



28th Annual **INCOSE**
international symposium

Washington, DC, USA
July 7 - 12, 2018

A UAV Case Study with Set Based Design

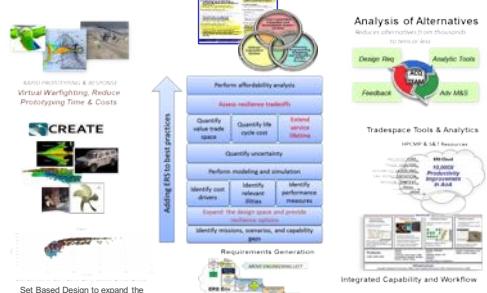
Colin Small, Dr. Greg Parnell, Dr. Simon Goerger, Dr. Ed Pohl, Dr. Matthew Cilli, Eric Specking, Zephan Wade

Bottom Line Up Front

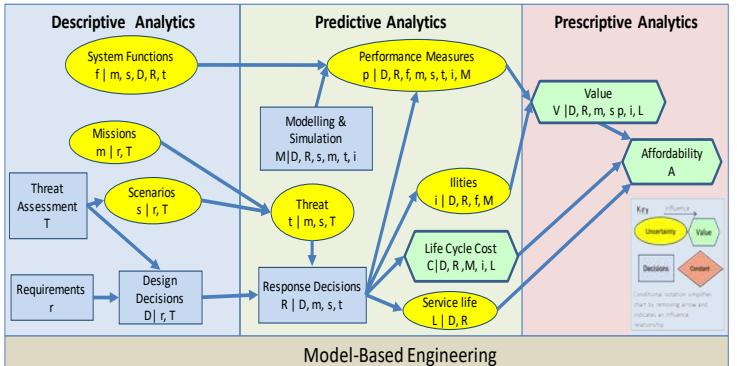


This research provides a case study using ERS/ARDEC UAV data that demonstrates the potential of Set-Based Design trade-off analytics in system decision making for Engineered Resilient Systems.

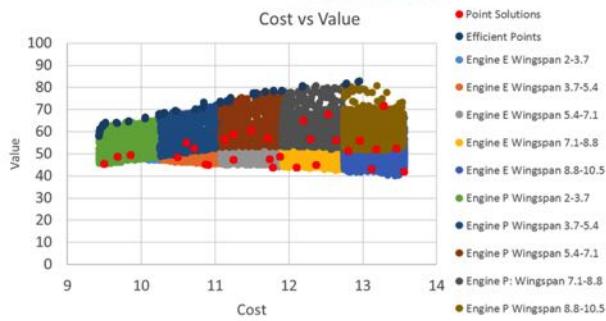
Incorporating ERS into AoAs



Trade-off Analytics Framework



PBD vs. SBD

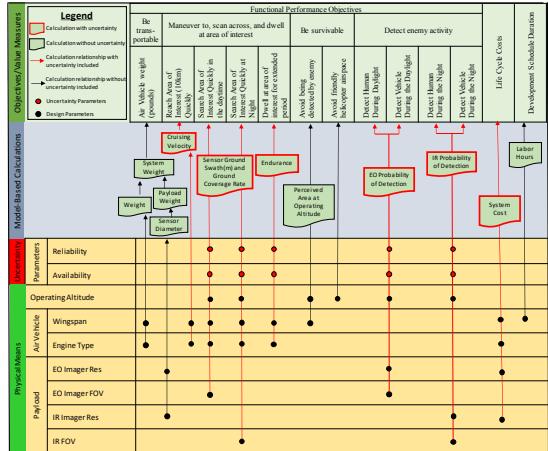


Small, C., Buchanan, R., Cilli, M., Parnell, G., Pohl, E., Wade, Z., "A UAV Case Study with Set-based Design," 28th Annual INCOSE International Symposium, 7-12 July 2018, Washington, DC.

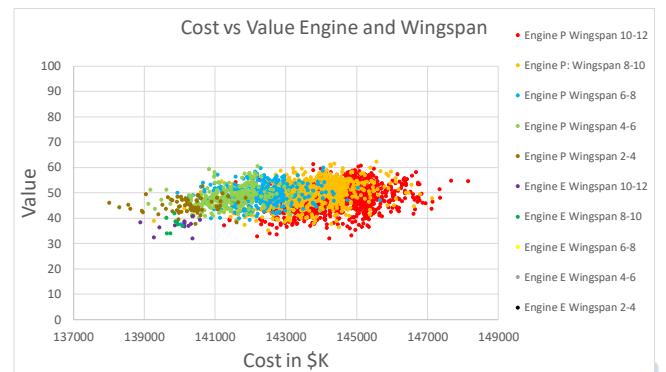
Perfect Options



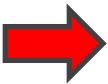
Implementing Trade-off analytics framework and SBD in UAV Tradespace Tool



Effects of uncertainty on SBD

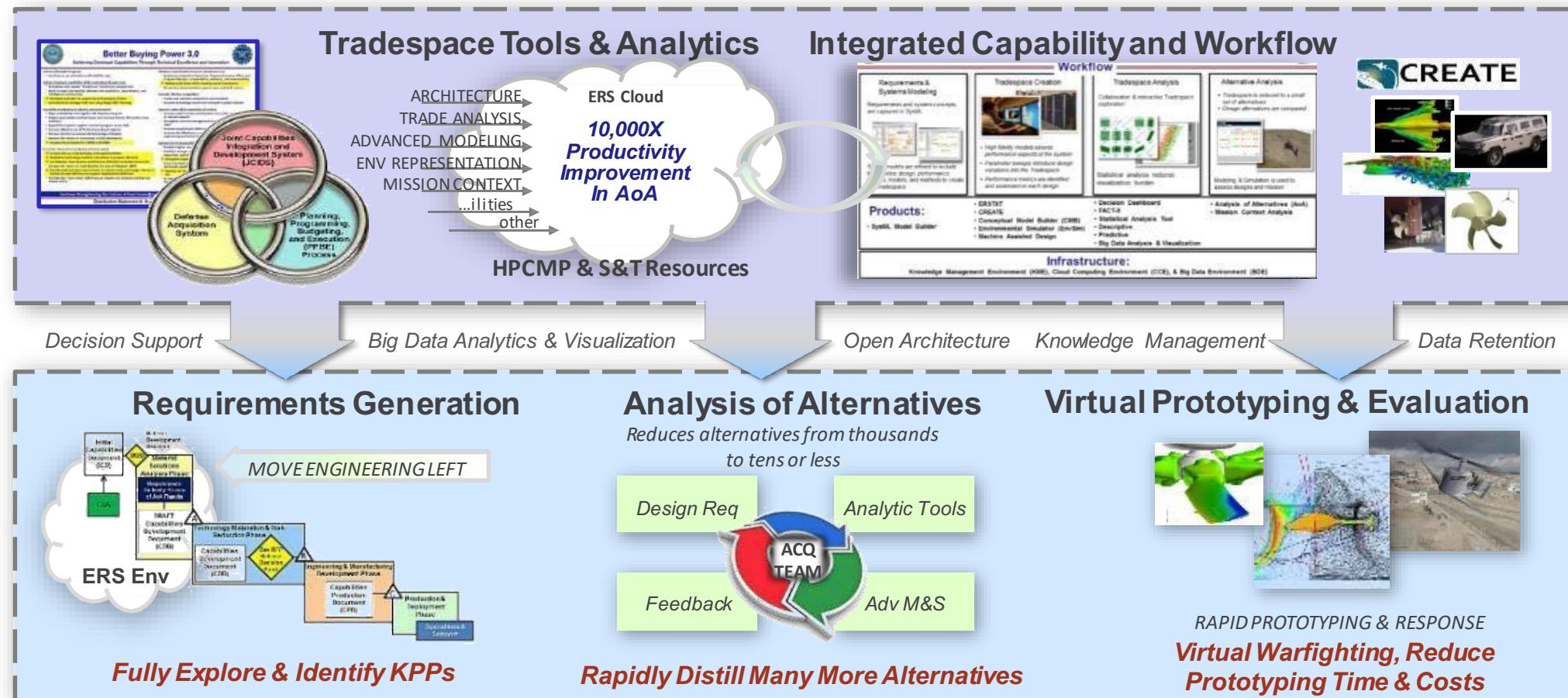


Overview



- Introduction
 - ERS
 - Trade-off Analytics Framework
 - Set Based Design
- Initial UAV Case Study
- UAV Tradespace Tool
 - Base model
 - Perfect Options
 - Uncertainty
- Conclusions
- Future Research

Engineered Resilient Systems (ERS)



Holland, J. P. (2015). Engineered Resilient Systems: Power of Advanced Modeling and Analytics in Support of Acquisition. NDIA 16th Science and Engineering Technology Conference. Springfield, VA.

