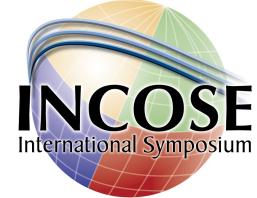


Recommended Verification Approach for Human Spaceflight Vehicles



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Introduction



- Human spaceflight necessitates increased scrutiny because of the risks to human life.
- Requirement and hardware verification is the way that this scrutiny is satisfied.
- A Program must be able to provide verification that is convincing, cohesive, and coherent.
- This paper provides a model for verification planning and execution based on lessons learned during a development program.

Application Of This Presentation



- New Government Development Programs
 - Commercial Crew Development
 - Shuttle retirement and replacement
 - Planetary exploration goals
- Space Tourism
- Applications outside of Aerospace Industry where public safety paramount or where strict performance measures must be satisfied.

About The Authors

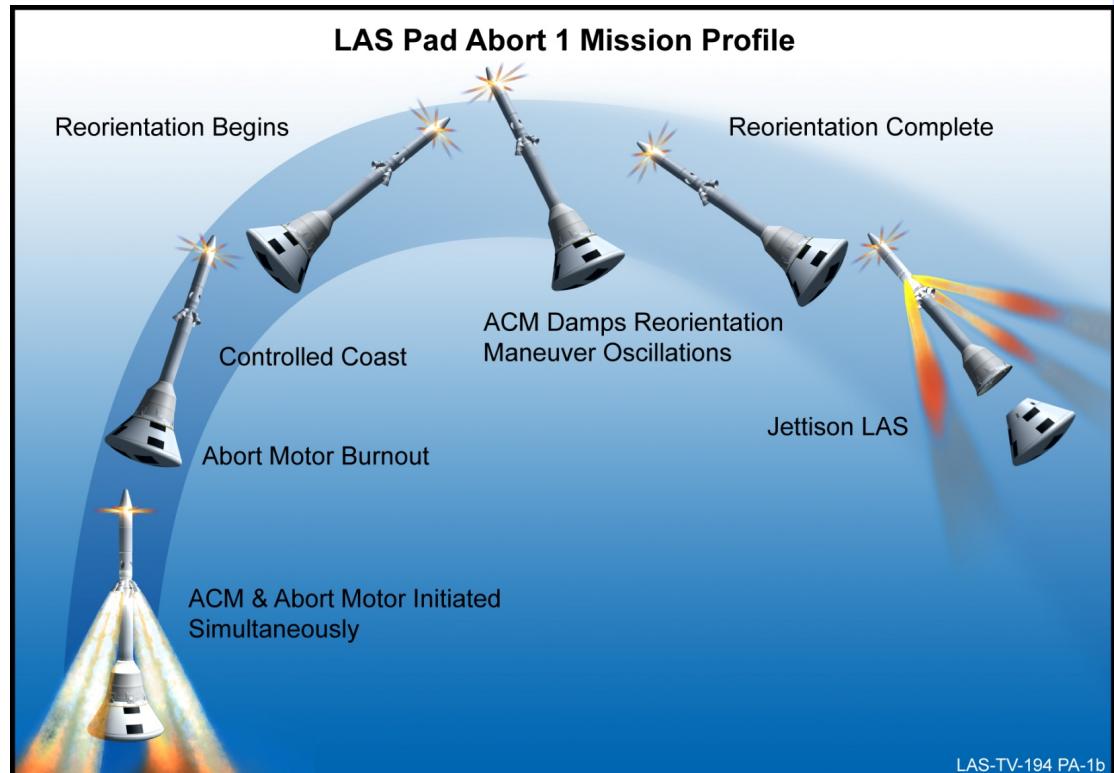


- Systems Engineering Team from Orbital Sciences Corp. for the Orion Launch Abort System (LAS)
 - Charlotte Pappageorge
 - Fitz Vernon
 - Darko Filipi
 - Ben Herbert
 - Ken Bocam



Launch Abort System Overview

- LAS is designed to remove the crew during a launch vehicle failure while on the pad and up to a nominal jettison at approximately 300 kft
- LAS accomplishes this mission using
 - Three solid motors
 - Trajectory optimization during the abort
 - Lightweight composite structures

