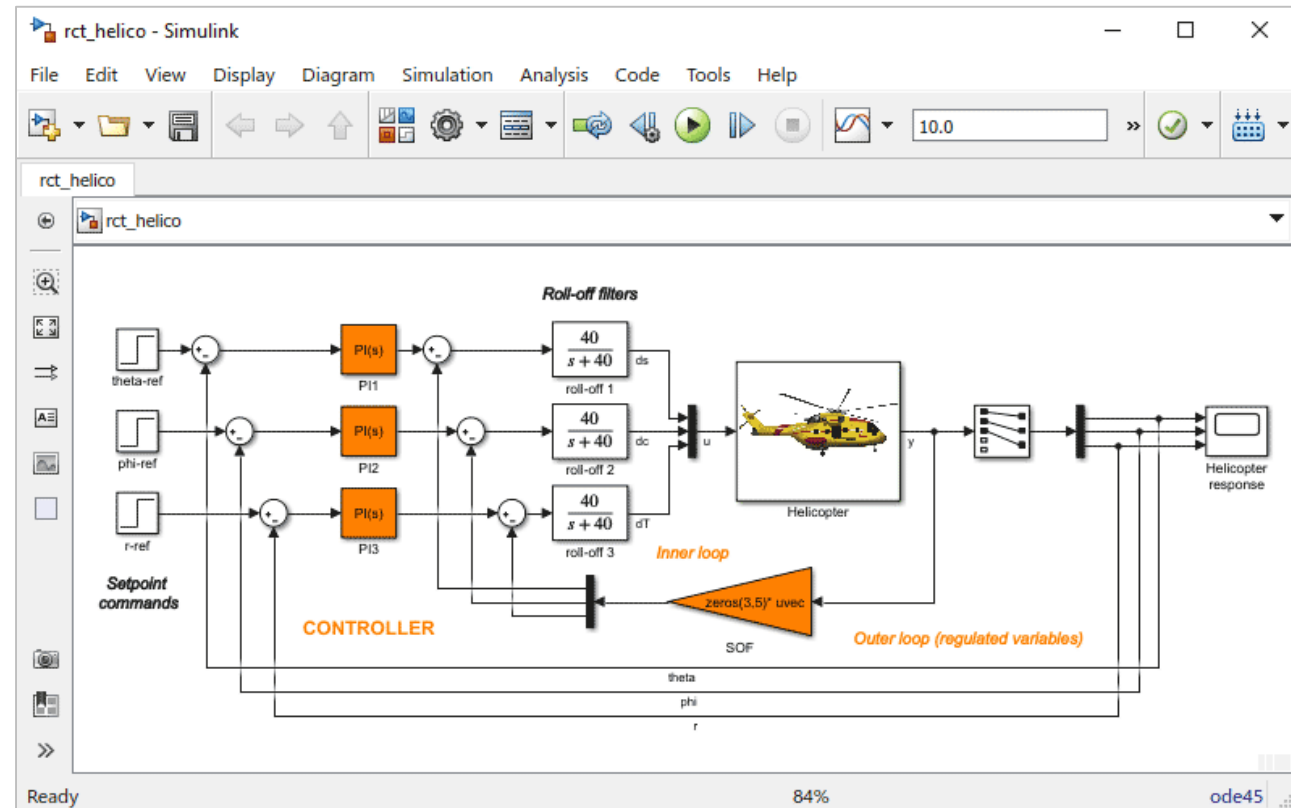
- Use one multi-domain environment
- Model the system under test and the plant
- Simulate how all parts of the system behave

Test early and often

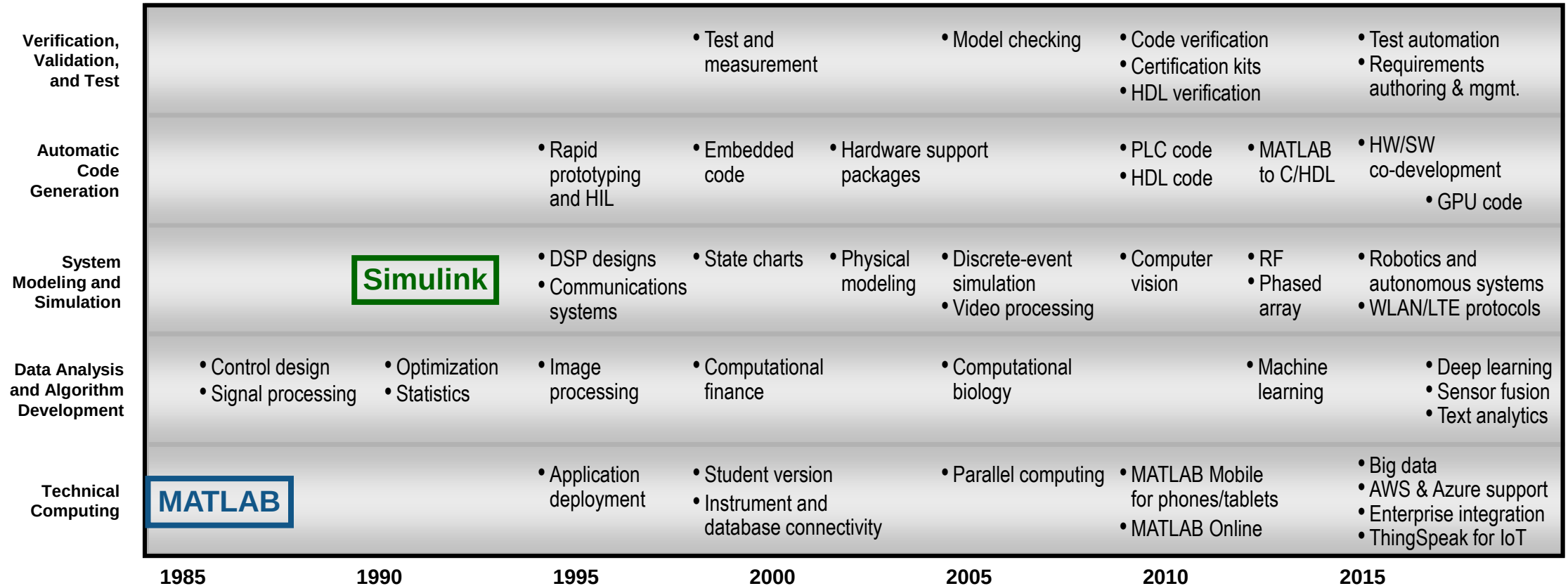
- Test your system under all conditions
- Validate your design with real-time testing
- Trace from requirements to design to code

Automatically generate code

- Generate production-quality C and HDL code
- Deploy directly to embedded processors or FPGA's/ASIC's

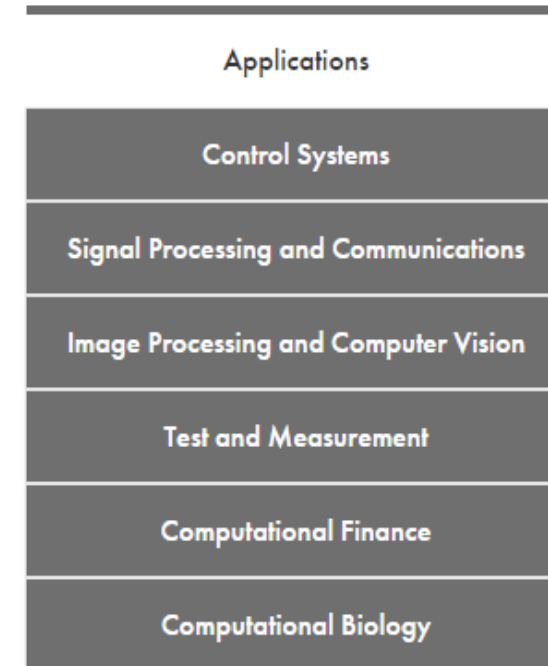
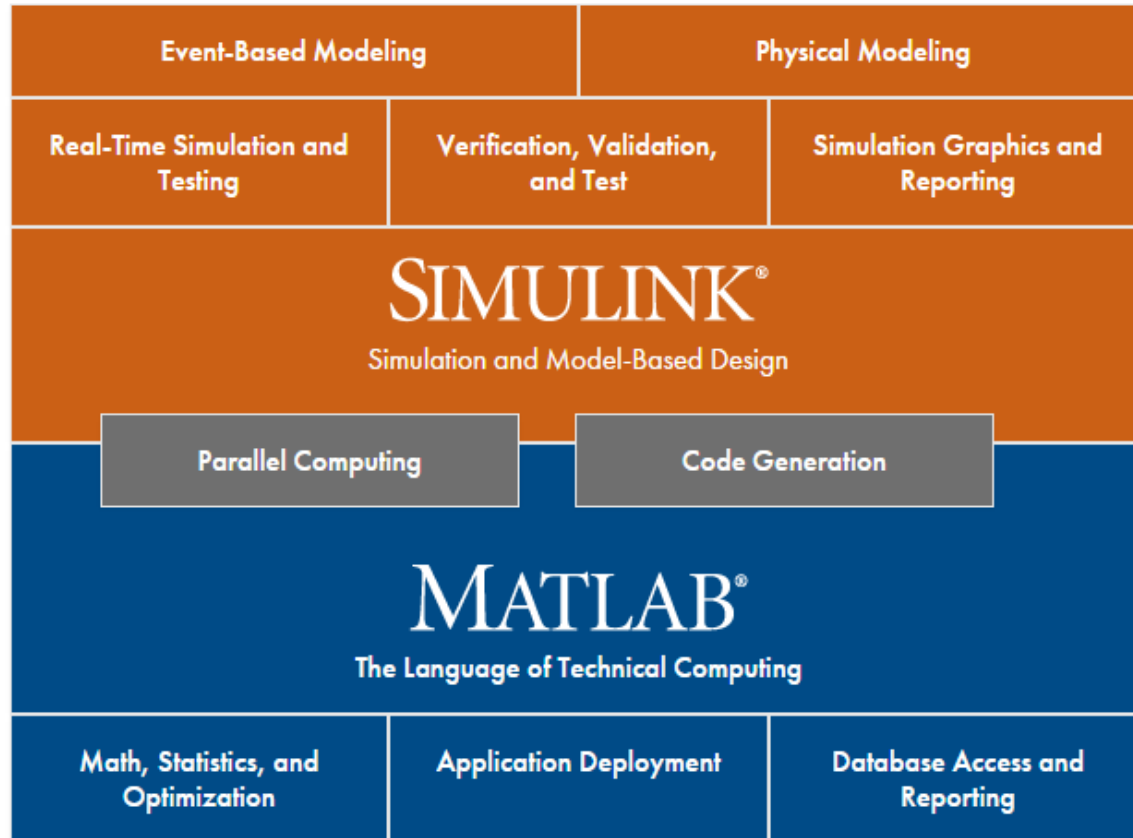


Key capabilities for engineers and scientists

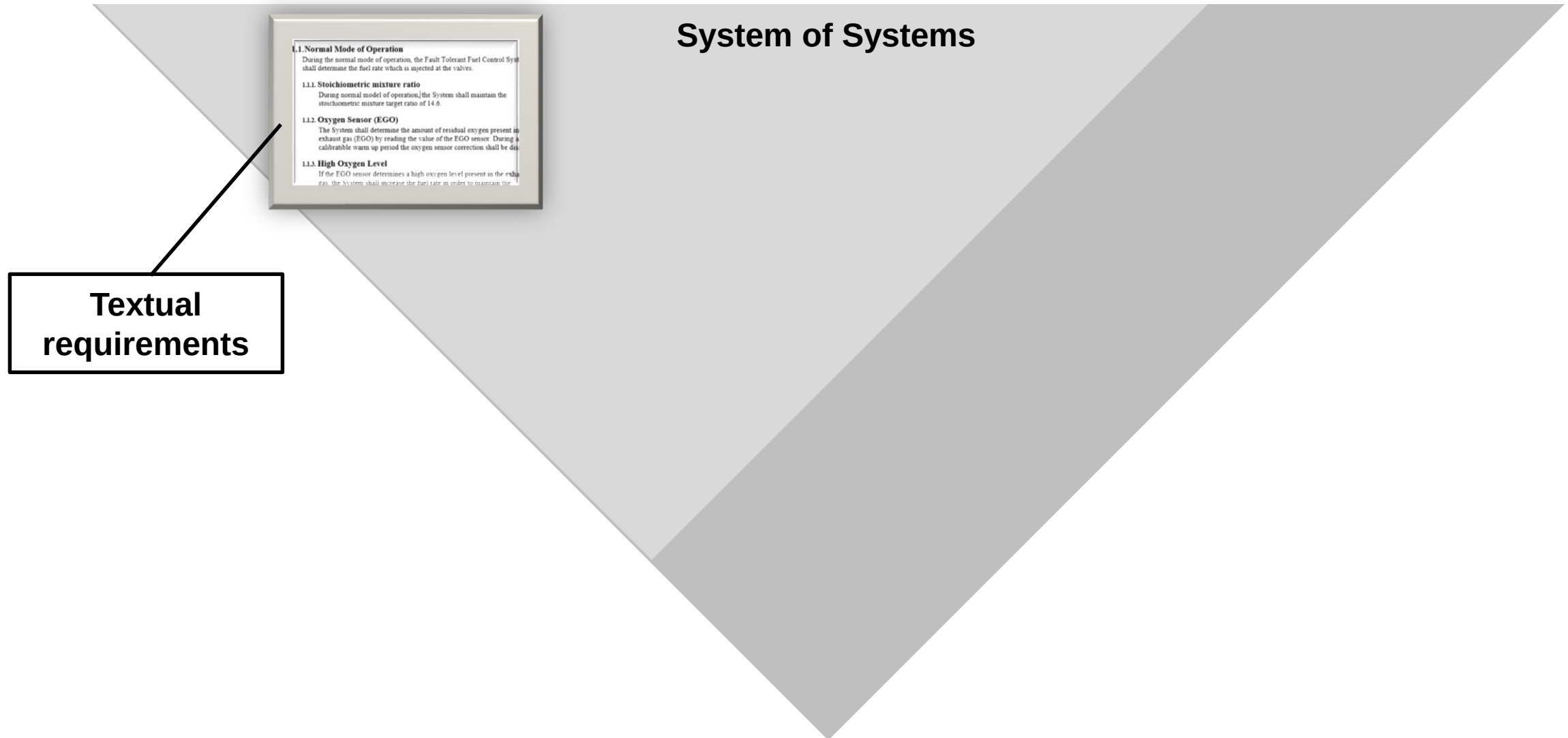


MathWorks
founded
in 1984

MathWorks Product Overview

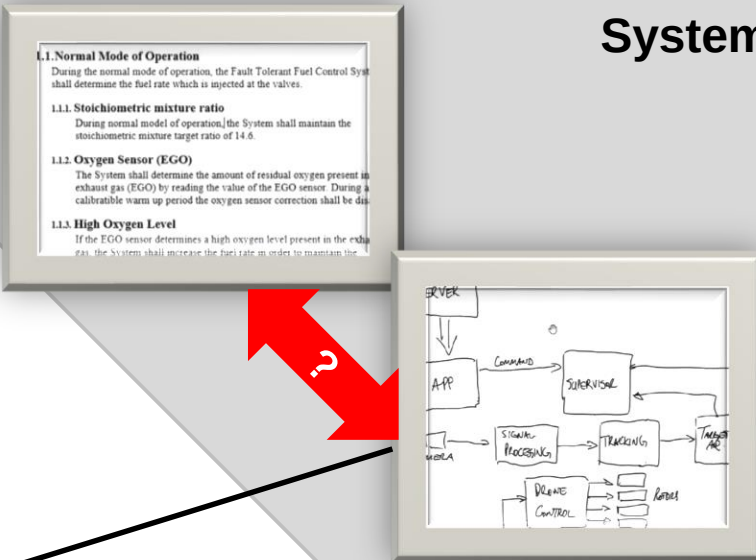


How is this done today?