Engineered Resilient Systems (ERS) is a Department of Defense (DoD) Program focusing on the effective and efficient design of complex engineered systems. This thesis is funded by the ERS program.

An Engineered Resilient System



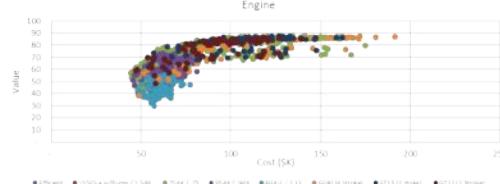
- “A resilient engineered system is able to successfully complete its planned mission(s) in the face of a disruption (environmental or adversarial), and has capabilities allowing it to successfully complete future missions with evolving threats”

Specking, E., Cilli, M., Parnell, G., Wade, Z., Cottam, B., & Small, C. (2017). *Tech Report: Graphical Representaiton of Resilient Engineered Systems*.

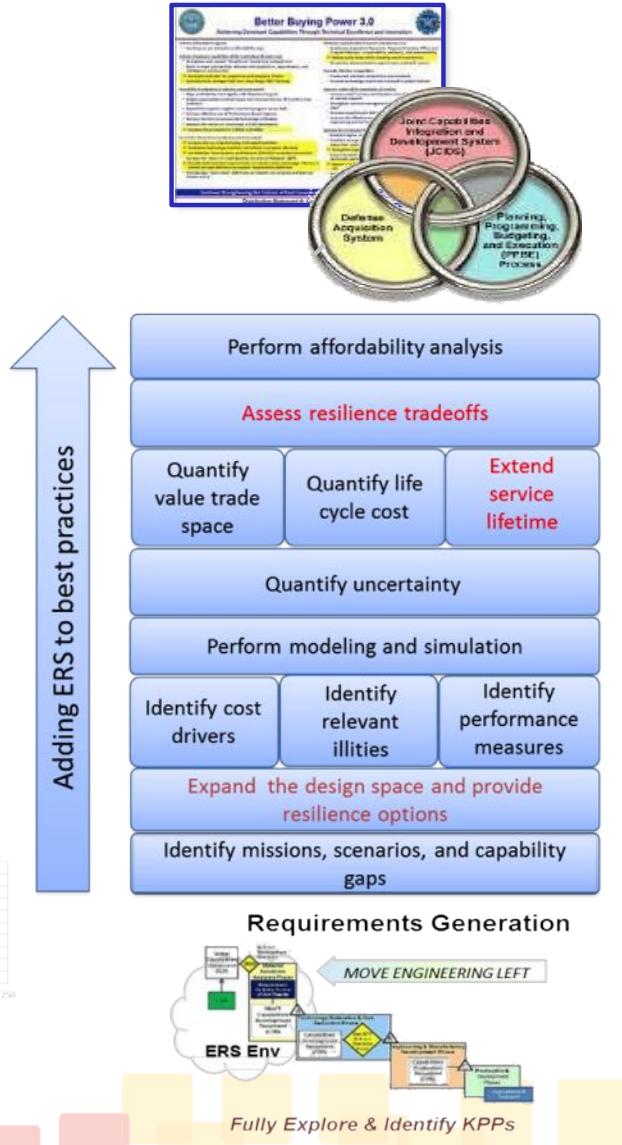
Model Based Systems Engineering



RAPID PROTOTYPING & RESPONSE
Virtual Warfighting, Reduce Prototyping Time & Costs



Set Based Design to expand the design space



Analysis of Alternatives

Reduces alternatives from thousands to tens or less



Tradespace Tools & Analytics

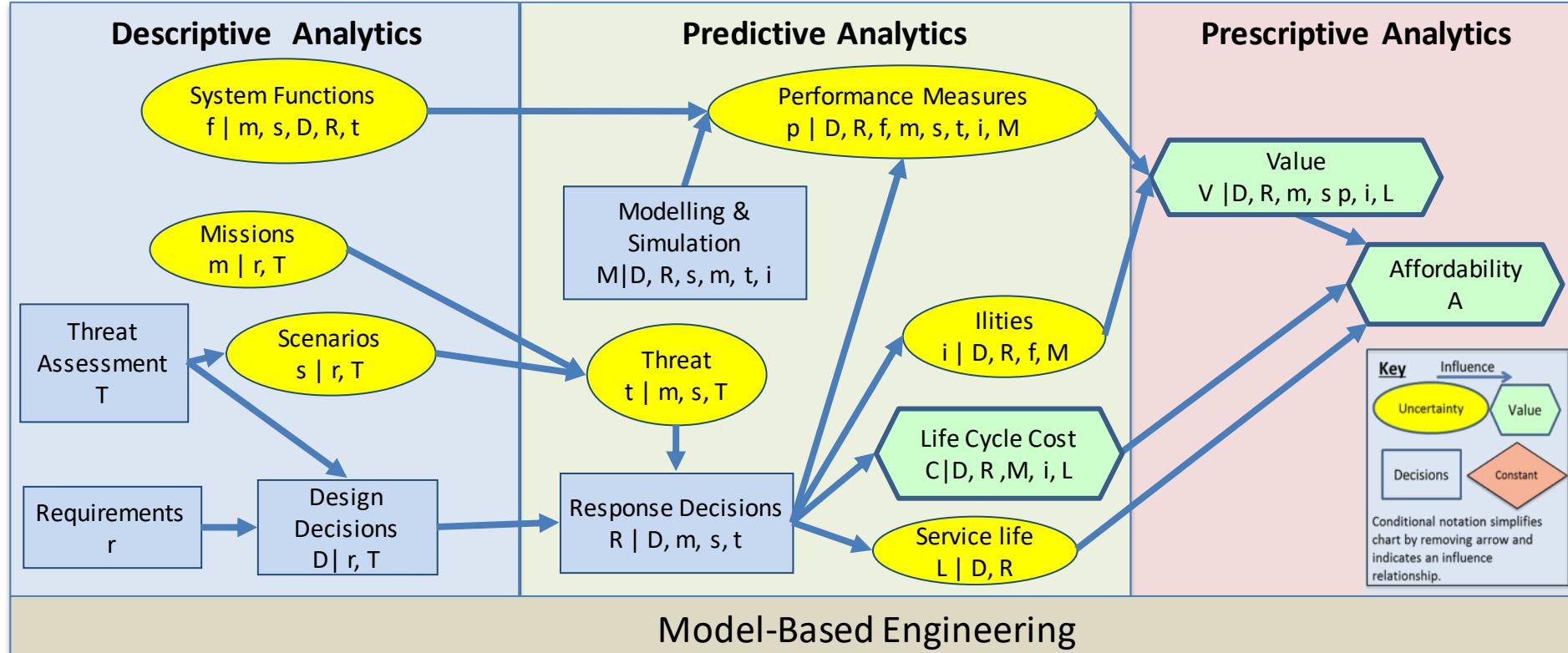


Integrated Capability and Workflow

Small, C., Parnell, G., Pohl, E., Goerger, S., Cottam, C., Specking, E., Wade, Z., (2018) Engineering Resilience for Complex Systems. In: Madni A., Boehm B., Ghanem R., Erwin D., Wheaton M. (eds) Disciplinary Convergence in Systems Engineering Research. Springer, Cham, pp. 3-15

The DoD and ERS seek to leverage the capabilities of model based engineering and set based design to improve decision making in AoAs.