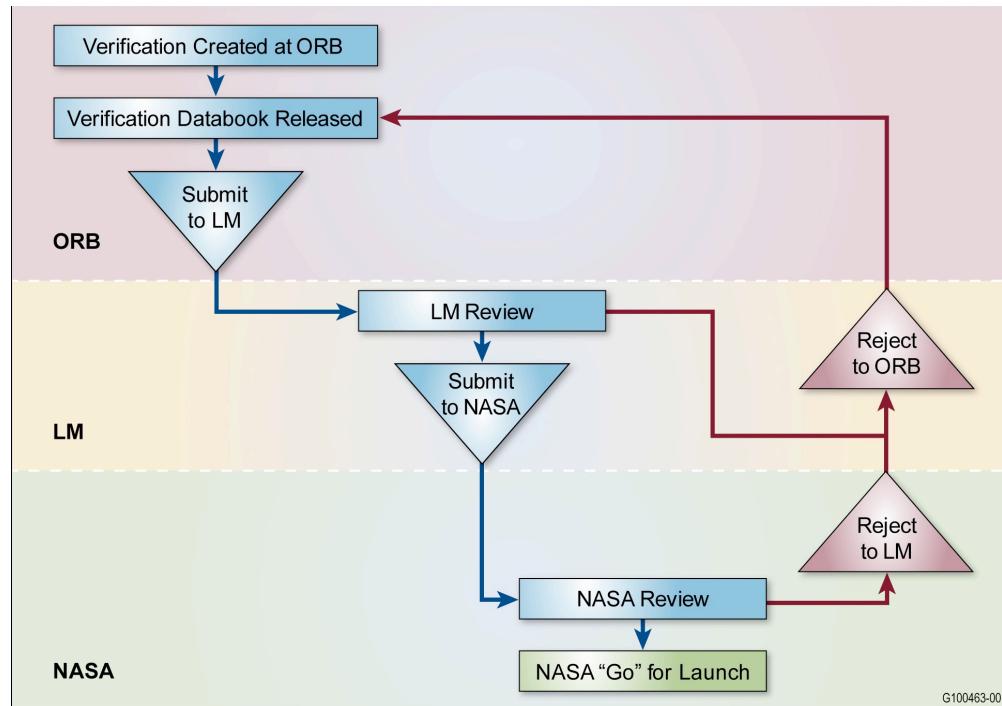


- To date, commercial spacecraft have been uncrewed systems that are usually variations on an existing design.
- Commercial verification is usually based on addressing changes from previously flown or qualified vehicles.
- Greater risks are accepted to reduce cost in absence of a risk to human life.