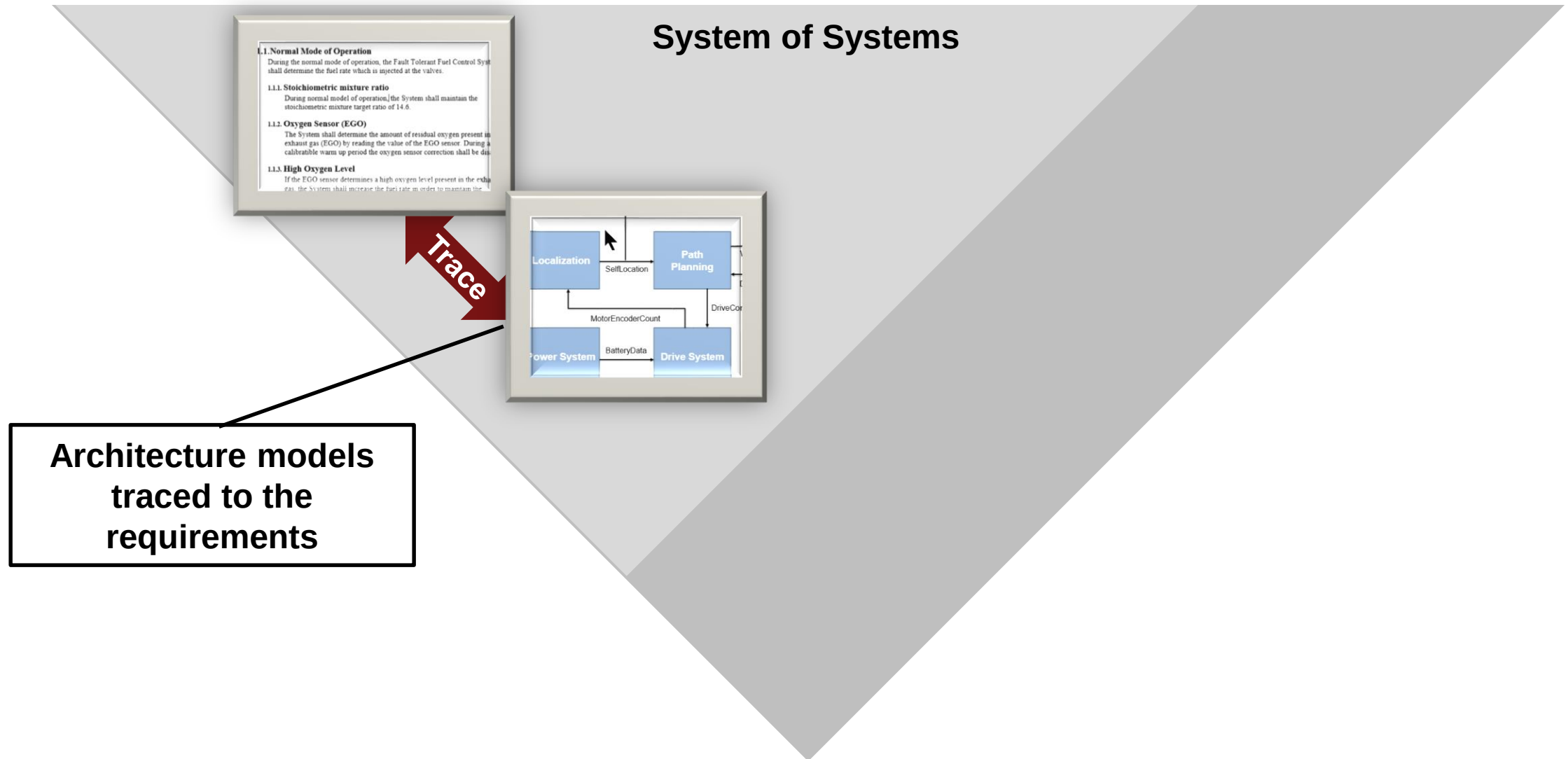
How is this done today?

System of Systems

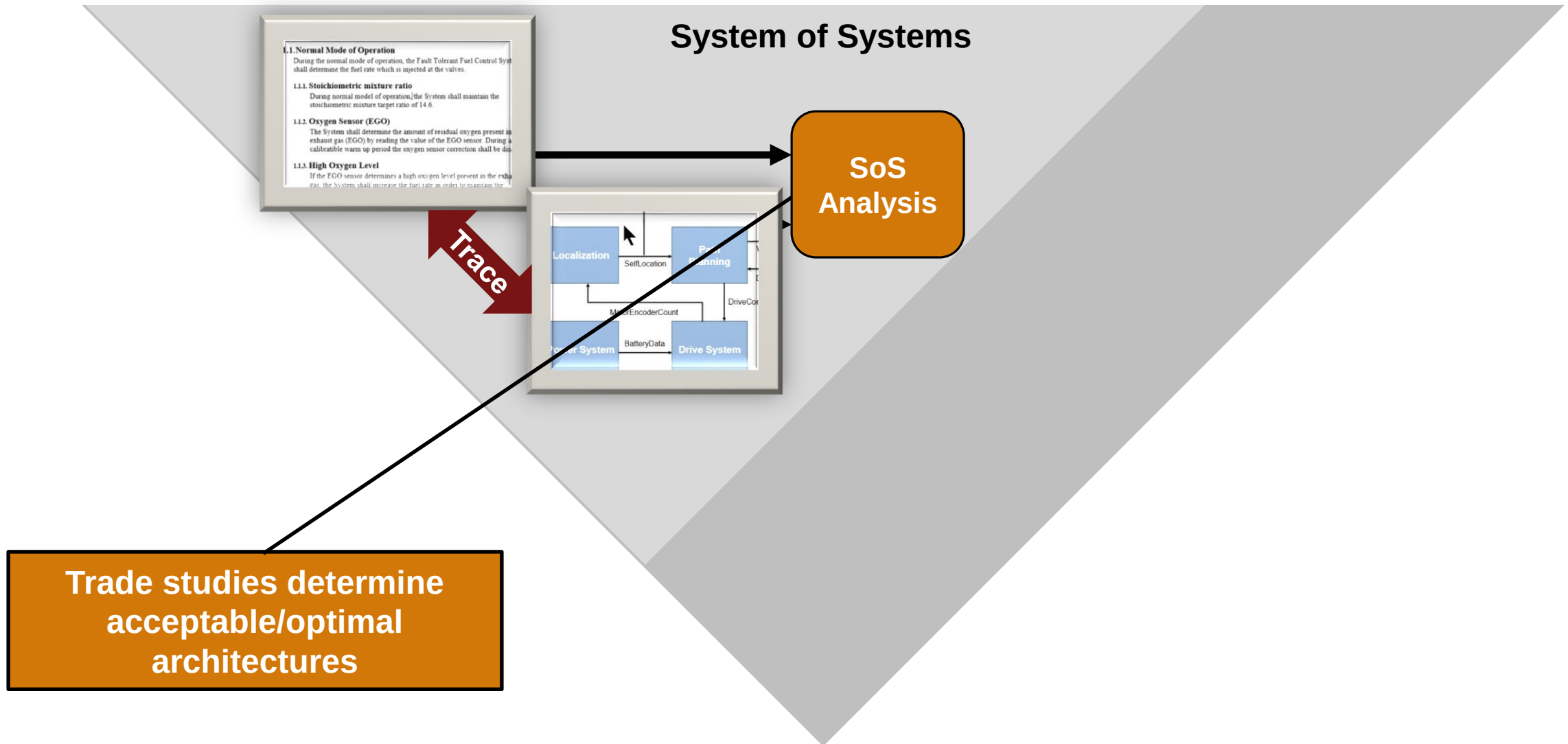


Architecture sketches are difficult/impossible to trace to requirements

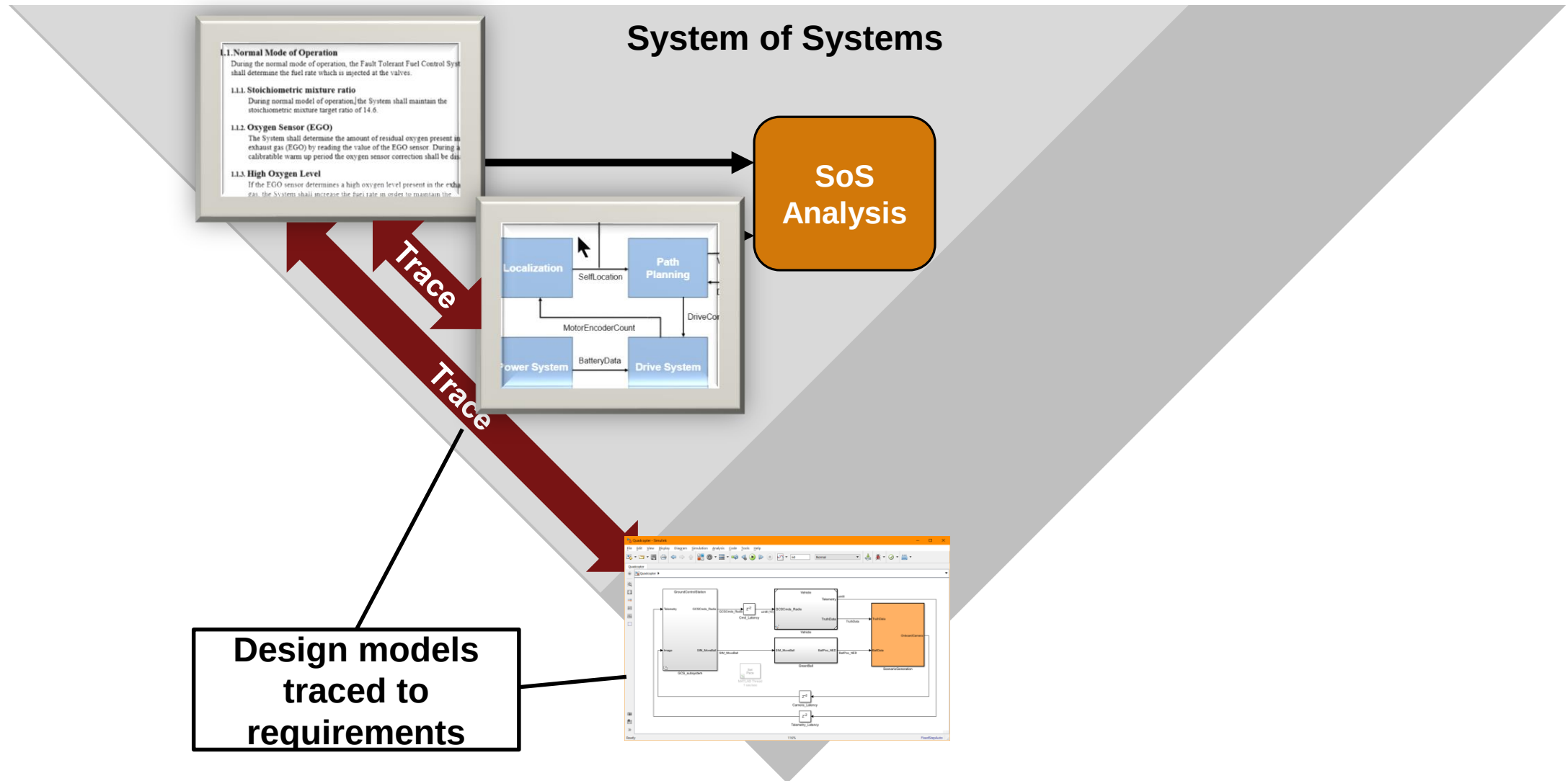
How is this done today?



How is this done today?