Integrated Trade-off Analytics Framework

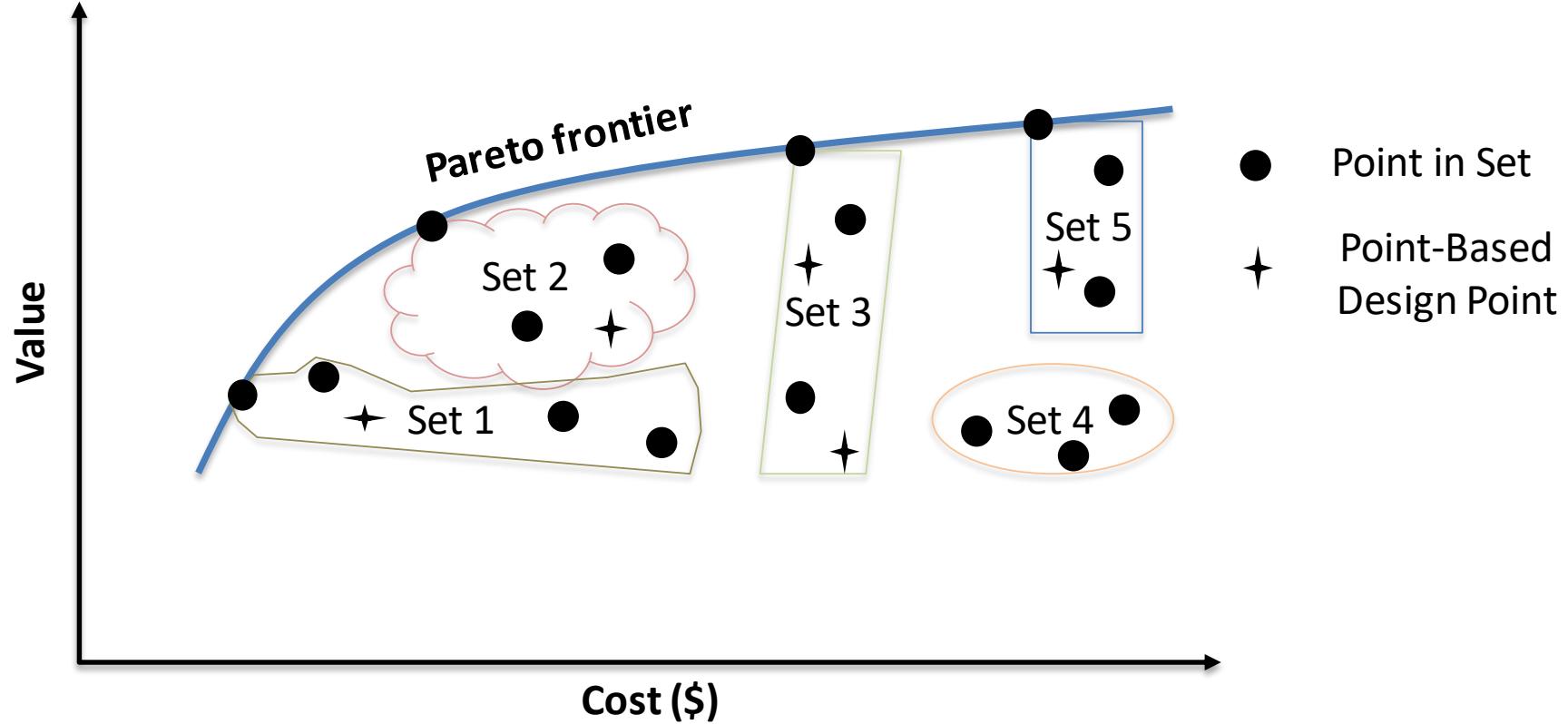


Modified from MacCalman, Alexander D., Gregory S. Parnell and Sam Savage. "An Integrated Model for Trade-off Analysis." Parnell, Gregory S. Trade-off Analytics: Creating and Exploring the System Tradespace. Wiley, 2016

Small, C., Parnell, G., Pohl, E., Goerger, S., Cottam, C., Specking, E., Wade, Z., (2018) Engineering Resilience for Complex Systems. In: Madni A., Boehm B., Ghanem R., Erwin D., Wheaton M. (eds) Disciplinary Convergence in Systems Engineering Research. Springer, Cham, pp. 3-15

Trade-off analytics requires descriptive, predictive, and prescriptive analytics using Model Based Engineering.

Set-Based Design



Wade, Z., Parnell, G., Goerger, S., Pohl, E., Specking, E. "Designing Engineered Resilient Systems Using Set-Based Design" 16th Annual Conference on Systems Engineering Research, Charlottesville, Virginia, May 8-9, 2018

Set-based design allows for further exploration of the design space over point-based design.

Set Drivers and Modifier



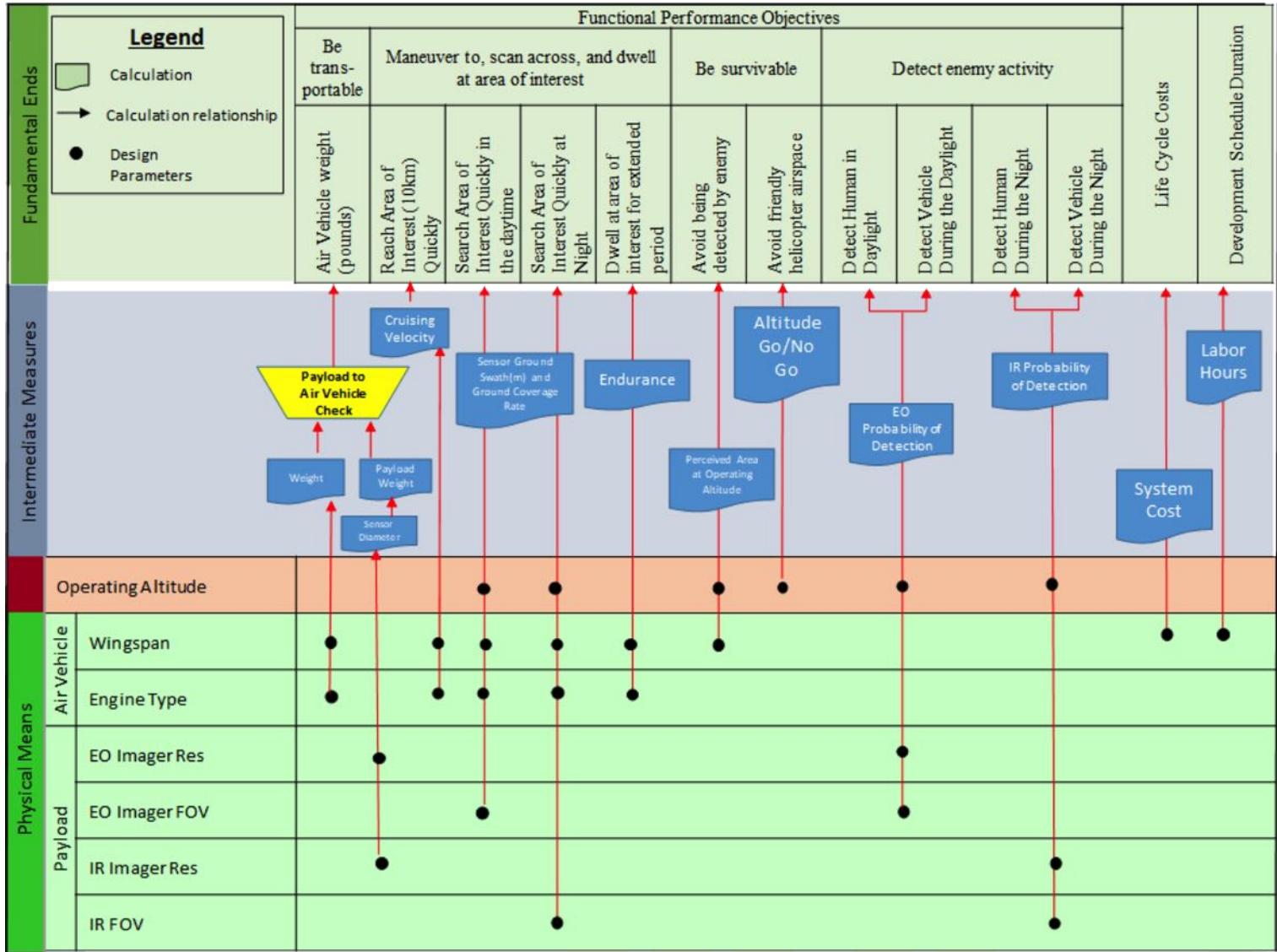
- In set based design, design decisions are split into two categories.
 - Set Drivers
 - Set Modifiers
- The sets in Set-Based design are determined by the set drivers.