Human Spaceflight Requires More Rigor

The PA-1 Verification Approach

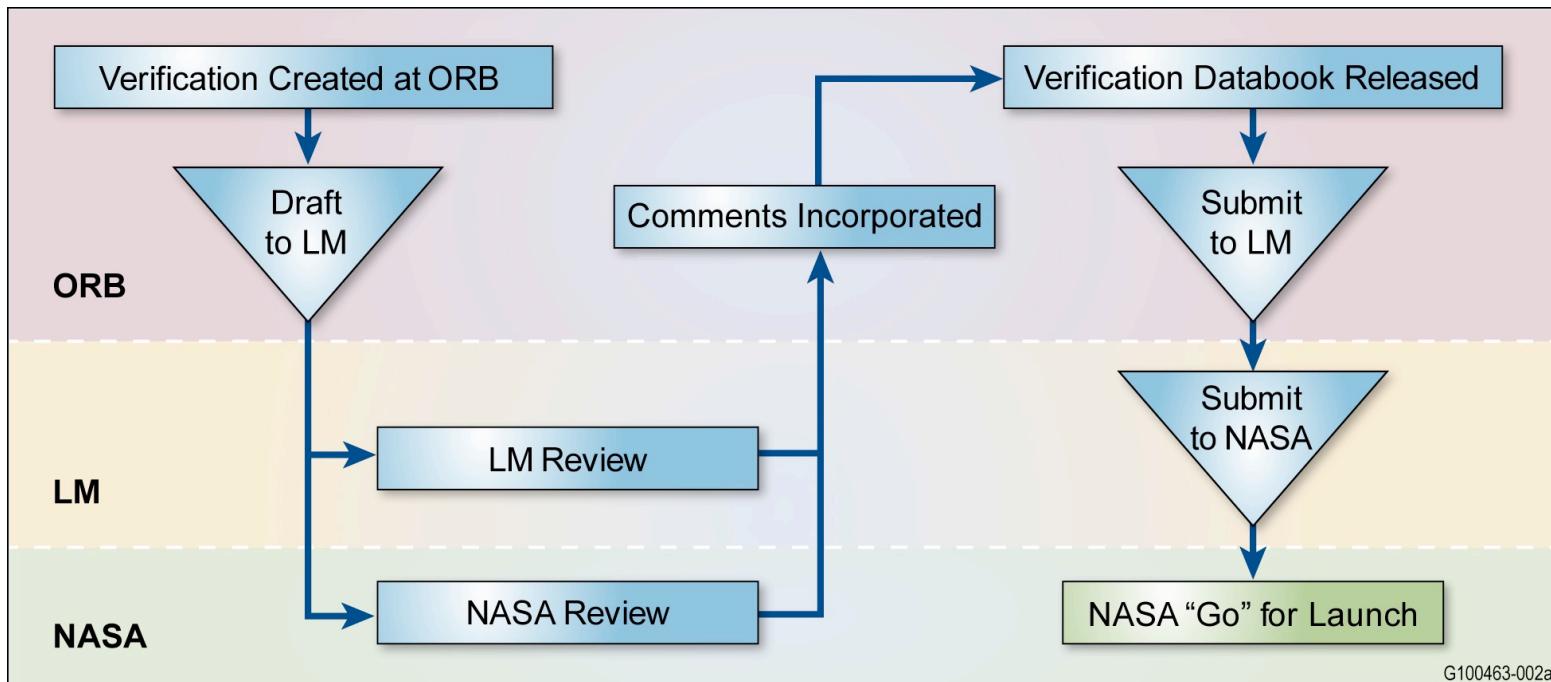
- Originally all verification data for the test flight would be delivered in a single data drop.



- This review process impacts the flight schedule if customers encounter any issues with the verification documentation provided.

The PA-1 Verification Approach

- Orbital, Lockheed Martin, and NASA came up with a new review process that allowed draft submissions and incorporation of review comments before the Verification Databook was delivered.



➤ Verification Method:

- An event or process that has a specific goal related to verification. Verification methods define whether verification is to be accomplished by a test, analysis, demonstration, or inspection.

➤ Verification Event:

- The specific activity used to provide verification data. Verification events define **WHAT** test, analysis, demonstration, or inspection will verify a requirement. The verification event includes a description of the expected event.

The PA-1 Verification Approach



- Flight test articles had the same requirements as production (crewed) vehicles.
- Qualification campaigns weren't needed for most vehicle components and subsystems used on PA-1
- An event centric approach was used to verify requirements for the flight tests.
- Test and analysis reports directly address requirement verification.

- Verification Methods and Events were coordinated in Working Groups with Lockheed Martin and NASA to decide what data were expected for requirement closure.
- A Verification Cross Reference Matrix (VCRM) was developed for each specification with the WG inputs.
- Completion of the Verification Events in the VCRM constitutes verification of the requirement.
- Documentation of the success of each event is submitted as verification evidence.