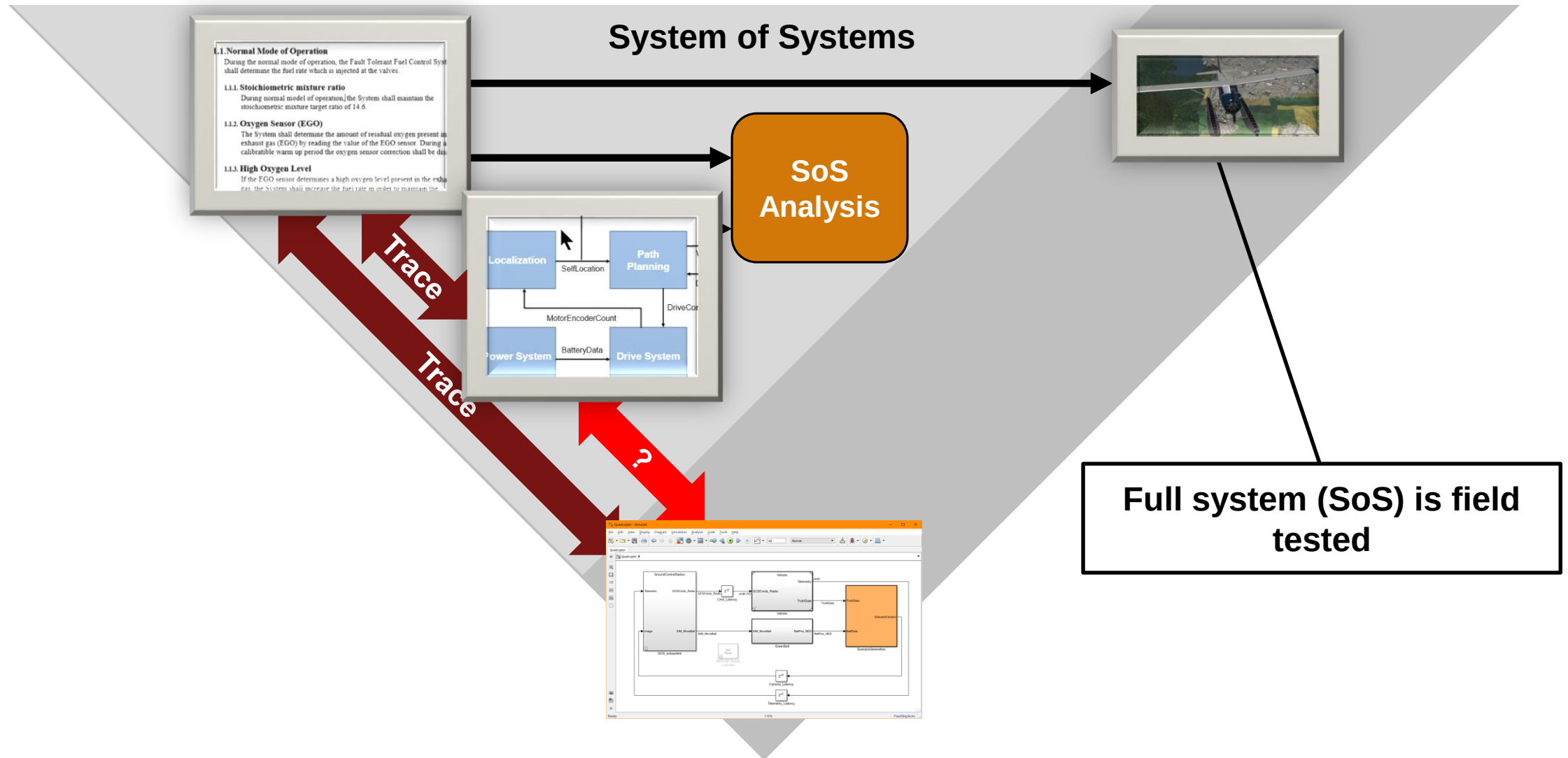


How is this done today?



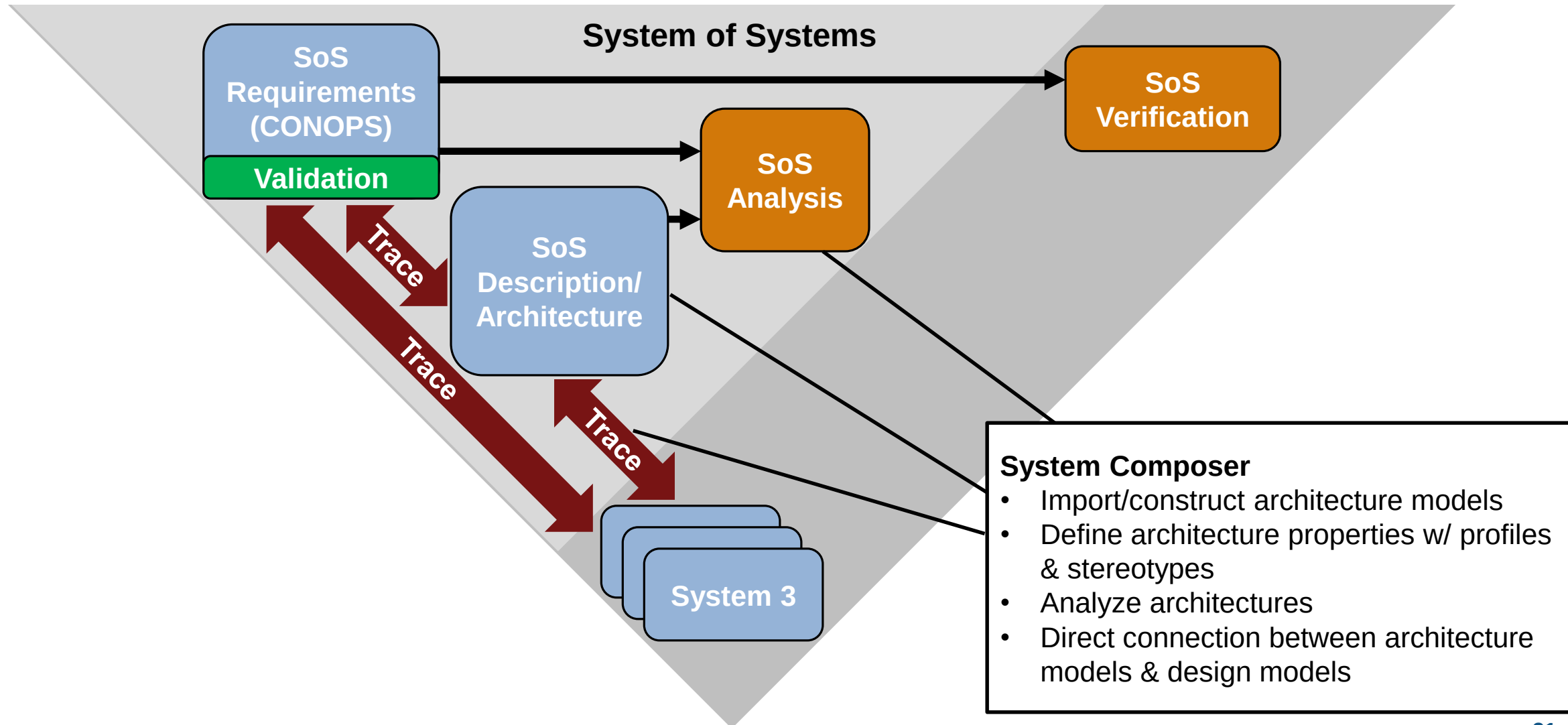


How is this done today?

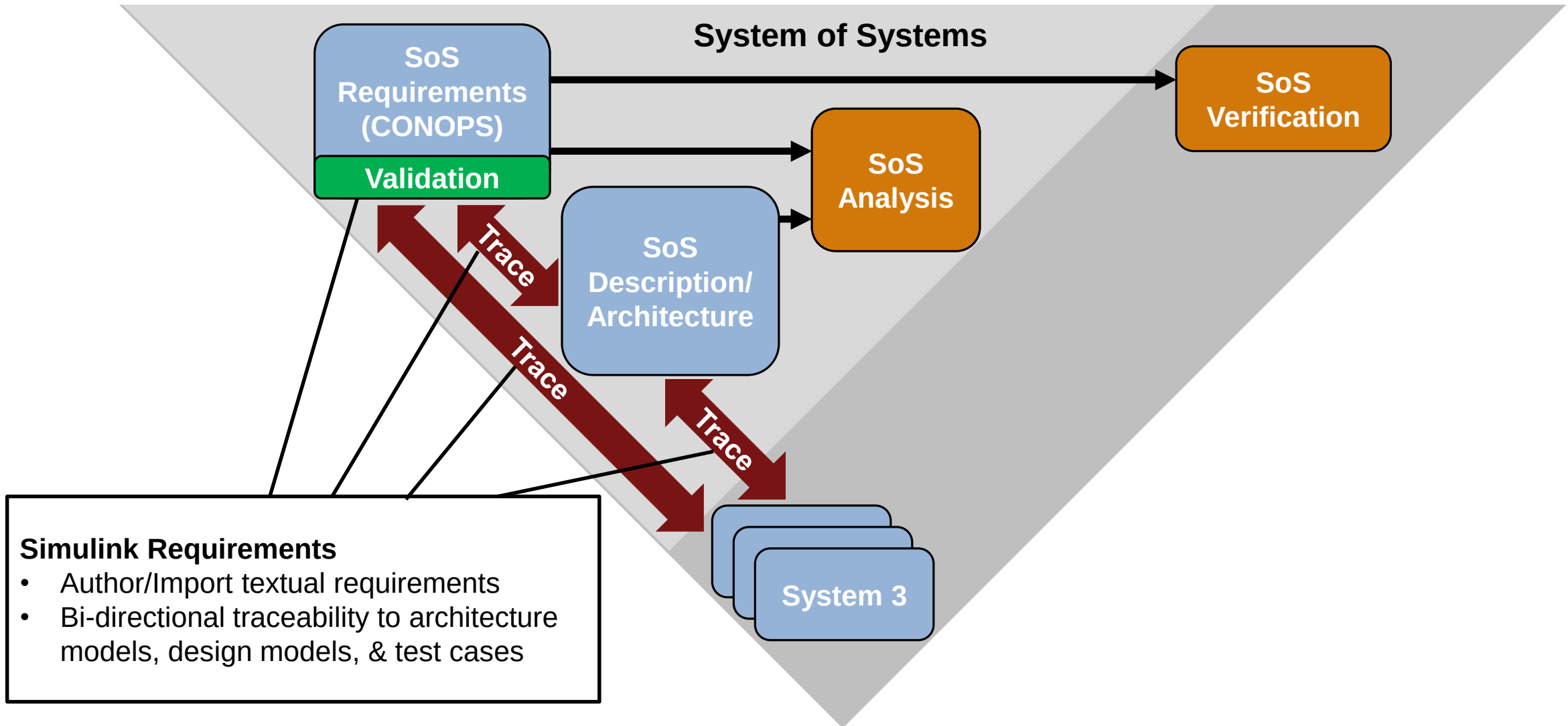


How can MathWorks address the gap?

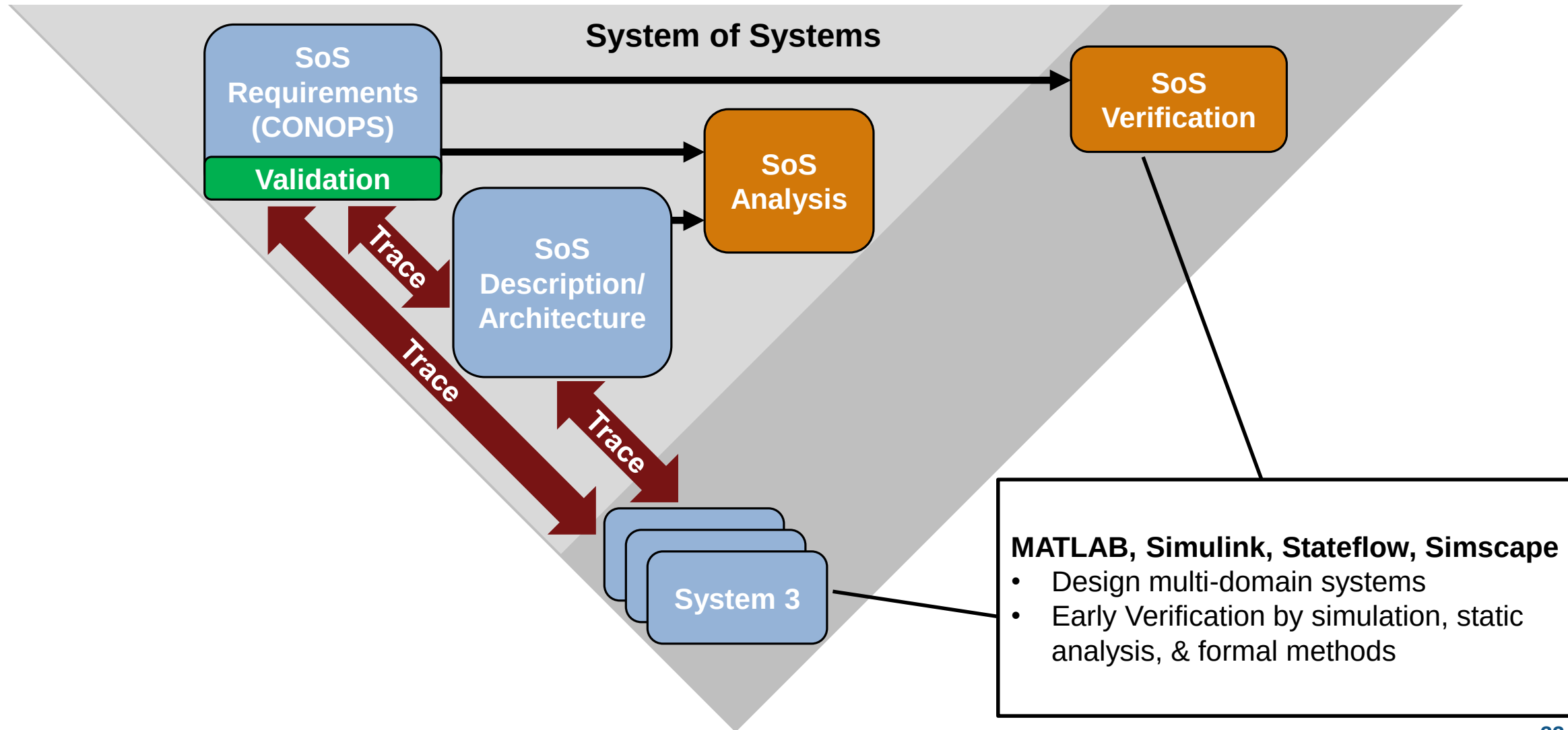
System Architecting & Technical Analysis



Requirements Management



Verification & Validation



Now let's see it in action



Requirements Editor

File Edit Display Analysis Report Help

View: Requirements Search

Index	Summary
UAS_reqs*	
1	Aircraft Capabilities
1.1	Airworthiness
1.2	Communications
1.3	Payload Capabilities
1.4	Construction
1.4.1	Modularity
1.4.2	Propulsion Power
1.5	Flying Qualities
2	Ground Station Capabilities
3	BLOS Capabilities