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UAV Case Study



Cilli, Matthew. "Decision Framework Approach Using the Integrated Systems Engineering Decision Management (ISEDM) Process." *Model Center Engineering Workshop, Systems Engineering Research Center (SERC)*. 31 July 2017.

Using a UAV Case Study developed by Dr. Cilli at ARDEC, this research has applied the Trade-off Analytics Framework and set based design to the case study.

In the initial case study, 7 design decision were propagated through physics-based models to performance measures and cost.

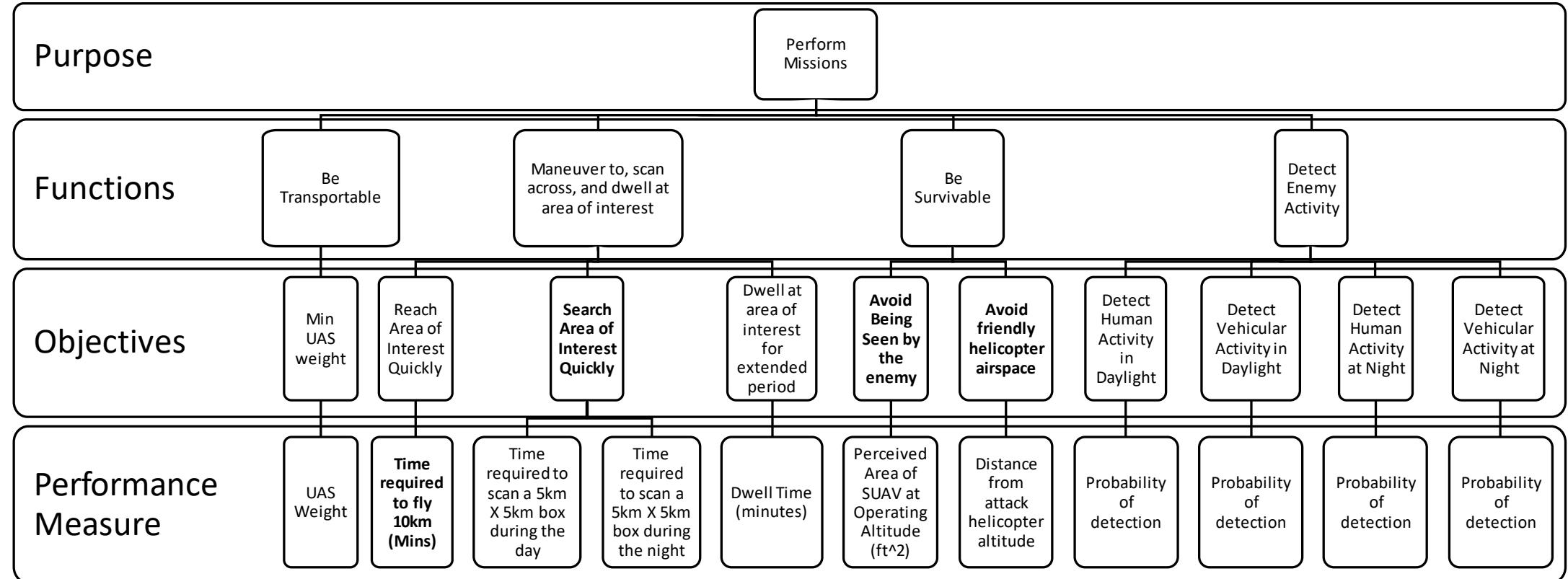
UAV Case Study Design Decisions



Design Choice	Options
Engine	Discrete Choice: <ul style="list-style-type: none"> Electric <ul style="list-style-type: none"> Piston
Wingspan	Continuous choice: <ul style="list-style-type: none"> 2 ft. to 12 ft.
Operating Altitude	Continuous choice: <ul style="list-style-type: none"> 300 m. to 1000 m.
Electro-Optical (EO) Sensor Resolution	Discrete Choice: <ul style="list-style-type: none"> 200 Pixels X 200 Pixels 400 Pixels X 400 Pixels 600 Pixels X 600 Pixels 800 Pixels X 800 Pixels <ul style="list-style-type: none"> 1000 Pixels X 1000 Pixels 1200 Pixels X 1200 Pixels 1400 Pixels X 1400 Pixels 1600 Pixels X 1600 Pixels 1800 Pixels X 1800 Pixels
EO Sensor Field of View	Discrete Choice: <ul style="list-style-type: none"> 15 Degrees 30 Degrees 45 Degrees <ul style="list-style-type: none"> 60 Degrees 75 Degrees 90 Degrees
Infrared (IR) Sensor Resolution	Discrete Choice: <ul style="list-style-type: none"> 200 Pixels X 200 Pixels 400 Pixels X 400 Pixels 600 Pixels X 600 Pixels 800 Pixels X 800 Pixels <ul style="list-style-type: none"> 1000 Pixels X 1000 Pixels 1200 Pixels X 1200 Pixels 1400 Pixels X 1400 Pixels 1600 Pixels X 1600 Pixels 1800 Pixels X 1800 Pixels
IR Sensor Field of View	Discrete Choice: <ul style="list-style-type: none"> 15 Degrees 30 Degrees 45 Degrees <ul style="list-style-type: none"> 60 Degrees 75 Degrees 90 Degrees

In the Case study there are seven design decisions propagated to value and cost.

Case Study Value Measures



In the Case study there are 4 functions with 11 performance measures.