Verification Closure Example



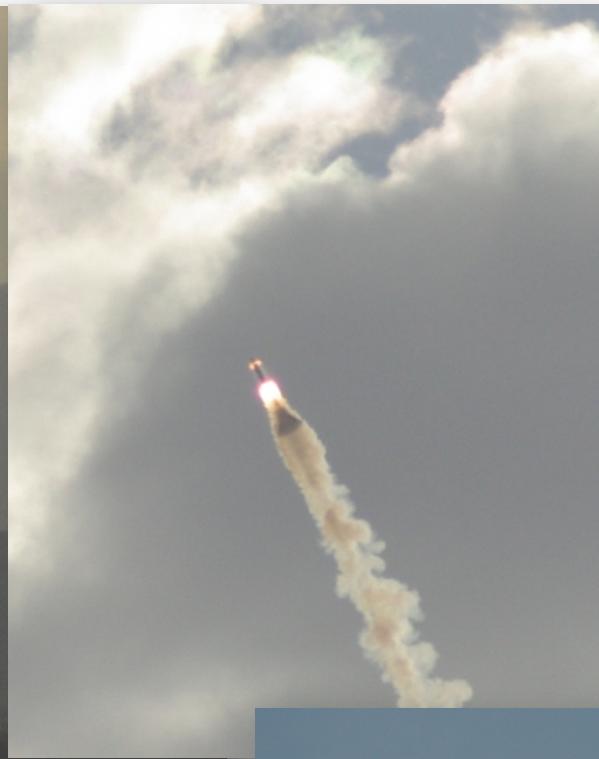
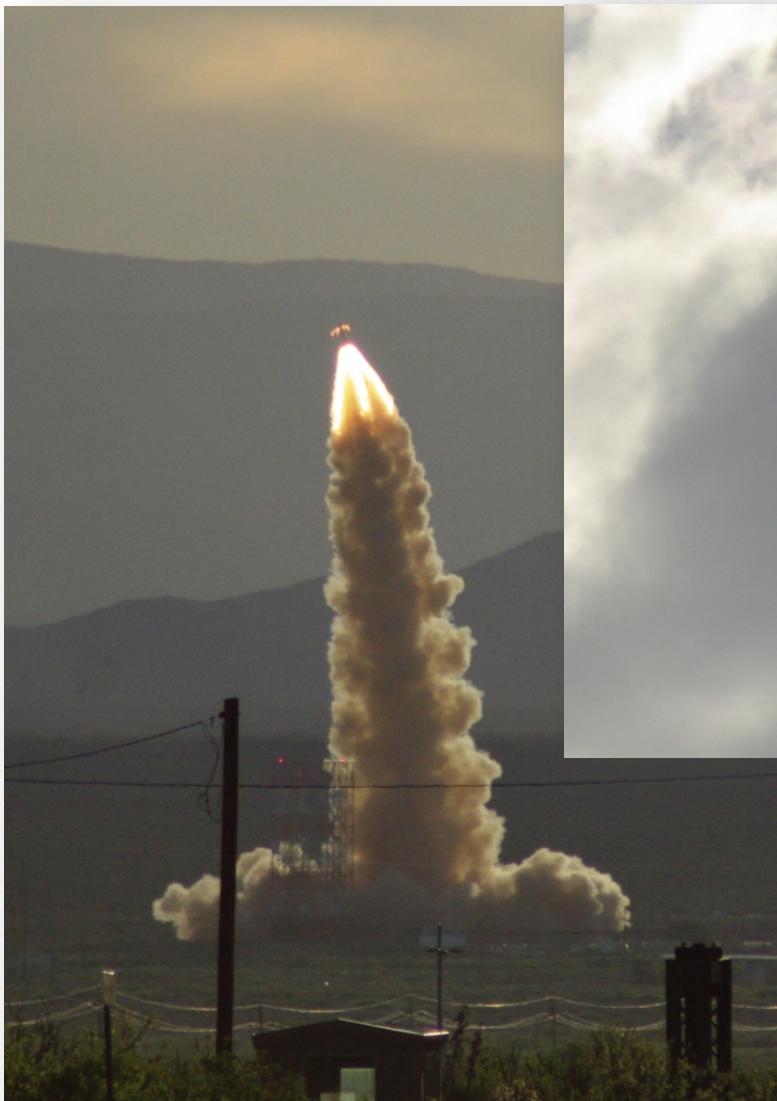
- In this example a requirement is verified by two tests and an analysis.

Verification Status: Verified



- Documentation references for these events is included in a compliance matrix and in the Verification Databook.

Mission Outcome: Success!



Problems With This Process

- Complicated reporting of status
- Usually, multiple verification reports and versions of the Databook are submitted to close a single requirement
- Reports separated from the Databook lost their context and are confusing or incomplete for reviewers.