Properties

Type: Functional

Index: 1.4.2

Custom ID: #35

Summary: Propulsion Power

Description Rationale

Gas Engine: Nine-cylinder, air-cooled, radial aircraft engine
Fuel type: 80/87 grade aviation gasoline
Dry weight: 290 kg
Power output:
400 hp (298 kW) at 2,200 RPM up to 5,000 ft (1,500 m);
450 hp (336 kW) at 2,300 RPM for takeoff

Keywords:

Revision information:

Links

No links

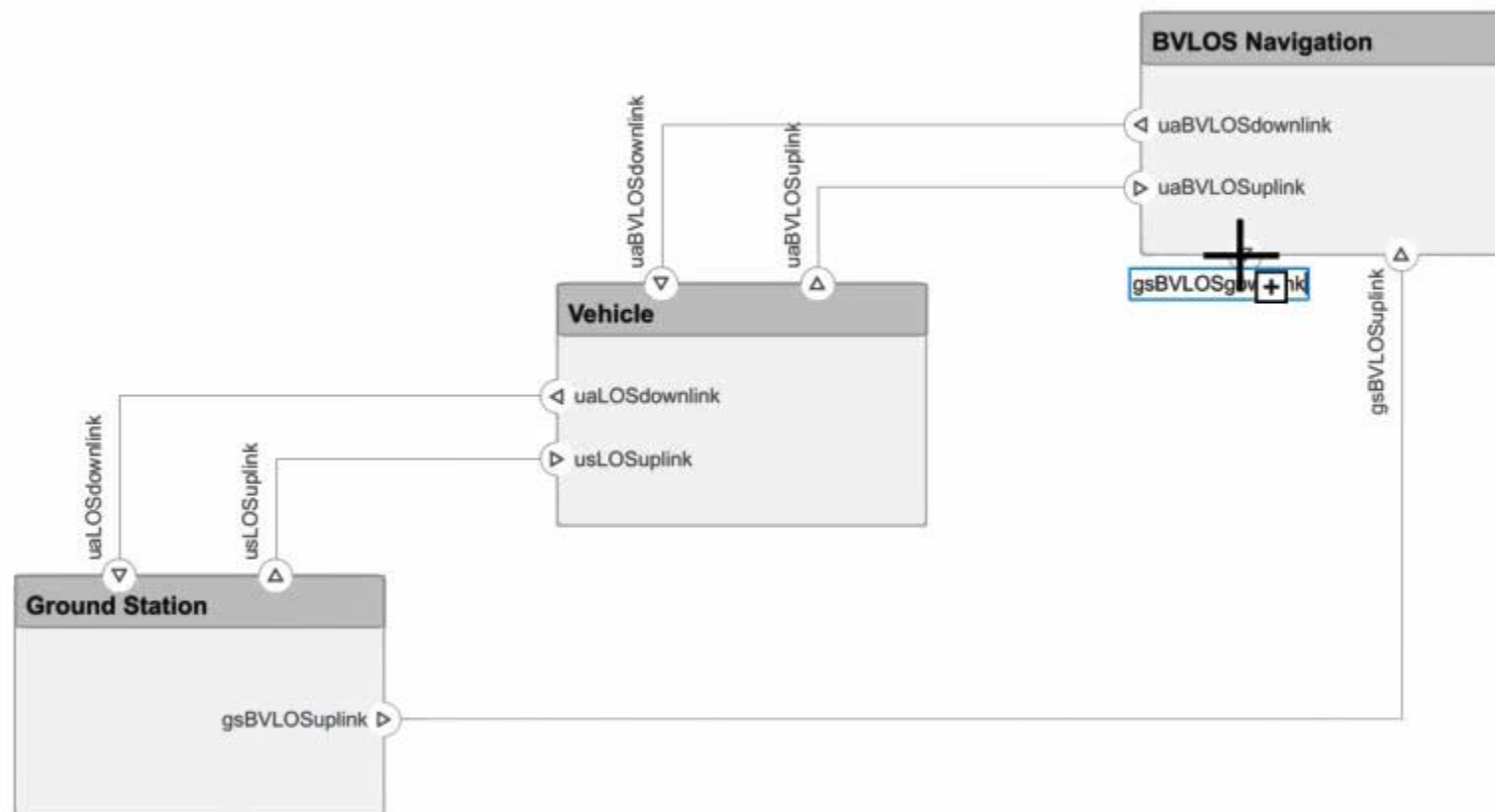
Comments



UAS_ref_arch

UAS_ref_arch ▶

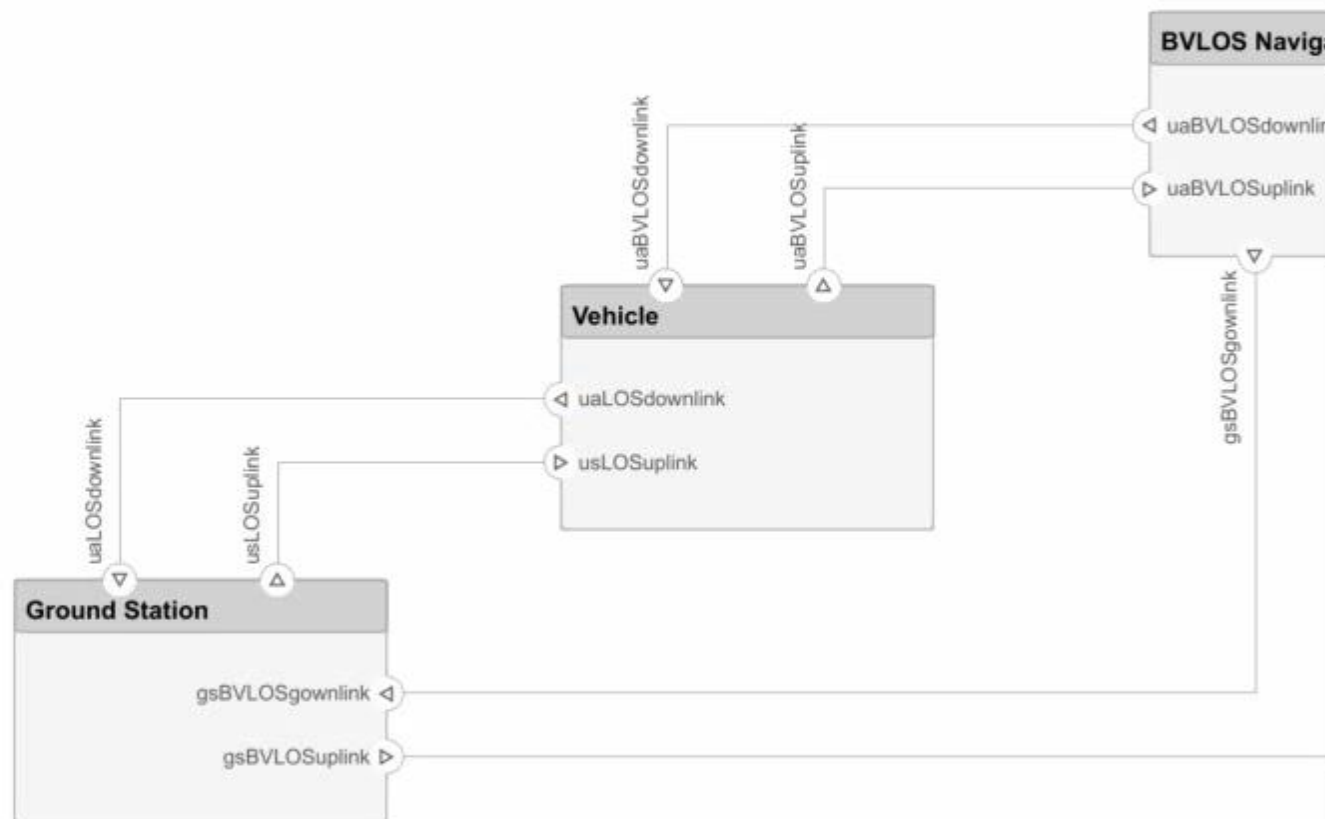
UAS_ref_arch





UAS_ref_arch

UAS_ref_arch ▶



Interfaces



Source: UAS_ref_arch.slx

Telemetry

Altitude

Climb rate

Ground speed

Heading

Latitude

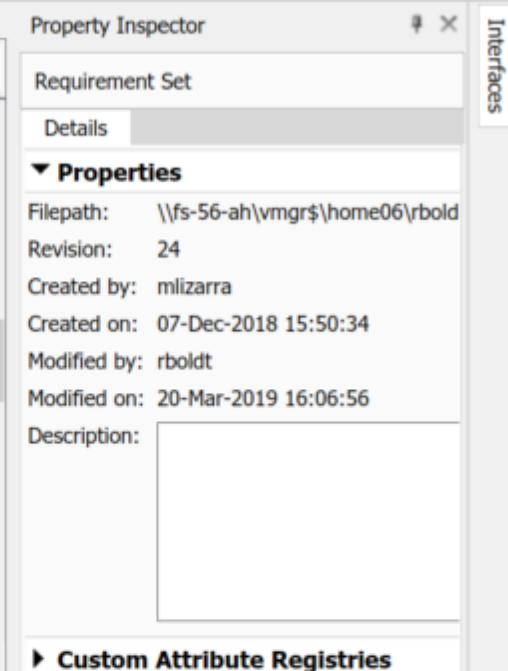
Longitude

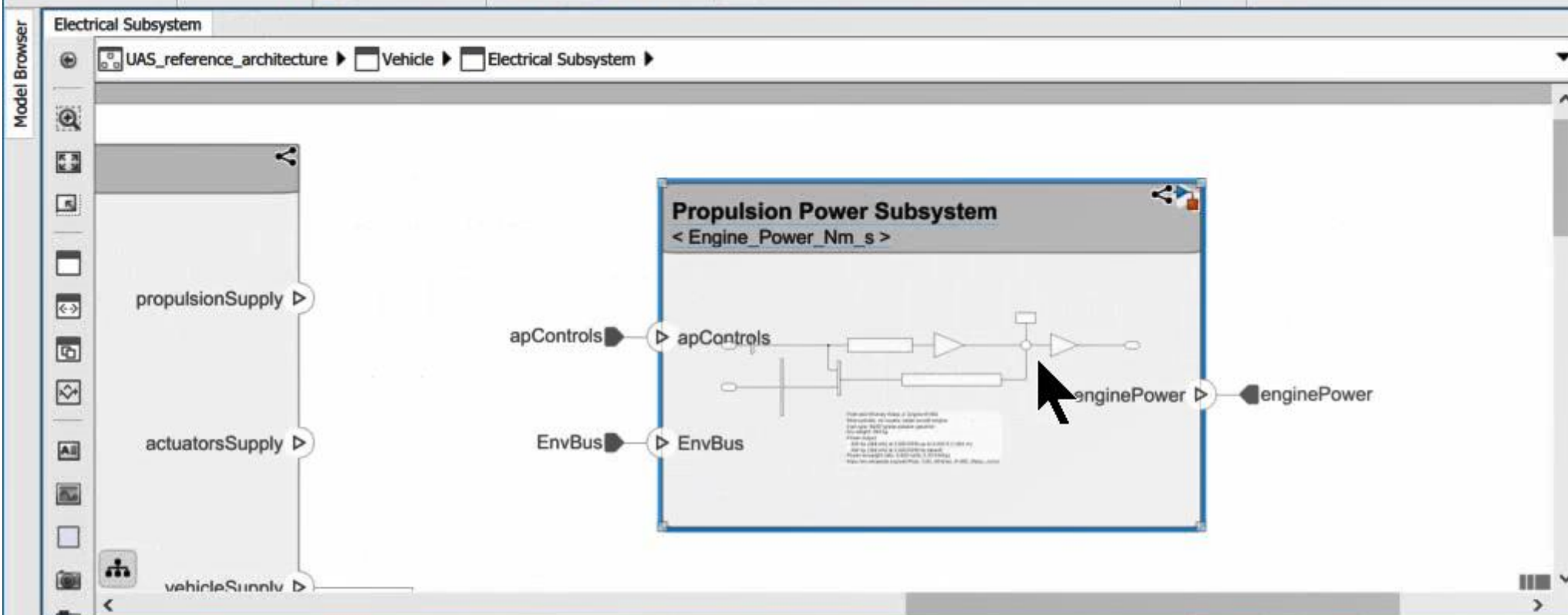
Property Inspector

Interface : Telemetry | Element : Altitude

Properties

NAME	VALUE
Type	double
Dimensions	1
Units	feet
Complexity	real
Minimum	0
Maximum	30000
Description	Altitude of plane

This image shows a blank white page from a document or presentation. At the very bottom, there are two small, dark gray navigation icons: a left-pointing arrow (<) on the left and a right-pointing arrow (>) on the right, separated by a thin vertical line. The rest of the page is completely empty and white.