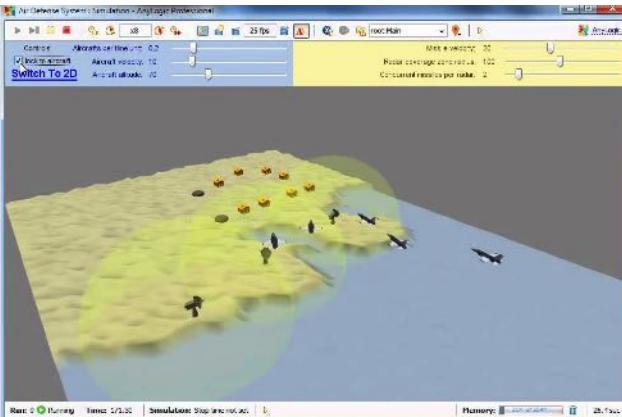
DoD Analysis of Alternatives



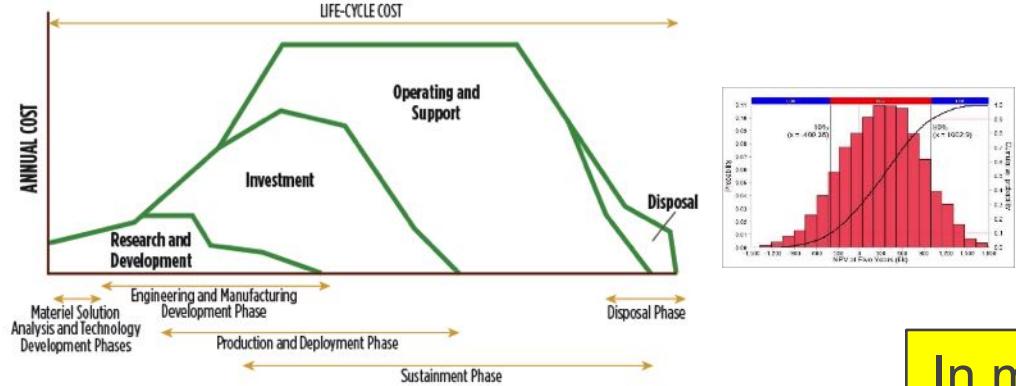
Point Design Concepts and Architectures

	Prototype/Deployed	Prototype
Over 30,000 lb	Panther w/SRS >40 tons Abrams Panther W/SRS >40 tons	DEUCE w/SRS 35,500 lb D7G w/SRS 55,500 lb AOE 67,000 lb
2,501 to 20,000 lb	Fielded ARTS 8000 lb DEMO II XUV 3,000 lb	Prototype Smoke HMMWV w/SRS 11,500 lb T3 Dozer w/SRS 18,600 lb
401 to 2,500 lb	Fielded SDD/ Deployed MRV -Flat/RCSS 2500 lb MDARS -I 1600 lb	SDD MDARS -E 1500 lb Prototype GLADIATOR 1600 lb
31 to 400 lb	Prototype/Deployed TALON 34 -80 lb URBOT 65 lb	MATILDA 40 lb BUGS 45-60 lb ODIS 45 lb

Capability and Military Value Models



Life Cycle Cost Model



Risk Analysis Models

Impact	Risk Management Actions		
	Considerable management required	Must manage and monitor risks	Extensive management essential
Moderate	Risks may be worth accepting with monitoring	Management effort worthwhile	Management effort required
	Accept risks	Accept, but monitor risks	Manage and monitor risks
	Low	Medium	High
Likelihood			

In many traditional AoAs, different groups such as cost analysts, capability analysts, risk analysts, or other groups perform the analysis on different areas of the AoA.

Case Study Changes



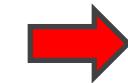
Iteration	Model	Descriptive	Predictive	Prescriptive
1	Initial Case Study			
2	Multiple Changes	The case study was redeveloped from the ground up and given new design choices.	An entirely new set of physics models was used. The only remaining model was the probability of detection.	A completely new value model and new cost model.
3	Design Choices	The set of design choices was expanded as new combinations of sensors were added.	None	None
4	Value Model	None	None	Preferences on value curves were changed. Changing preferences for alternatives.
5	Value Model	None	None	The value curves were changed once more to allow more feasible solutions.
6	Value Model	None	New calculation for distance to attack helicopter added for all alternatives.	A new value measure (distance from attack helicopter) was added.
7	Design Choices	New alternatives for sensor FOV were added and altitude options were reduced after a discussion with Dr. Ham.	None	None
8	Swing Weights	None	None	Swing weights were changed after a discussion with Dr. Ham.
9	Cost Model	None	Cost model was changed to a lifecycle cost model.	None

- Similar to real world AoAs, the case study changed several times
- However, the integrated and simultaneous MBE methodology is robust to changes.

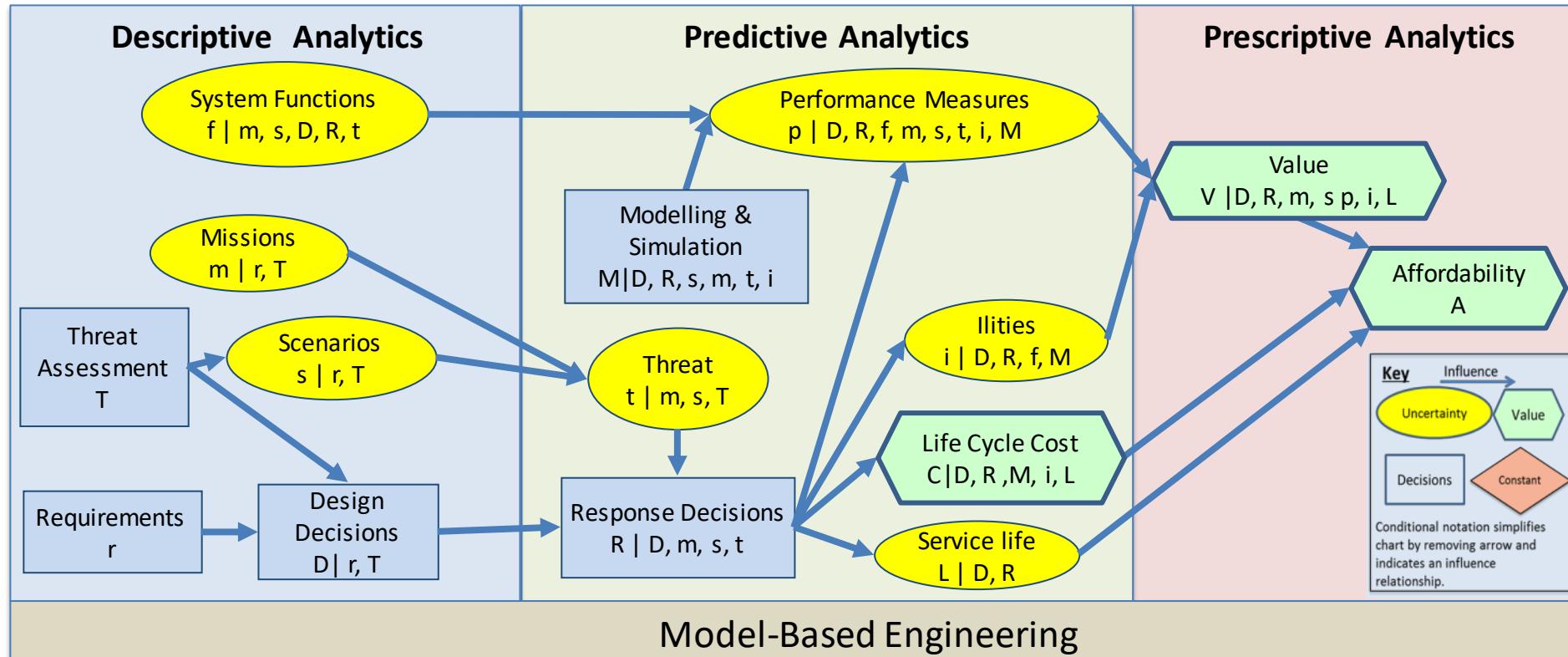
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Influence Diagram for Integrated Trade-off Analytics: Current Status