Requirements									
Number	Name	Text	Status	Associated Verification			Verification Documentation		
				Number	Name	VMStatus	Document Number	Rev	Document Name
CP-7	Loads	The component shall meet all functional and performance requirements after exposed to the loads in the table below	Verified	CP-A5	Stress Analysis	Complete	6029-ER3093	-	Structural Analysis Report
							6029-ER30918	A	Structural Analysis Report
							6029-ER30926	A	Secondary Structural Analysis Report
							6029-TR3161	-	Bracket Assembly Static Loads Test
				CP-T6	Static Loads Test	Complete	6029-ER201153	-	Static Loads Test Verification Report
							6029-TR3161	-	Bracket Assembly Static Loads Test
							6029-TR3092	-	Structural Analysis Report

Requirement Detail Verification Method Verification Documentation

G100366-002

Problems With This Process



- The presentation of compliance data is fragmented.
 - This doesn't allow for **CONVINCING** verification to be easily reported to a reviewer.
- Multiple events that have related outcomes are not tied together for system considerations.
 - This doesn't allow for **COHERENT** verification to be easily reported to a reviewer.
- Systems Engineers and subject matter experts must explain results to tie fragments together.

Convincing



Coherent



Concise



Lessons Learned:



- Define verification expectations during requirements definition rather than vague verification tasks.
 - Stakeholders need to understand and communicate what they want early in the requirement development process.
- Relate verification data to requirements in the document where the data is generated.
 - When analysis or test reports didn't directly address requirement statements, additional reports had to be written to relate results to requirements.
- Prevent duplication of effort in verification definition and reporting.

Recommended Approach: Overview

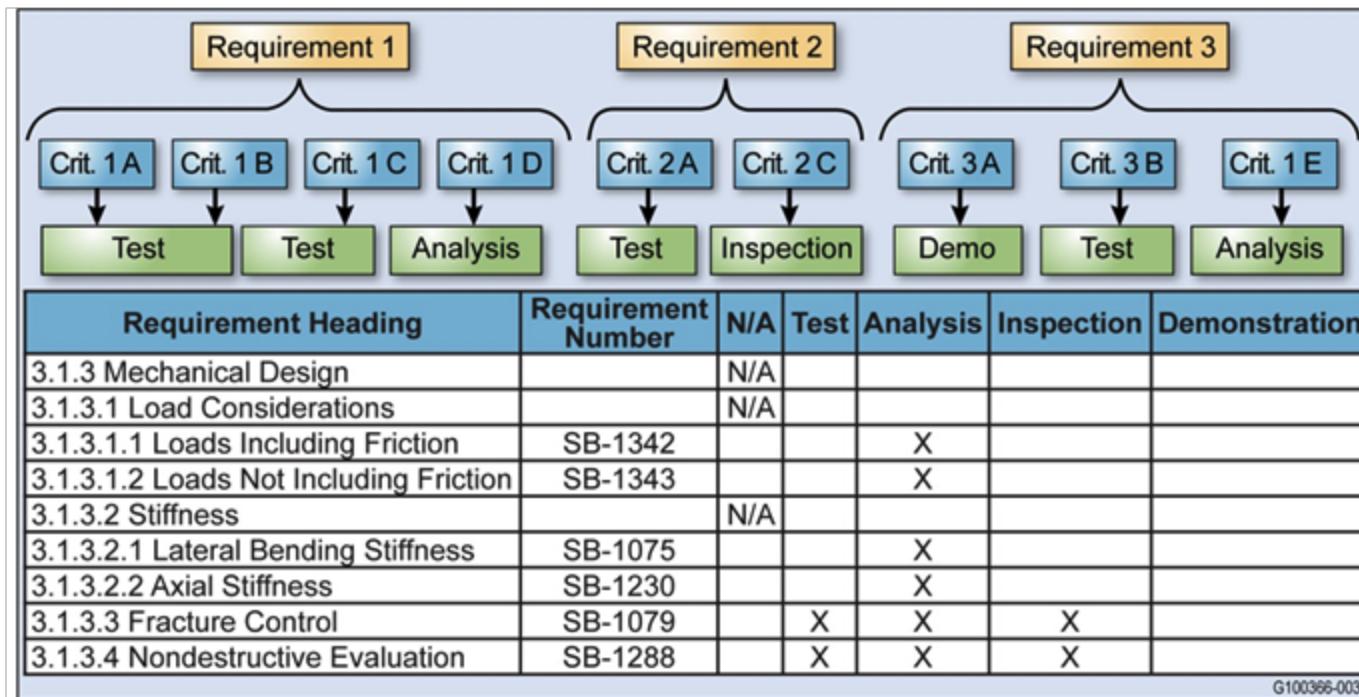


- Develop the verification plan concurrently with requirement development.
- Identify what inputs and results are needed for closure of a requirement.
- Document verification proof for each requirement in a single location.