Property Inspector

Component

Architecture	Info
NAME	VALUE
Main	
Name	Propulsion Power ...
Stereotype	Add..
SubsystemBud...	Select

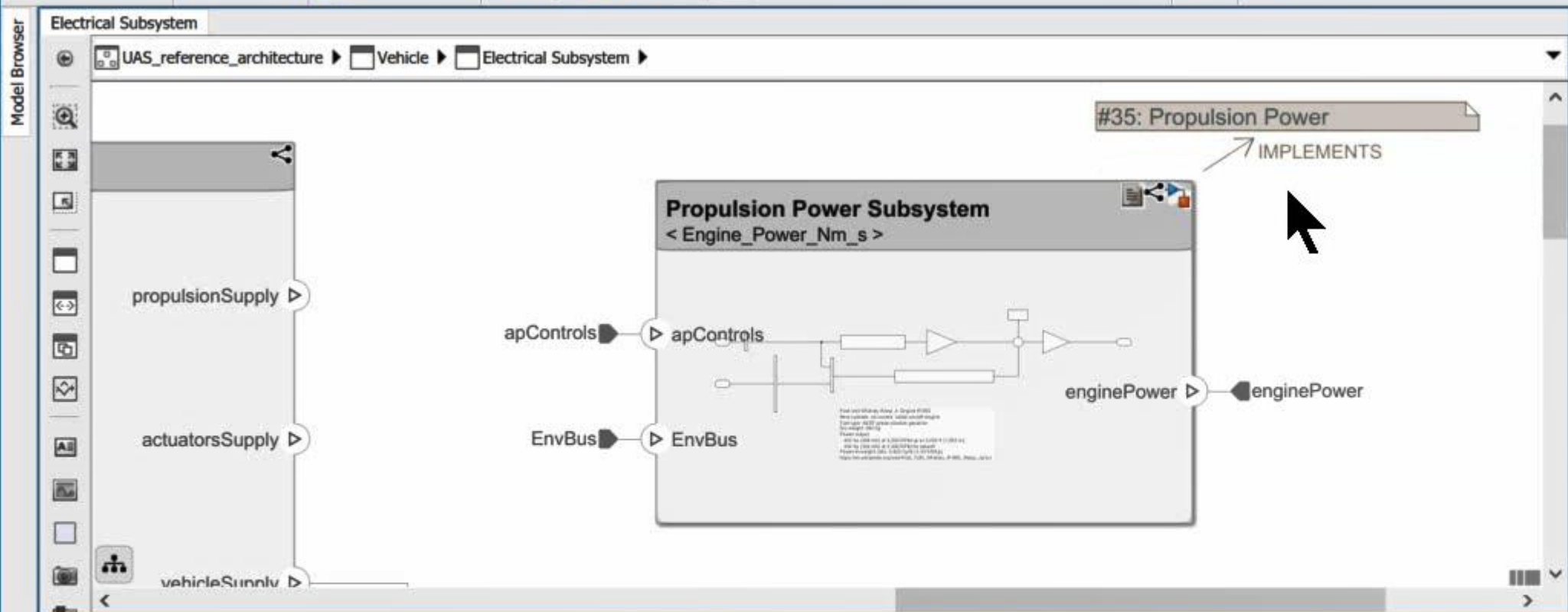
Requirements - UAS_reference_architecture

View: Requirements



Search

Index	Summary	Implemented
> 1.2	Communications	
> 1.3	Payload Capabilities	
> 1.4	Construction	
1.4.1	Modularity	
1.4.2	Propulsion Power	
> 1.5	Flying Qualities	



Property Inspector

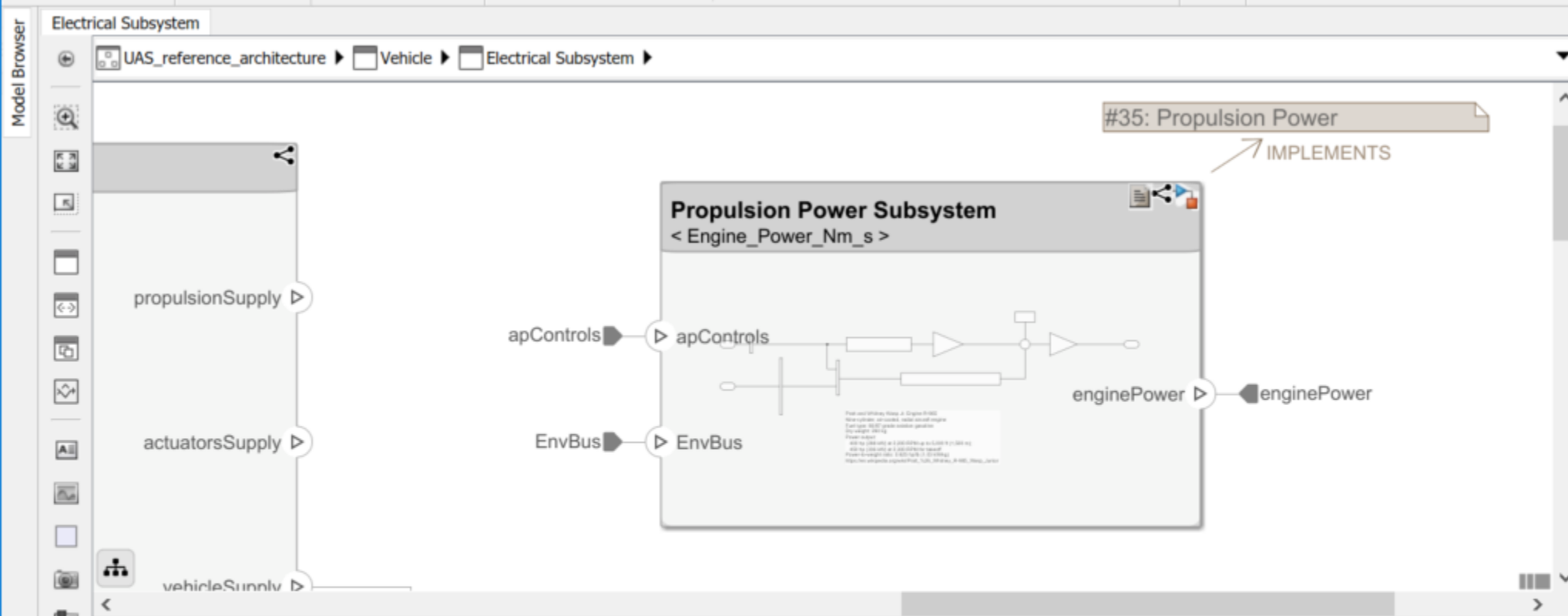
Component

Architecture	Info
NAME	VALUE
▼ Main	
Name	Electrical Subsystem
Stereotype	Add..

Requirements - UAS_reference_architecture

View: Requirements

Index	Summary	Implemented
UAS_reqs		
1	Aircraft Capabilities	
1.1	Airworthiness	
1.1.1	Range	
1.1.2	Rain Conditions	
1.1.3	Power	



Property Inspector

Requirement Set

Details

Properties

Filepath: \\fs-56-ah\vmgr\$\home06\rbold
Revision: 24
Created by: mlizarra
Created on: 07-Dec-2018 15:50:34
Modified by: rboldt
Modified on: 20-Mar-2019 16:06:56
Description:

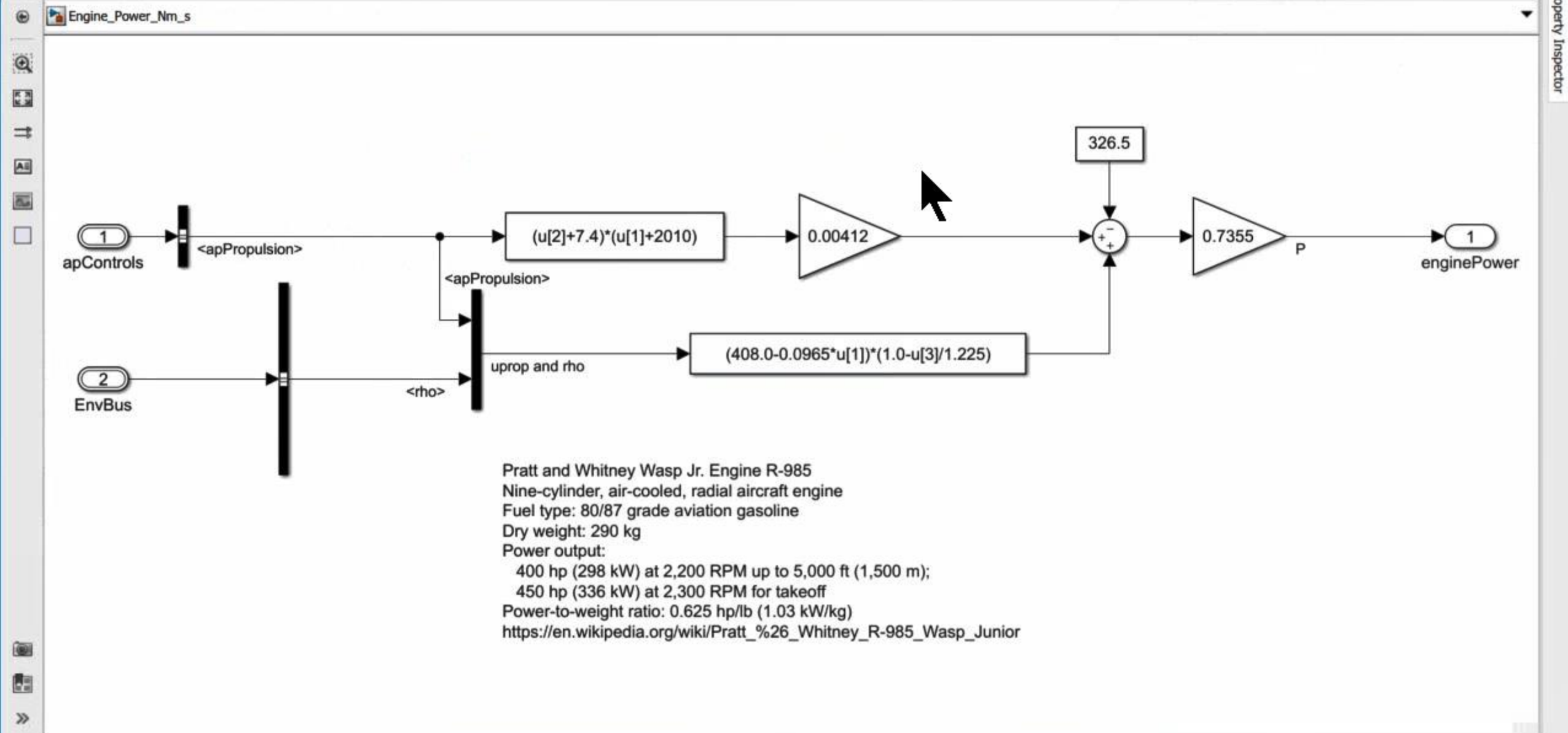
Custom Attribute Registries

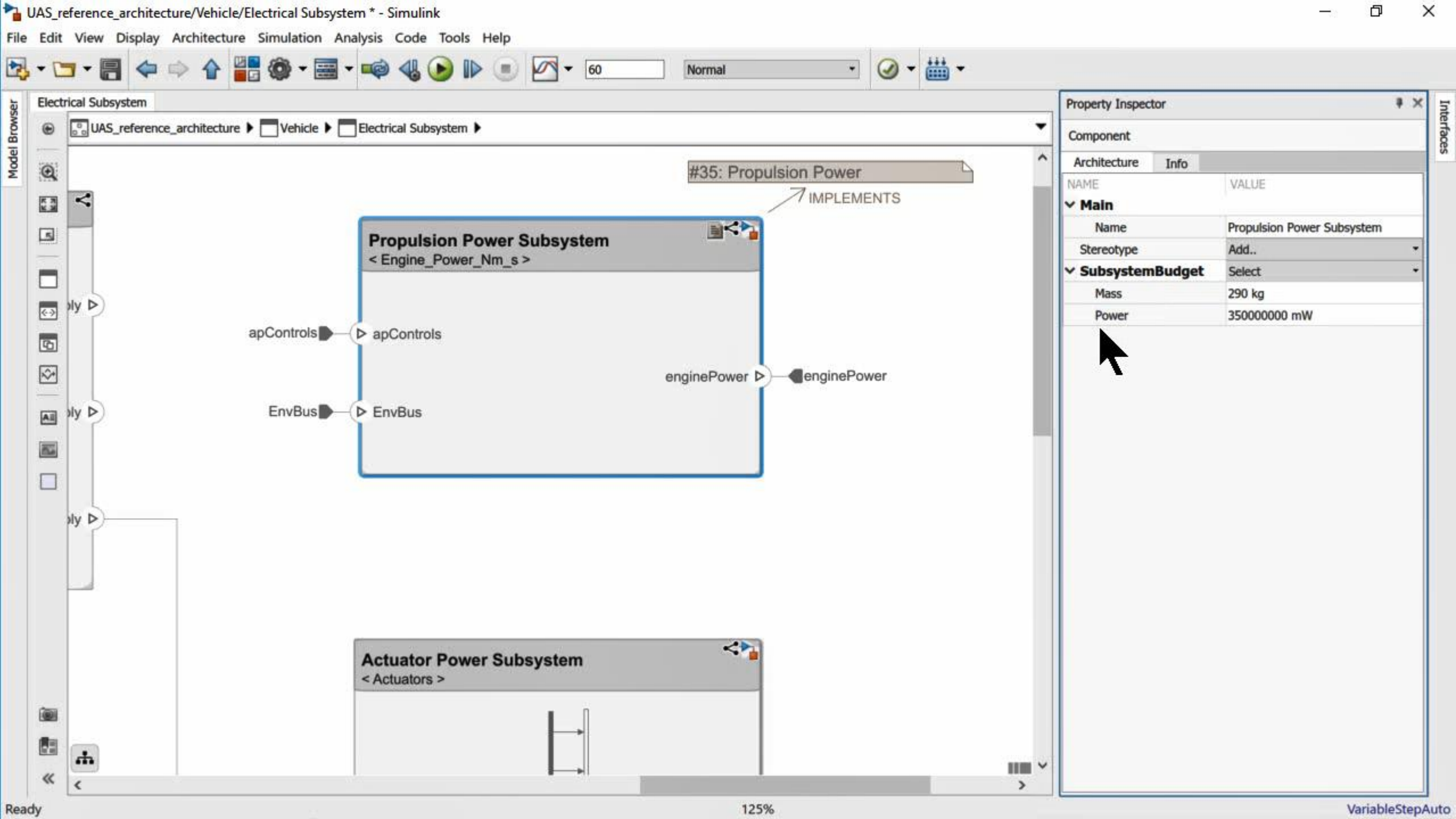
Requirements - UAS_reference_architecture

View: Requirements

Index	Summary	Implemented
UAS_reqs		
1	Aircraft Capabilities	Implemented: 27, Justified: 0, None: 0, Total: 27
1.1	Airworthiness	
1.1.1	Range	
1.1.2	Rain Conditions	
1.1.3	Power	

Engine_Power_Nm_s





UAS_reference_architecture/Vehicle - Simulink

File Edit View Display Architecture Simulation Analysis Code Tools Help

Model Browser: UAS_reference_architecture > Vehicle

Vehicle Subsystem Diagram:

- Environment Subsystem
- Flight Visualization
- Communications Subsystem
- Payload or Cargo
 - UAS Capability Levels: Aircraft shall be able to carry up to 500 kg of payload
 - UAS Payload Capacity: Aircraft shall provide a payload volume with at least 8 m³
- Electrical Subsystem
 - UAS Power: Aircraft shall provide electrical power to the payload and avionics
- Vehicle Airframe and Aerodynamics
 - UAS Flight Modes:
 - UAS1: Phugoid Mode (longitudinal) flying qualities shall meet at least Level 2 criteria from the MIL-F-8785C standard
 - UAS2: Short Period Mode (longitudinal) flying qualities shall meet Level 1 criteria from the MIL-F-8785C standard
 - UAS3: Dutch Roll Mode (lateral-directional) flying qualities shall meet Level 1 criteria from the MIL-F-8785C standard
 - UAS4: Roll Mode (lateral-directional) flying qualities shall meet Level 1 criteria from the MIL-F-8785C standard
 - UAS5: Spiral Mode (lateral-directional) flying qualities shall meet Level 1 criteria from the MIL-F-8785C standard

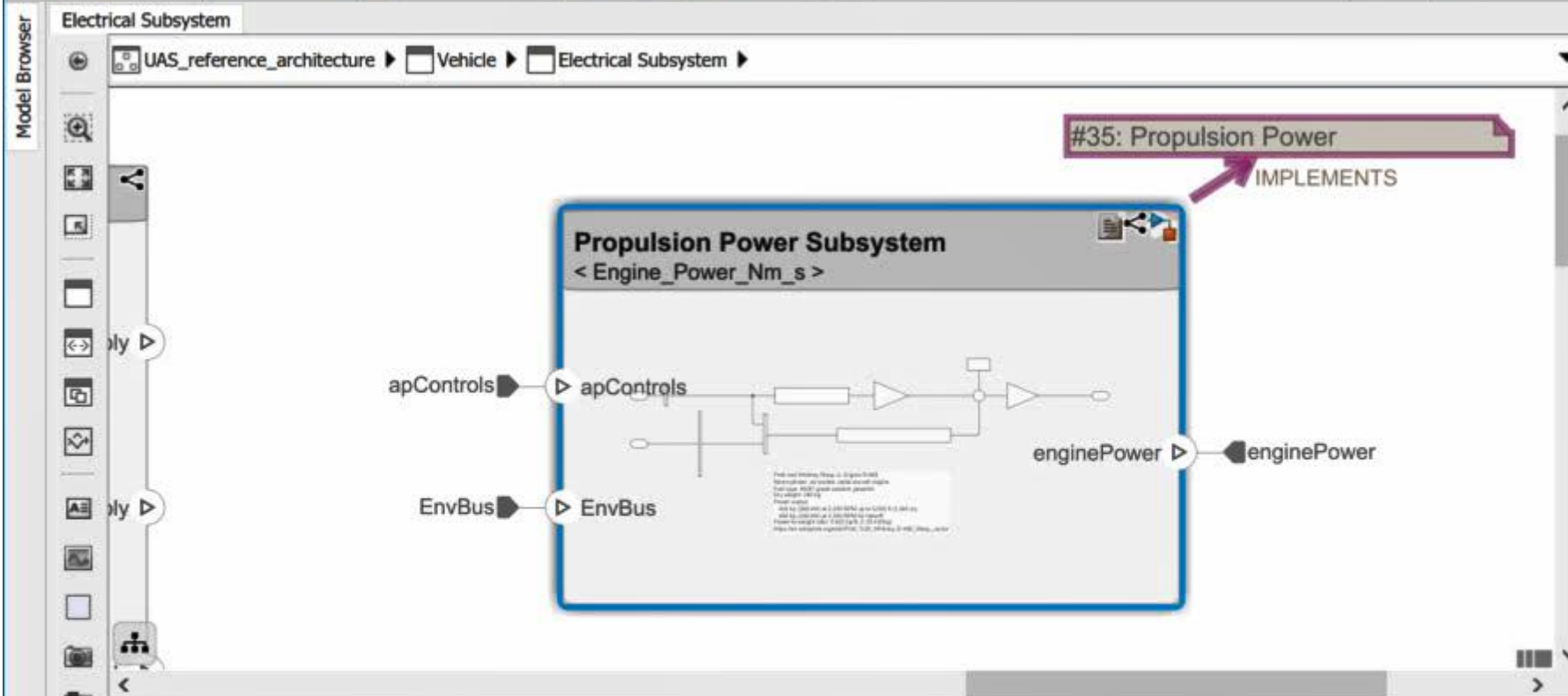
FlightGear Window:

File View Location Autopilot Environment Equipment AI Multiplayer Debug Help Beaver



Running

50% T=15.600 26% auto(ode23t)



Requirements - UAS_reference_architecture

View: Requirements

Index	Summary	Implemented
1.4	Construction	
1.4.1	Modularity	
1.4.2	Propulsion Power	
1.5	Flying Qualities	
2	Ground Station Capabilities	

Property Inspector

Requirement: #35

Details

▼ Properties

Type: Functional

Index: 1.4.2

Custom ID: #35

Summary: Propulsion Power

Description

Rationale

Arial 10 B I U

The original gas engine of the aircraft shall be replaced by an equivalent output electrical motor, able to supply at least 350 kW of mechanical power at 2,300 RPM.

Keywords:

Revision information:

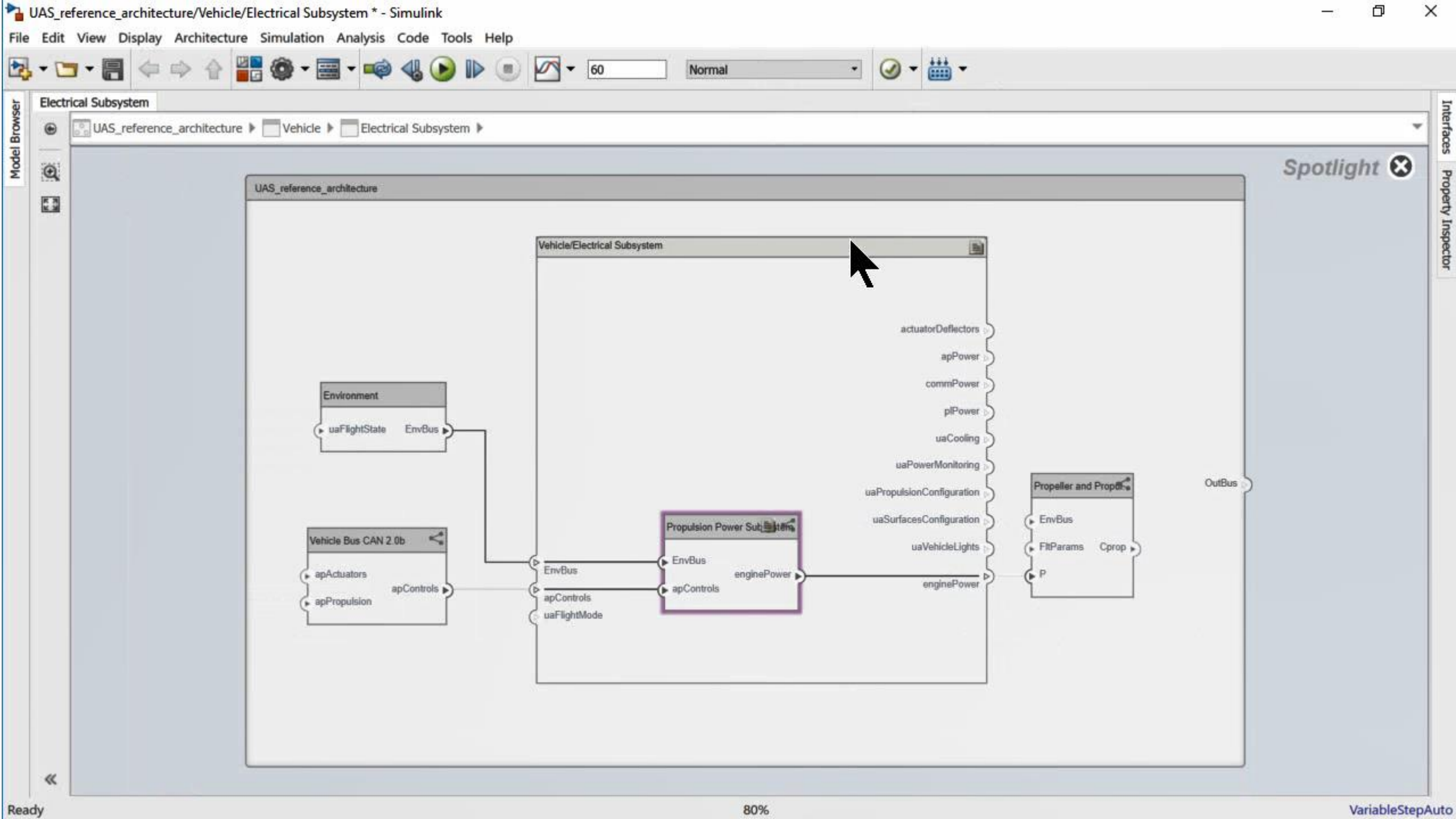
▼ Links

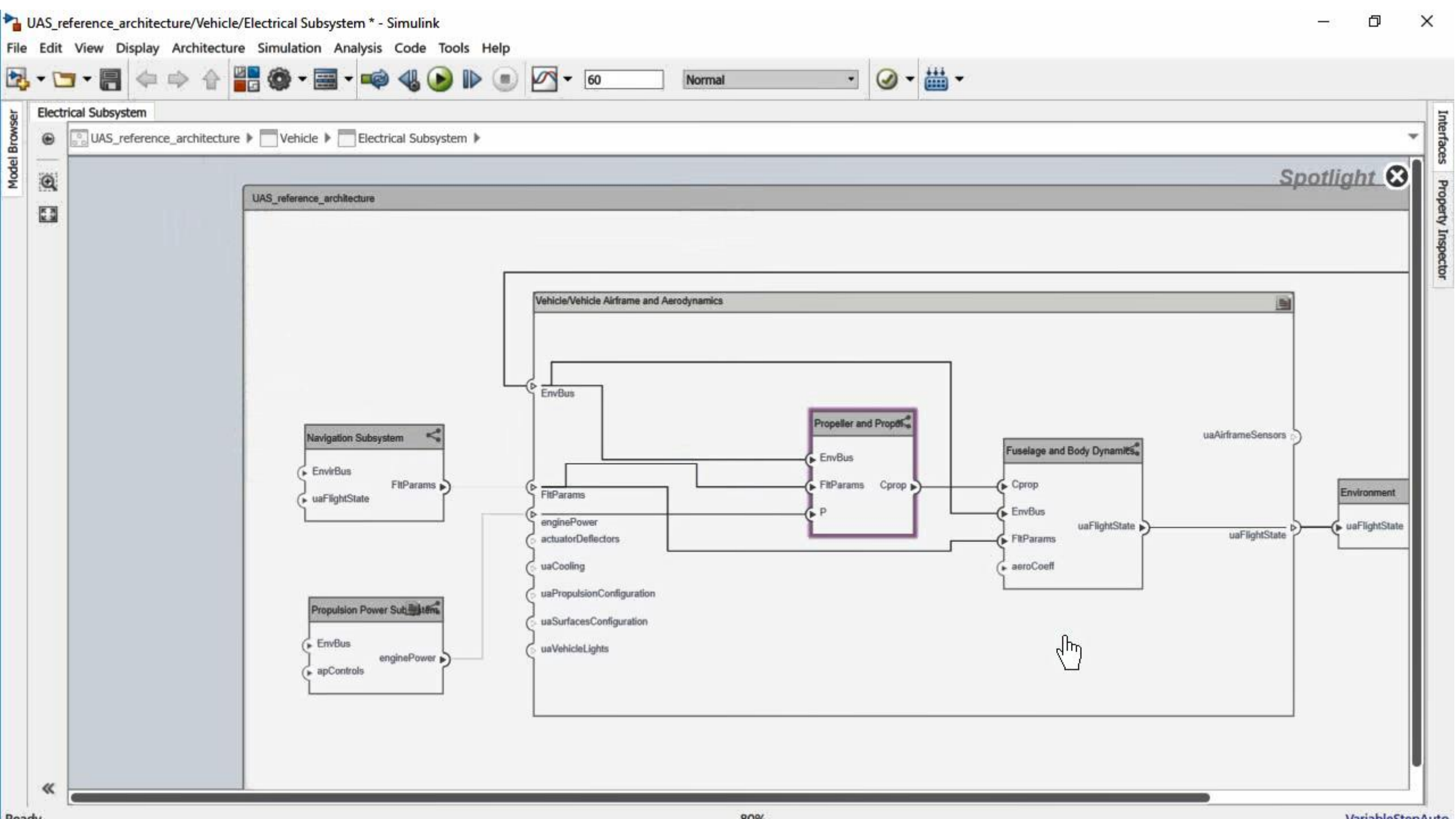
Implemented by:

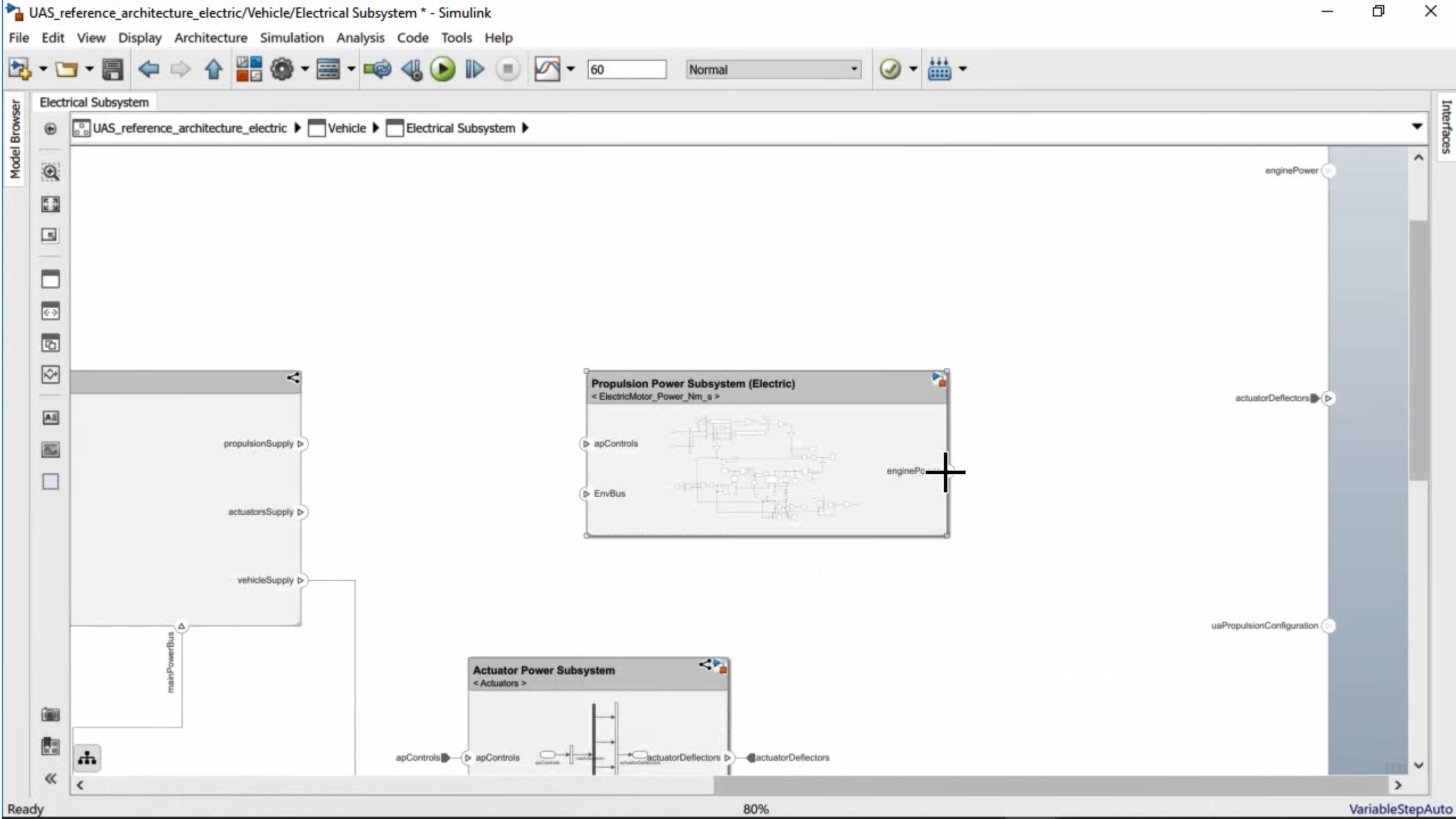
UAS_reference_architecture/.../Electrical Subsys

Change issue found for this link. Go to Links view

Comments

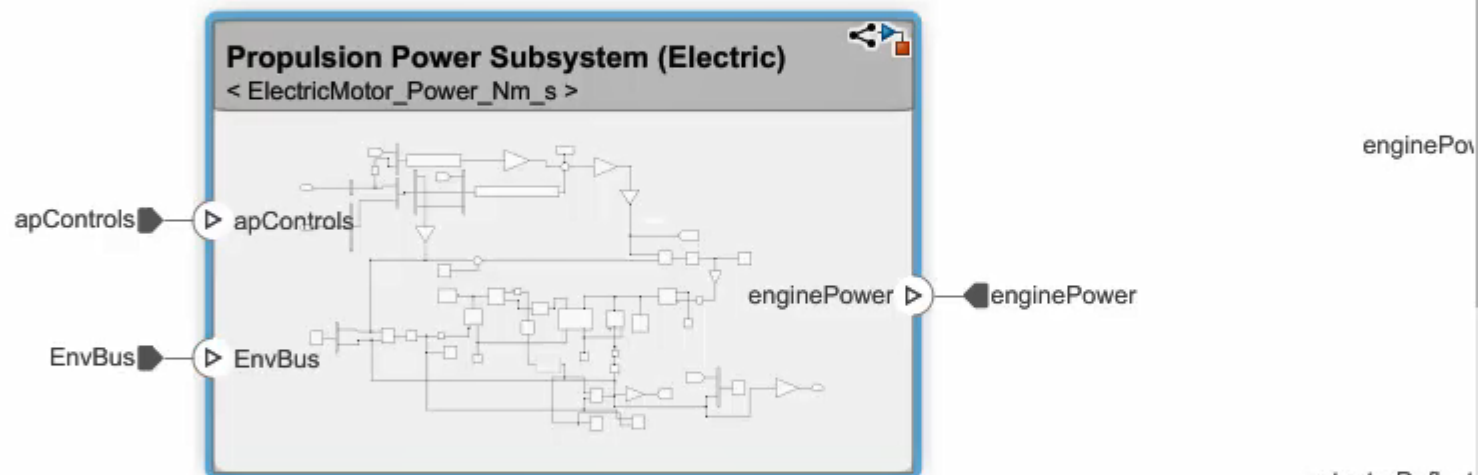






Electrical Subsystem

UAS_reference_architecture_electric ▶ Vehicle ▶ Electrical Subsystem ▶



enginePow

actuatorDeflect

uaPropulsionConfigui

Propulsion Power Subsystem (Electric)

< ElectricMotor Power Nm s >

Actuator Power Subsystem

< Actuators >

Property Inspector

Component	Value
1. Introduction	10%
2. Body	70%
3. Conclusion	10%
4. References	10%

Architecture

Info

NAME _____

VALUE

▼ Main

Name

Propulsion Power S...

Stereotype

Add.. ▼

▼ SubsystemBudget

Select

Mass

100 kg

Power

175000000 mW



Instantiate Architecture Model

Description

Create an instance model from this architecture model by flattening out all referenced models and their components. Such an instance model may be used for system-level analysis expressed as MATLAB functions.

Step 1: Select Stereotypes

Select the stereotypes to make available on the instance model

- ☒ UAVComponent
- ☒ SubsystemBudget

Step 2: Configure Analysis

Function

Analysis function:

Function arguments (comma-separated):

```
>> budgetRollUp(instance)
```

Model Iteration

Mode:

Instance model name

Don't see your profile? [Profile Editor ...](#)

	VALUE
me	Propulsion Power S...
otype	Add..
ystemBudget	Select
ss	100 kg
wer	175000000 mW

HOME






☐ Continuous
 
☐ Automatic

New Open Save Delete Analyze Arguments BottomUp Update ☐ Overwrite

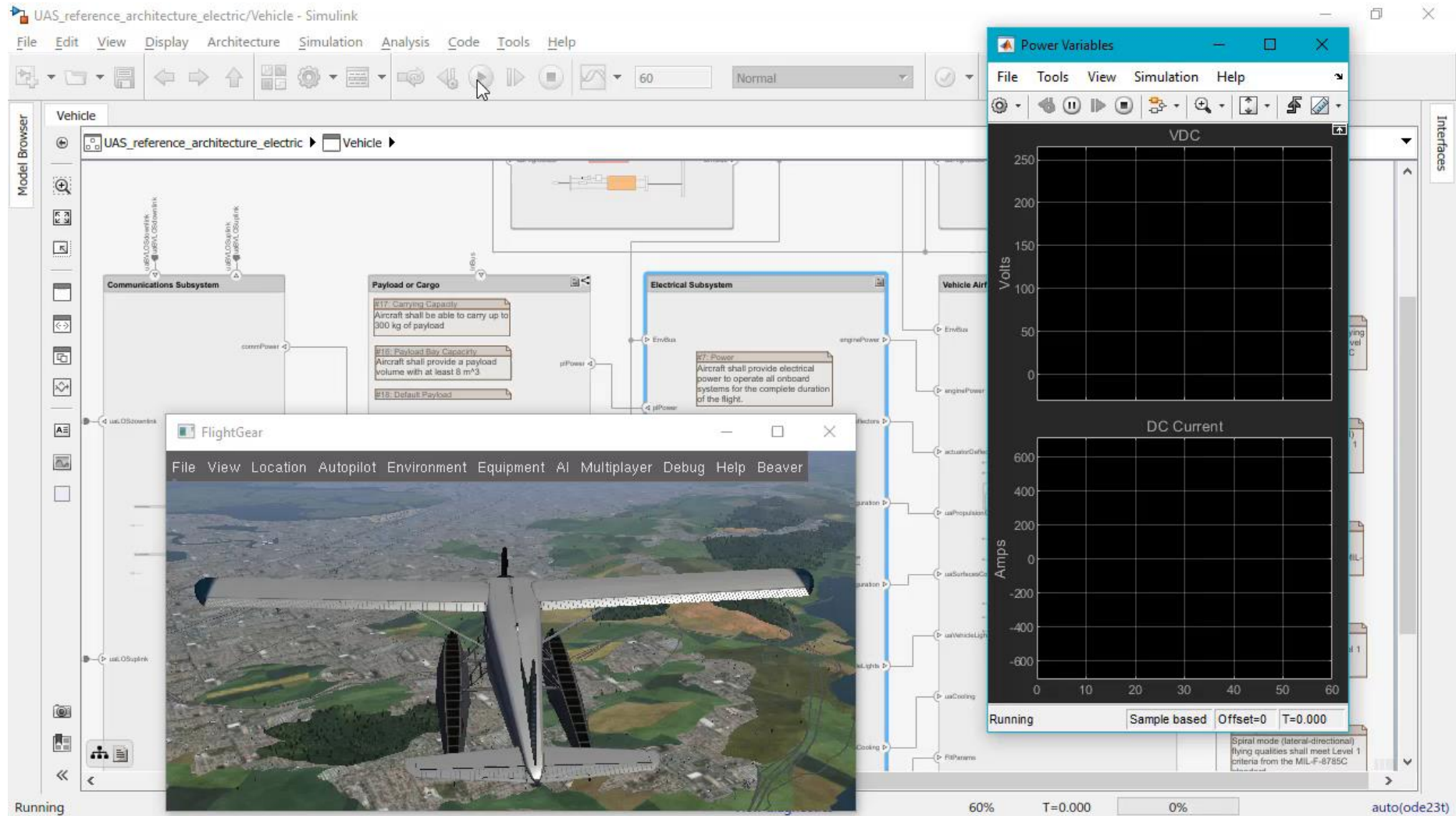
INSTANCE MODEL ANALYSIS UPDATE

Instances	Mass	Power
UAS_reference_architecture_electric_budgetRollup	392.33	200614300
BVLOS Navigation	0	0
Ground Station	0	0
Communication Box	0	0
Ground Station GPS interface	0	0
USB Serial Converter	0	0
Wireless Communication Subsystem	0	0
GPS receiver	0	0
Guidance and Navigation Computer	0	0
Flight Commands	0	0
Payload Computer	0	0
Vehicle	392.33	200614300
Communications Subsystem	2.63	58050
Automatic Dependent Surveillance-Broadcast	0.05	5000
C-Band Radio Modem	0.05	2000
KU-Band Radio TX/RX	2.5	50000
On-Board GPS	0.01	50
Radio RX PPM/PWM	0.02	1000
Electrical Subsystem	143.15	200355090
Actuator Power Subsystem	8	300000
Power Distribution	10	1000
Power Monitor	0	0
Power Source	20	1000
Propulsion Power Subsystem (Electric)	100	200000000
Vehicle Power Subsystem	5	50000
apRegulator	0.05	20
commRegulator	0.05	1070
piRegulator	0.05	2000
Environment	0	0

INSTANCE PROPERTIES

NodeInstance: Propulsion Power Subsystem (Electric)

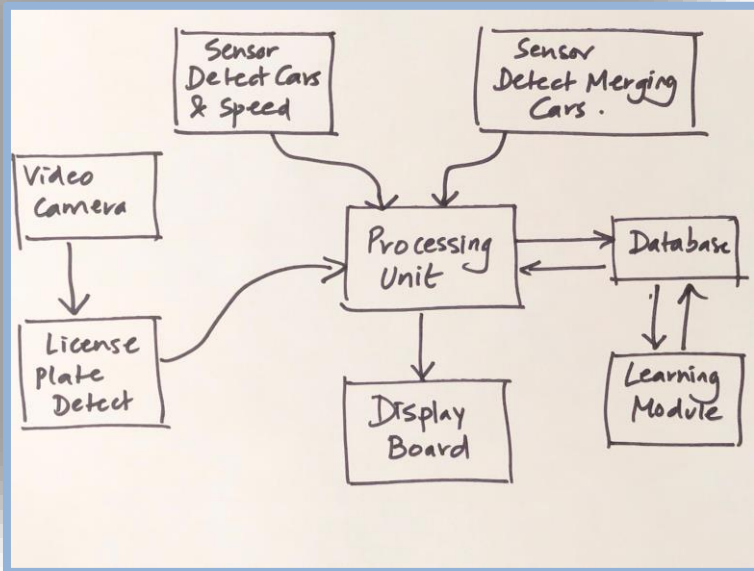
Property	Units	Value	Edit
SubsystemBudget			
Mass	kg	100	
Power	mW	200,000,000	