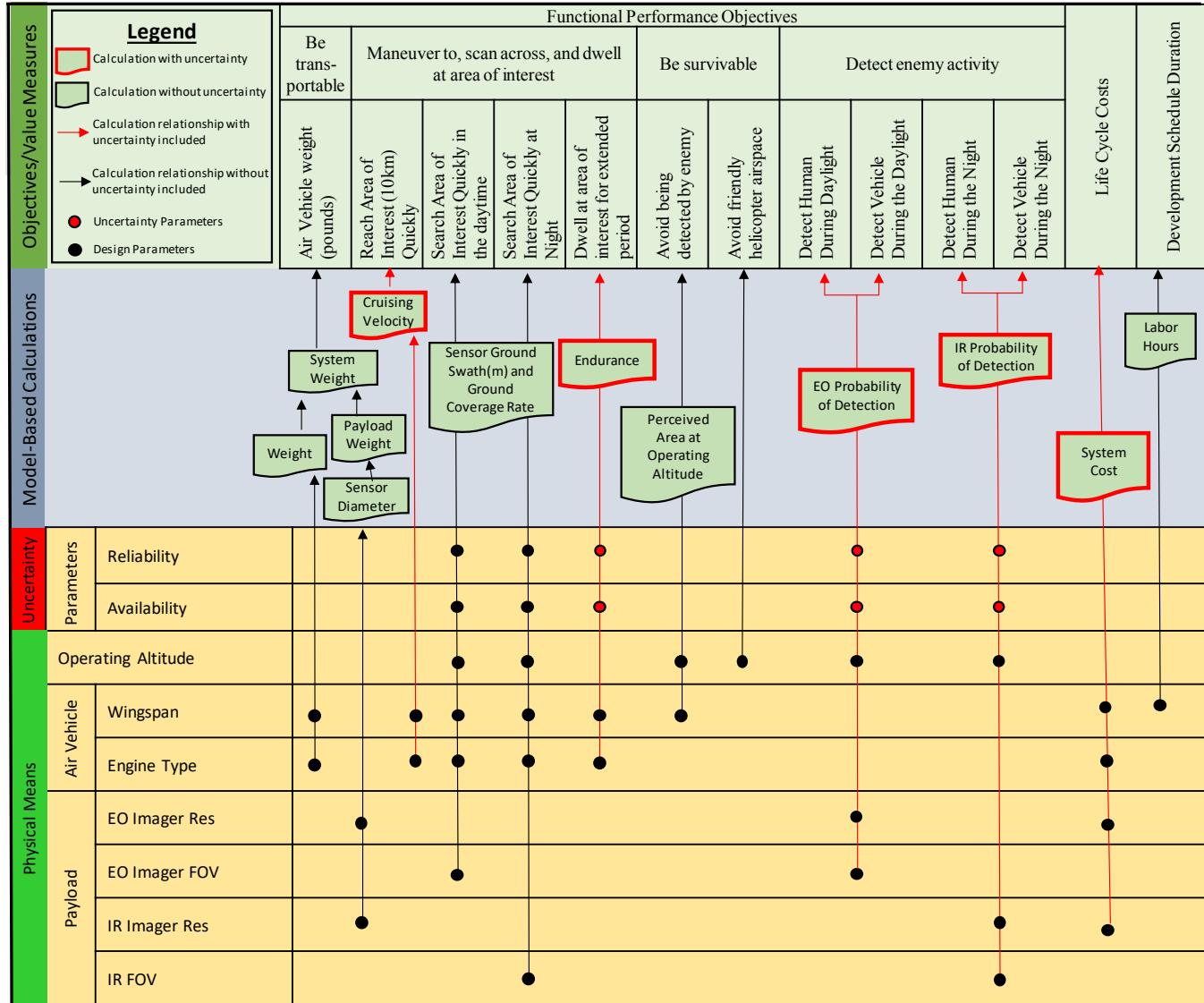


Modified from MacCalman, Alexander D., Gregory S. Parnell and Sam Savage. "An Integrated Model for Trade-off Analysis." Parnell, Gregory S. Trade-off Analytics: Creating and Exploring the System Tradespace. Wiley, 2016

Small, C., Parnell, G., Pohl, E., Goerger, S., Cottam, C., Specking, E., Wade, Z., (2018) Engineering Resilience for Complex Systems. In: Madni A., Boehm B., Ghanem R., Erwin D., Wheaton M. (eds) Disciplinary Convergence in Systems Engineering Research. Springer, Cham, pp. 3-15

We have expanded the case study and analysis to include the entire trade-off analytics framework.

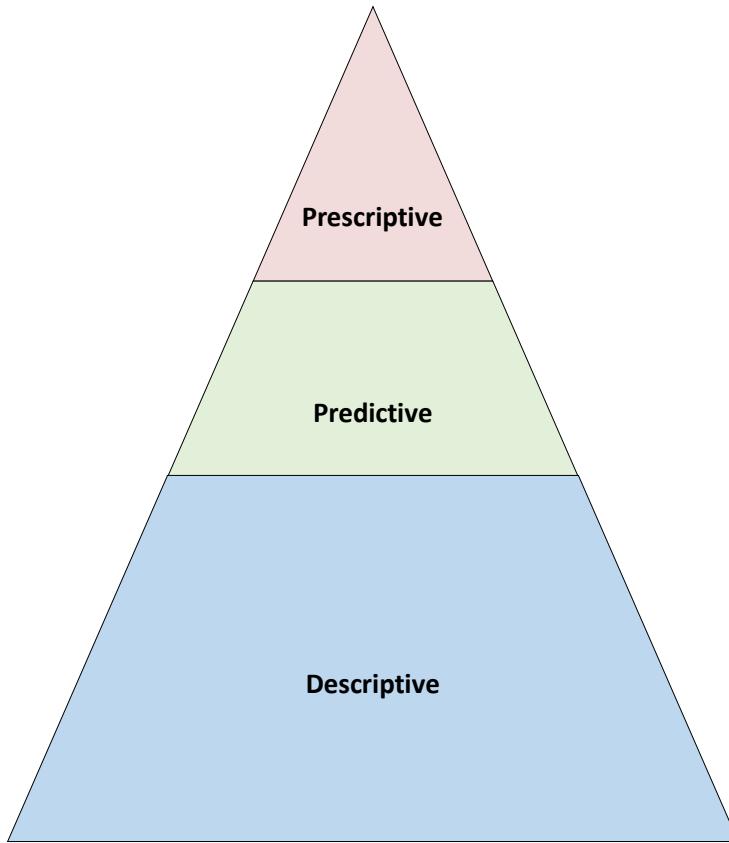
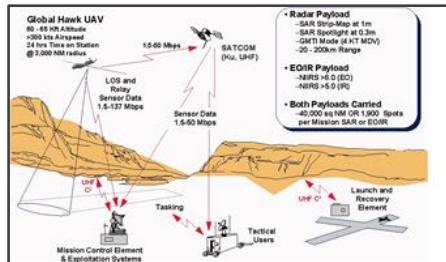
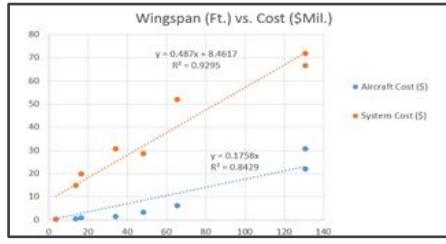
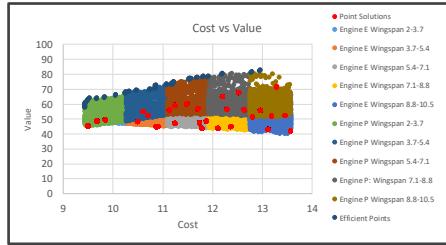
UAV Case Study Updated Assessment Flow Diagram



We have included uncertainty in the performance, cost, and preferences. In addition, we have included a life cycle cost model as well as included resilience in the performance calculations.