Recommended Approach: Verification Planning



- Focus on what data (criteria) are necessary for verification of a requirement.
- Data requirements drive verification method selection.



Recommended Approach: Verification Criteria



- Verification criteria specify the minimum standard for showing the requirement is met.
- Verification criteria provide the plan and success criteria to verify a requirement.
- The criteria should be documented and approved internally, possibly in the Master Verification Plan.
- Verification criteria should be defined in a way that allows an engineer to know when the requirement is met in the context of the other requirements.

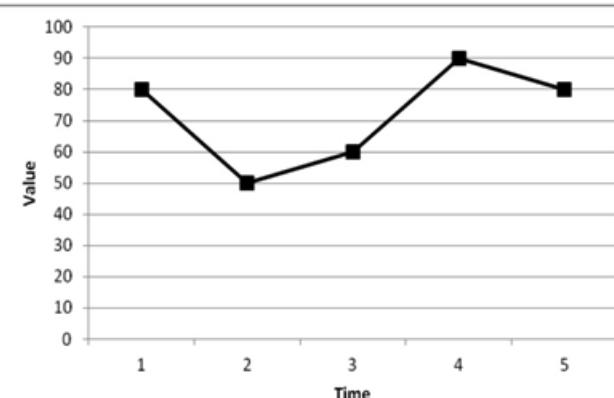
Implementation Examples: Verification Criteria



- All information about this requirement is available for easy reference and review

1.1 SB-1006: Subsystem Minimum Value

The subsystem value shall be above the curve in the table, titled "Subsystem Minimum Value," below.



Time	Value
1	80
2	50
3	60
4	90
5	80

Figure 1 – Subsystem Minimum Value

The above requirement must be met and the following Verification Criteria satisfied.

Verification Criteria:

- 1) The analysis(s) conducted to verify the requirement shall be validated with qualification test data from at least three tests. Developmental data may be used to supplement qualification data as appropriate.
- 2) The analysis(s) conducted to verify the requirement shall include the effects of shelf life requirements.
- 3) The analysis(s) conducted to verify the requirement shall include the effects of environmental requirements.
- 4) The analysis(s) conducted to verify the requirement shall show statistical compliance using three sigma uncertainties.

Table 1 - Verification Methods Assigned To SB-1006

Test	Analysis	Inspection	Demonstration
X	X		

Requirement Text

Requirement Tables and Figures

List of Verification Criteria

Verification Methods from Specification

G100368-007b

Presentation for the INCOSE Symposium 2011 Denver, CO USA

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Example Verification Criteria



3.2 SB-AM-1007: Delivered Impulse

The abort motor shall provide a minimum impulse of 996,600 lbf-sec during the action time burn at vacuum, a propellant mean bulk temperature of 30 degrees F, -3 sigma burn rate conditions and -3 sigma total impulse conditions (note that total impulse variability encompasses specific impulse and propellant mass variabilities).

The above requirement must be met and the following verification criteria satisfied.

Verification Criteria:

- 1) The analysis task(s) conducted to verify the requirement shall be validated with qualification test data from at least three static fires. Developmental data may be used to supplement qualification data as appropriate.
- 2) The analysis task(s) conducted to verify the requirement shall include the effects life requirements (e.g. propellant aging).
- 3) The analysis task(s) conducted to verify the requirement shall include the effects of environmental requirements (e.g. temperature, pressure).
- 4) The analysis task(s) conducted to verify the requirement shall show statistical compliance using three sigma uncertainties.
- 5) Test data from at least three qualification static fires shall show compliance to the requirement in the cold condition with one TBI and in the hot condition with two TBIs, at a minimum. Developmental data may be used to supplement qualification data as appropriate.