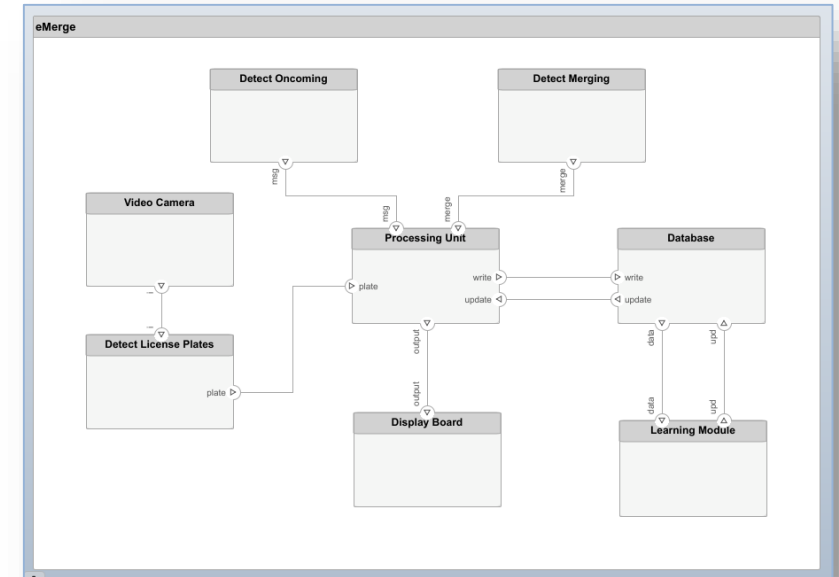
System Composer

Intuitively design system and software architectures

R2019a



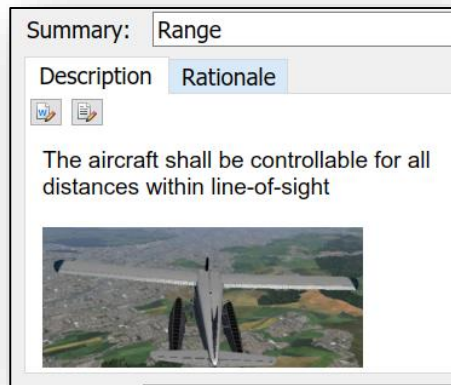
Description
==
Architecture



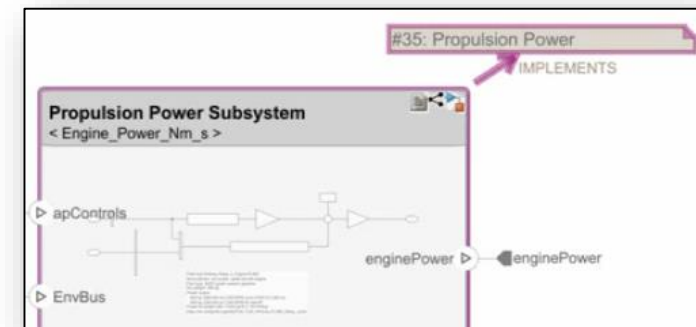
Simulink Requirements

Digital Thread from Requirements to Architecture and Design

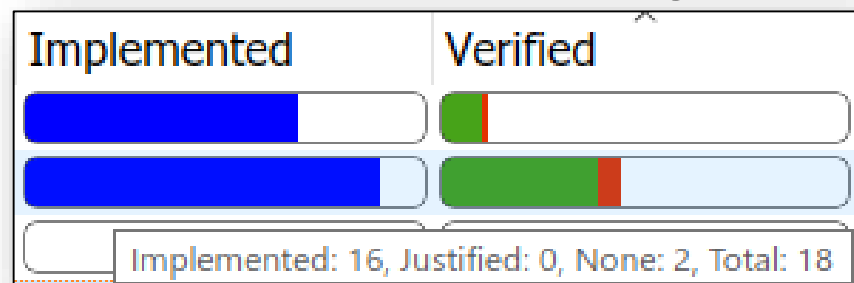
Author requirements or
view from external source



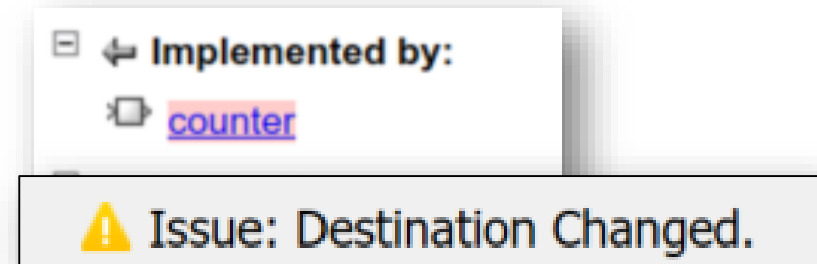
Link requirements, architectures,
design, code and test



Identify gaps in
architecture or design



Identify impact of requirement changes

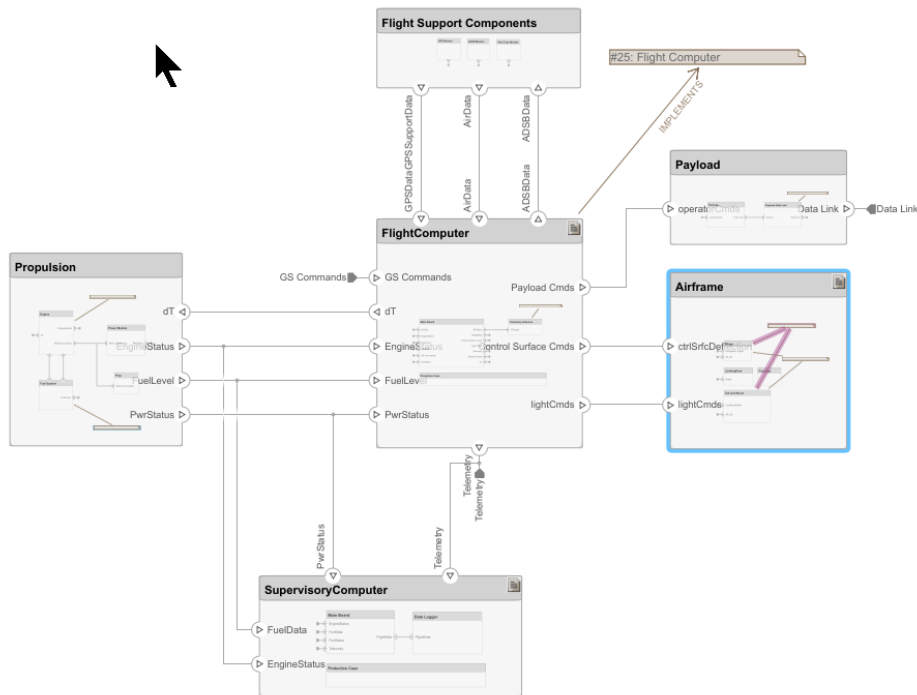


System Composer

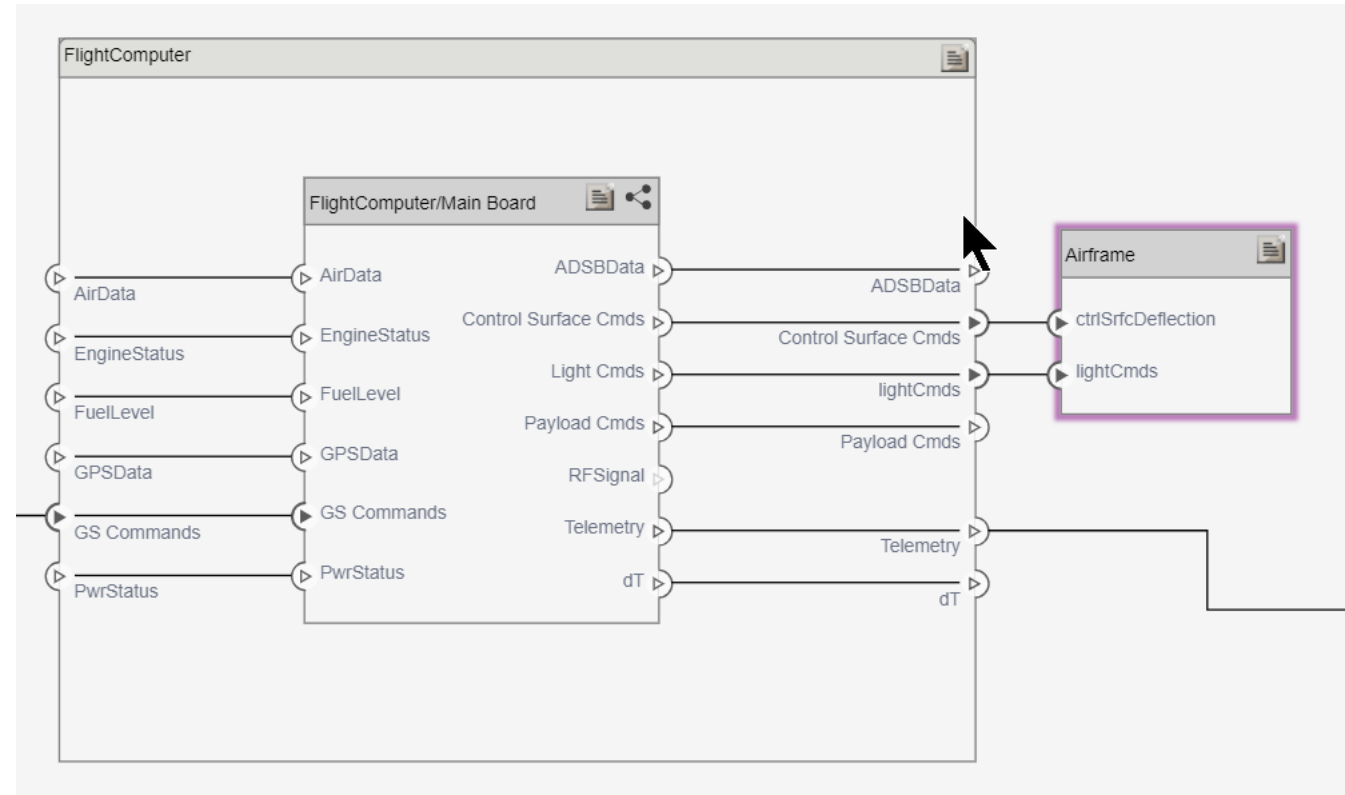
Tackle Architecture complexity with spotlight views

R2019a

Composition



Spotlight



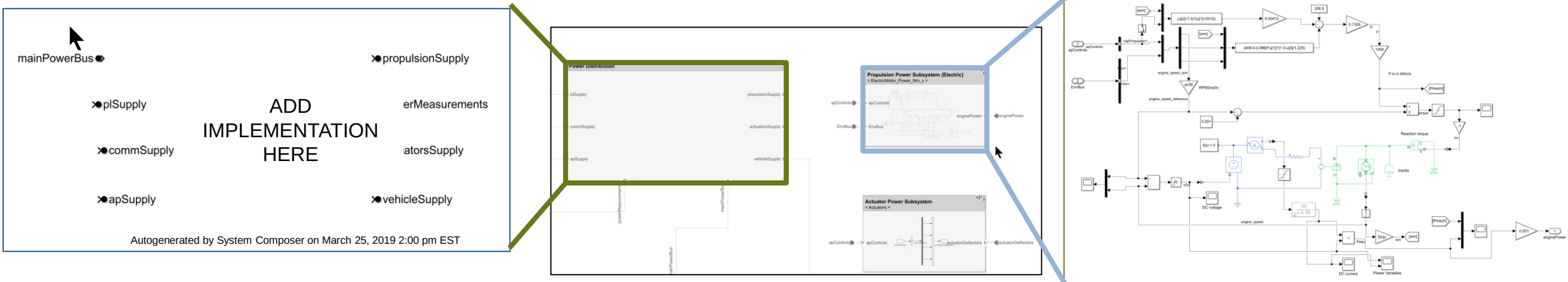
System Composer

System and software architectures connected to implementations in Simulink

R2019a

Generate Simulink models from
architecture components

Link Simulink models to
architecture components

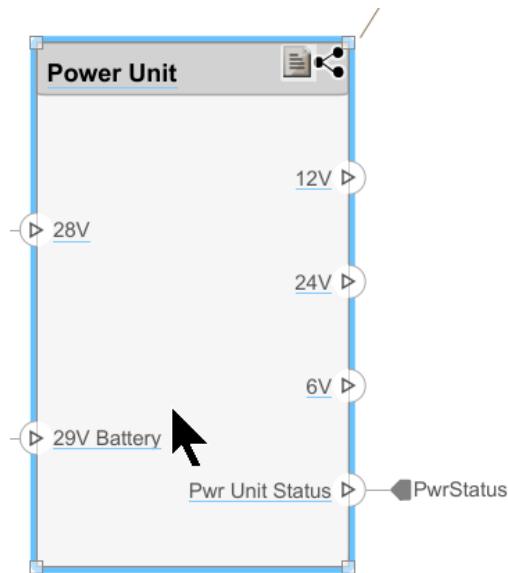


System Composer

Perform trade studies based on data driven analysis to optimize architectures

R2019a

Add custom data



Architecture	Info
NAME	VALUE
▼ Main	
Name	Power Unit
Stereotype	Add..
▼ OnboardElement	Select
Mass	0.217 kg
Power	0 mW
RFHarnessLength	0 cm

Create analysis model

Instances	Mass(kg)
SmallUAV	0
Airframe	0
Fuselage	1.7
LandingGear	1.65
Tail and Boom	2.7
Wings	3.2
Flight Support Components	0
ADSB Module	0
ABDSB Antenna	0.058
ADSB Board	0.098
GPS Module	0
GPS Antenna	0.128
GPS Board	0.27
Pitot Tube Module	0.075
FlightComputer	0
Main Board	0.145
Protective Case	0.195

Calculate mass roll-up data

Instances	Mass(kg)
SmallUAV	15.932
Airframe	9.25
Fuselage	1.7
LandingGear	1.65
Tail and Boom	2.7
Wings	3.2
Flight Support Components	0.629
ADSB Module	0.156
ABDSB Antenna	0.058
ADSB Board	0.098
GPS Module	0.398
GPS Antenna	0.128
GPS Board	0.27
Pitot Tube Module	0.075
FlightComputer	0.388
Main Board	0.145
Protective Case	0.195

Simulink: A Multi-Language Simulation Environment



Dynamic Systems



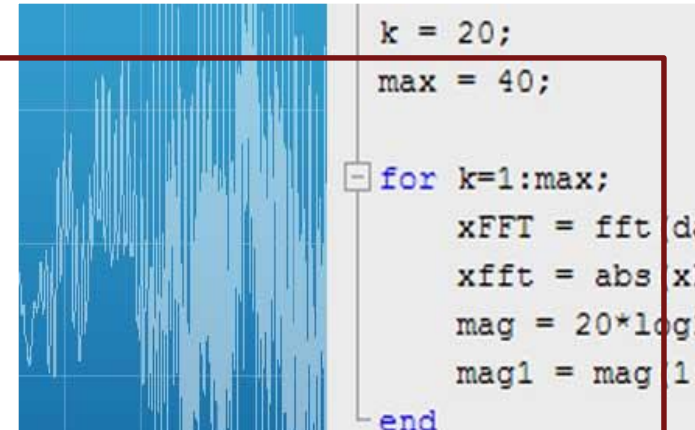
State Machines



Discrete-Event Systems



Physical Modeling



Programming Languages

Learn More

- [Simulink Requirement Webpage](#)
- [System Composer Webpage](#)
- [System Modeling and Simulation Webpage](#)
- [Trial](#)

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