Small, C., Demonstrating Set-Based Design Techniques: A UAV Case study, Master's Thesis, Industrial Engineering, University of Arkansas, 2018

Trade-Off Analytics Hierarchy



Integrated Value and Cost Model	
Multiple Objective Decision Analysis	
2,576	Feasible Cost vs Value points
97,424	Total infeasible designs
95,549	Infeasible designs with stochastic parameters
1,874	Infeasible designs with deterministic parameters

Predicted design performance and costs	
21,900,000	Physics model calculations
98,070	Designs with stochastic parameters
1,930	Designs with deterministic parameters
1,100,000	Value measure estimates
100,000	Cost estimates

Design definition and uncertainty specification	
7	Design Parameters
145,800	Combinations of design parameters using bins
100,000	Designs generated by SIPmath
47	Physics models and formulas
19	Physics models with uncertainty
4	Illities
2	Illities with Uncertainty
11	Value Measures
8	Value measures with uncertainty

Using Excel and an Excel add-in called SIPmath from Probability Management, this research has created an integrated tradespace tool to apply the analytics framework and explore set-based design.

Small, C., Buchanan, R., Cilli, M., Parnell, G., Pohl, E., Wade, Z., "A UAV Case Study with Set-based Design," 28th Annual INCOSE International Symposium, 7-12 July 2018, Washington, DC.

Analytics hierarchy provides transparency into the complexity of the trade-off analytics

UAV Tradespace Tool



UAV Integrated Set-Based Design Tradespace Tool
Research sponsored by ERDC ERS program and data provided by ARDEC (Dr. Matthew Cilli and his UAV team)

ERDC ERS **ARDEC ARMAMENTS** **Probability Management** **UNIVERSITY OF ARKANSAS**

Air Vehicle

Wingspan	Engine Type	Operating Altitude
Wingspan must be between 2 and 12	Engine Type must be either E or P	Flying altitude must be between 300 and 1000 M
9	P	Operating Altitude 565

Service Life 5 years

Payload

EO Imager			IR Sensor		
EO Sensor Pixel Width Choice:	Horizontal Pixels	Vertical Pixels	EO Sensor Pixel FOV Choice:	Field of View	IR Sensor Pixels Choice:
1	200	200	1	15	1
2	400	400	2	30	2
3	600	600	3	45	3
4	800	800	4	60	4
5	1000	1000	5	75	5
6	1200	1200	6	90	6
7	1400	1400	7	1400	7
8	1600	1600	8	1600	8
9	1800	1800	9	1800	9
4	800	800	6	90	6

Value Calculations

Value Measure	Weighted Value Score
UAS Weight	4
Time required to fly 10km (Mins)	8
Time Required to scan day	8
Time Required to scan night	8
Dwell Time (Mins)	7
Perceived Area of SUAV at Altitude	3
Difference from attack helicopter altitude	1
Probability of detecting a human day	3
Probability of detecting a vehicle day	7
Probability of detecting a human night	0
Probability of detecting a vehicle night	7
Total Value	57

Cost Analysis

Initial Cost of UAVs	\$9,260
Unit Manpower Cost	\$6,250
Unit Operations Cost	\$7,257
Maintenance Cost	\$4,176
Sustaining Support Cost	\$2,205
Indirect Support Cost	\$6,913
Total Cost in millions	\$142.740

Cost Uncertainty

Uncertainty included in Cost?	TRUE
Include Deterministic	TRUE
Measure	Percent Variation Allowed
Initial Cost of UAVs	5% -0.01
Unit Manpower Cost	5% -0.02
Unit Operations Cost	5% 0.01
Maintenance Cost	5% -0.02
Sustaining Support Cost	5% 0.00
Indirect Support Cost	5% -0.02

Performance Uncertainty

Uncertainty included in model	TRUE
Include Deterministic	TRUE
Uncertainty in Ities	Uncertainty in Performance Models is based on a normal distribution
Ability	Minimum: 0.9, Most Likely: 0.95, Best: 0.97
Availability	95% Endurance: (0.69)
Reliability	0.92, 0.95, 0.97
Performance Model Deviations	0.039
Cruising Velocity	(0.39)