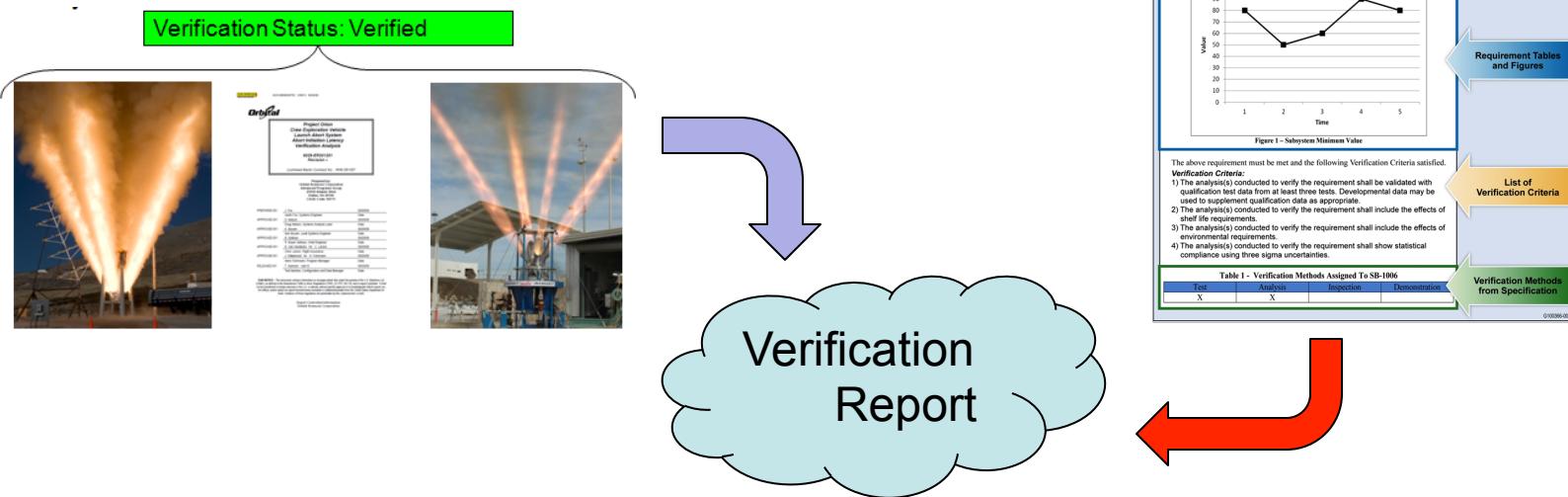
The following are the verification methods assigned to SB-AM-1007

Test	Analysis	Inspection	Demonstration
X	X		

Recommended Approach: Verification Reporting



- Address requirements one at a time and completely in a dedicated report
- Present verification data with verification criteria and the requirement for context
- Include the actual data – do not just reference it
- Address all verification criteria



Implementation Examples: Verification Closure Notice



4.3. Req.0003: Length

The subsystem length shall be 100.0 +/- 0.2 inches from the forward edge of the forward flange to the aft edge of the aft flange.

Req.0003 is verified by Analysis and Inspection via the following criteria:

- 1) The analysis(s) conducted to verify the requirement shall show that the dimension and tolerance are achievable via tolerance stack up of the contributing components (as appropriate).
- 2) The inspection(s) conducted to verify the requirement shall identify the location and tolerance of the required motor length on the applicable drawing(s).

On drawing XXX (Subsystem Assembly Drawing), sheet 2 of 4, zone A5 is the design length. The drawing length is specified from the forward edge of the forward flange to the aft edge of the aft flange (see Figure 2). The length on the drawing is specified to be 100.0 +/- 0.1 inches which is within the specified bounds of the requirement.

[FIGURE SUPPRESSED]
Figure 2.

The tolerance stackup analysis calculated the tolerance stackup to be +/- 0.09 (see Section X.X of document PLXXX).

Through the drawing inspection and tolerance stackup analysis compliance was shown to the Req.0003. The Req.0003 requirement compliance is Closed: Verified.

All Data About Requirements:
Number, Name, Text, Figures,
Verification Criteria

All Proof Required
Or Available For Showing
Required Compliance

G100366-004b

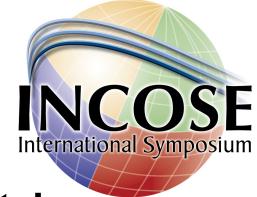
Implementation Proof



- These documents and techniques have been employed with considerably more success.



Conclusion



- Verification is reported in a summary document that has a focused scope.
 - This enables **CONCISE** verification statements to be written by referencing instead of including the background of a verification event.

VCN ID	Mission	Rev	Verification Closure Notice (VCN)		Status
Owner - NASA	Owner - Orbital				Estimated Closure Date
REQ Document	Paragraph #	SS # (Orbital)	REQ Title		
SSP 50808	3.3.11.1.3.4	SS_8IRD_0727, [Safety]	VERIFIABLE SEAL LEAKAGE PATHS		
REQ Description Paths through which the atmosphere of any habitable pressurized section might leak to its external environment shall have the redundancy and verifiability requirements contained in Table 3.3.11.1.3.4-1, Seal Redundancy and Verified Requirements.					
VER Description					
Verification Event Description					
Doc #	Revision	Document Title	Para #		
Requirement Owner			NASA T&V		
Signature	Date	Signature	Date		

Concise



Coherent

Convincing

Conclusion



- The presentation of compliance data is in one place with all facets of the requirement addressed.
 - This enables **COHERENT** verification to be easily reported.

VCN ID	Mission	Rev	Verification Closure Notice (VCN)		Status
Owner - NASA	Owner - Orbital				Estimated Closure Date
REQ Document	Paragraph #	SS # (Orbital)	REQ Title		
SSP 50808	3.3.11.1.3.4	SS_8IRD_0727, [Safety]	VERIFIABLE SEAL LEAKAGE PATHS		
REQ Description					
Paths through which the atmosphere of any habitable pressurized section might leak to its external environment shall have the redundancy and verifiability requirements contained in Table 3.3.11.1.3.4-1, Seal Redundancy and Verified Requirements.					
VER Description					
Criteria 1 – The inspection performed shall identify all of the leak paths across the component by inspection of drawings or CAD models. Criteria 2 – The inspection performed shall identify the number and major diameter of seals across each leak path. Criteria 3 – The analysis performed shall allocate leak rates to each seal (or set of seals) to support the system leak rate requirement. Criteria 4 – The inspection performed shall identify acceptance tests procedures and success criteria that implement the seal redundancy requirements.					
Verification Event Description					
Doc #	Revision	Document Title	Para #		
Requirement Owner			NASA T&V		
Signature		Date	Signature		Date

Concise



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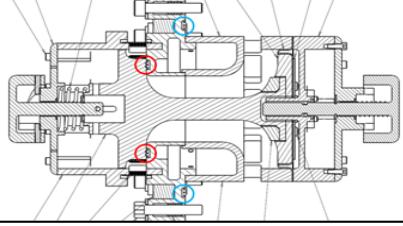


Convincing

Conclusion



- Verification events with related outcomes can be tied together by interspersing details from all applicable sources into a single narrative.
 - This enables **CONVINCING** verification by documenting system considerations that may extend beyond the scope of a single test

VCN ID	Mission	Rev	Verification Closure Notice (VCN)	Status
Owner - NASA	Owner - Orbital			Estimated Closure Date
REQ Document	Paragraph #	SS # (Orbital)	REQ Title	
SSP 50808	3.3.11.1.3.4	SS_8IRD_0727. [Safety]	VERIFIABLE SEAL LEAKAGE PATHS	
REQ Description				
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Verification Event Description				
PPRV Seals – The PPRV seals in two places. It has 2 seals with the PCM primary structure that are located inside the PCM near the threaded ends of the 6 fasteners that hold the PPRV in place. REMAINING TEXT SUPPRESSED FOR ITAR.				
				
Doc #	Revision	Document Title	Para #	
Requirement Owner		NASA T&V		
Signature	Date	Signature	Date	

Concise



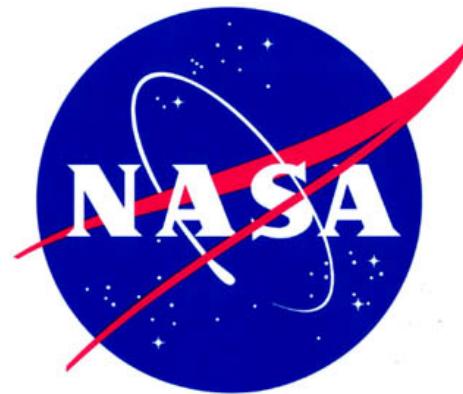
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Thank Yous



Questions