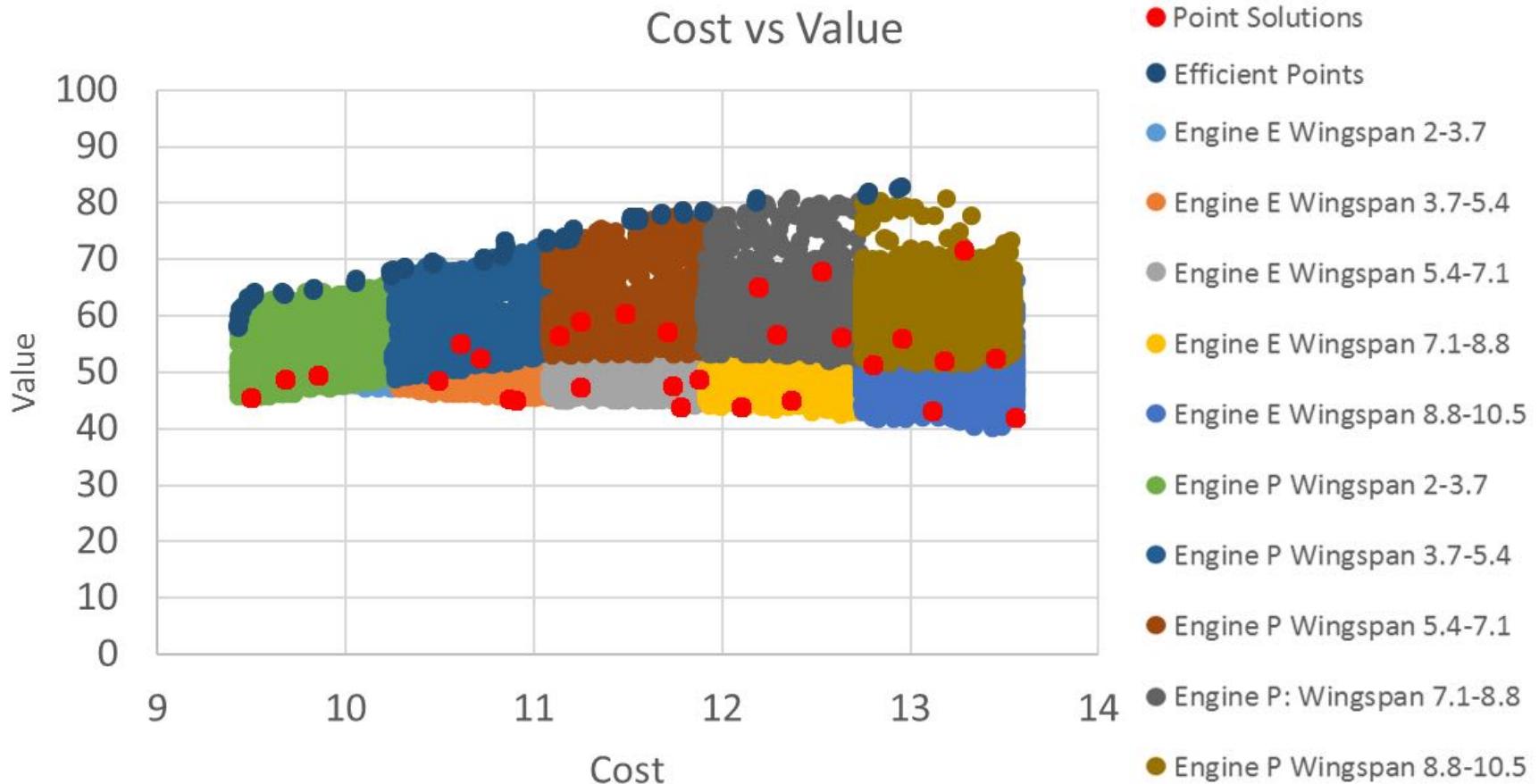
Perfect Options

Allow Perfect Options	FALSE
Perfect Option Allowed?	Used?
Perfectly Available Sensors	FALSE, FALSE
Perfectly Reliable Sensors	FALSE, FALSE
Perfectly Recoverable Options	FALSE, FALSE
Perfectly Available Options	FALSE, FALSE
Perfectly Restorable Options	FALSE, FALSE
Perfectly Detecting Sensors	TRUE, FALSE
Allow Perfect Options?	FALSE
All Perfect Options?	FALSE

Perfect Detection

Using random numbers generated by SIPmath, the tradespace tool uniformly explores the entire design space. In addition, the control panel allows the user to select the level of uncertainty on performance, cost, and preferences.

PBD vs. SBD

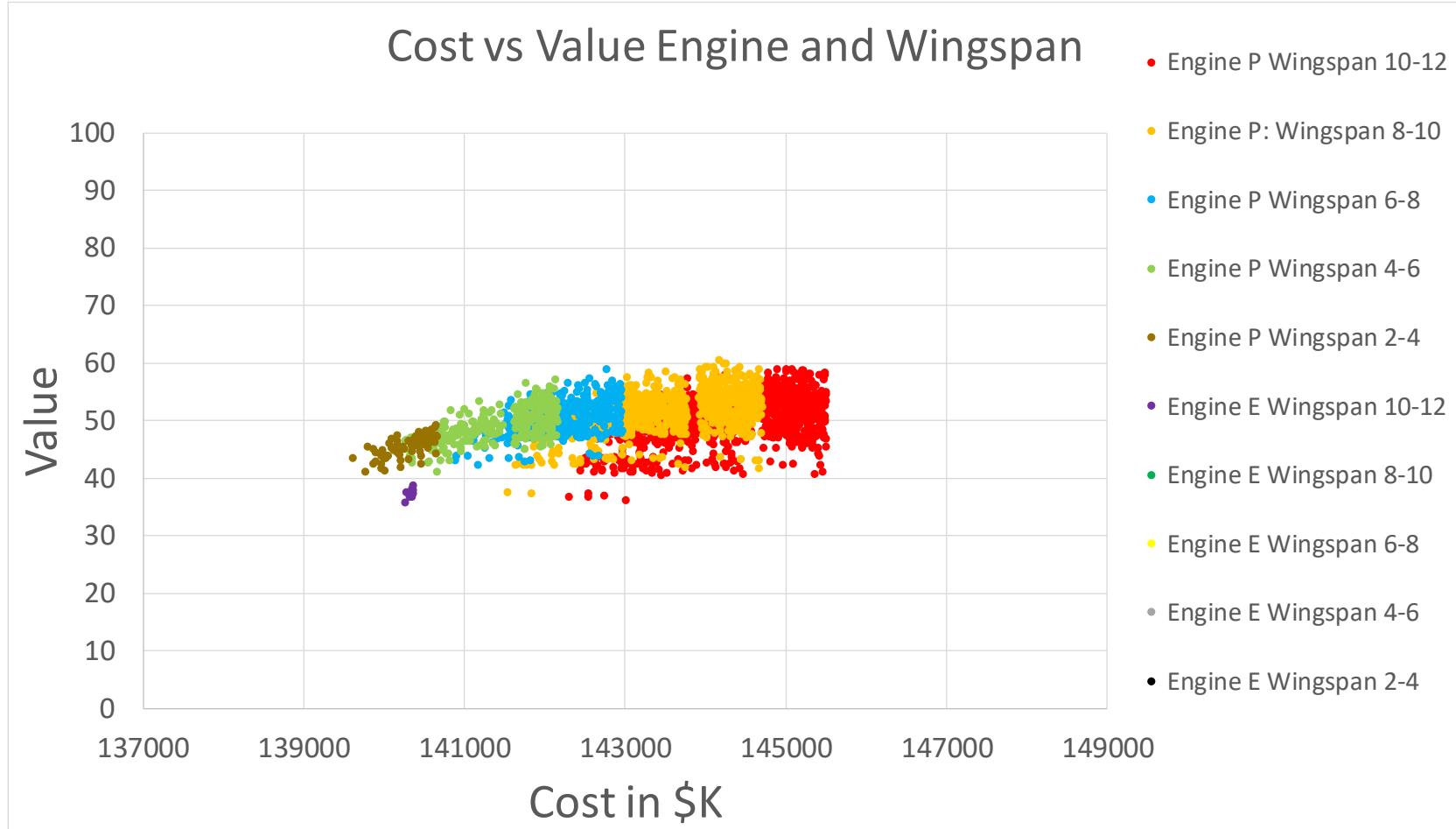


Sets are determined by engine type and wingspan. Deterministic analysis shows the value vs cost for the 10 sets.

Small, C., Buchanan, R., Cilli, M., Parnell, G., Pohl, E., Wade, Z., "A UAV Case Study with Set-based Design," 28th Annual INCOSE International Symposium, 7-12 July 2018, Washington, DC.

In iteration 2 of the case study, set-based design identified improved solutions compared to a finite number of point solutions.

Value Vs Cost



Throughout the changes, sets are still determined by engine type and wingspan. Deterministic analysis shows the value vs cost for the 10 sets.

Life Cycle Cost Model



Hardware Cost

$$\text{Air Vehicle Recurring Unit Cost } (\$/K 2013) = \text{FlyWeight} * 1.002$$

$$\text{Air Frame Unit Recurring Cost } (\$/K 2013) = \text{PayloadWeight} * 5.607$$

$$\text{Propulsion Unit Recurring Cost } (\$/K 2013) = (\text{FlyWeight} - \text{PayloadWeight}) * 1.808$$

$$\text{Payload Average Unit Cost } (\$/K 2013) = 0.5 * \text{AirFrameUnitCost}$$

$$\text{Total Hardware Cost} (\$/K 2013) = \text{TotalGroundStation} + \text{AirVehicleUnitCost} + \text{PayloadAverageCost} + \text{PropulsionUnitCost} + \text{AirFrameUnitCost}$$

Support Costs

$$\text{Unit Level Manpower Cost } (\$/K 2013) = 250 * 0.5 * \text{NumberOfSystems}$$

$$\text{Unit Operations Cost } (\$/K 2013) = (24676 + 0.8286 * 1156 * \text{TotalAirCraftInventory}) * 1/10$$

$$\text{Maintenance Cost } (\$/K 2013) = \left(\frac{41223 + 0.1261 * \text{AirElementsWeight} * 1}{\text{AgeOfAircraft} * \text{TotalAircraftInventory}} \right) * \frac{1}{10}$$

$$\text{Sustaining Support Cost } (\$/K 2013) = \text{TotalHours}^{0.7303} * \text{NumberOfSystems}$$

$$\text{Indirect Support Cost } (\$/K 2013) = 2777 * e^{(0.01824 * \text{NumberOfSystems})}$$

Life Cycle Cost

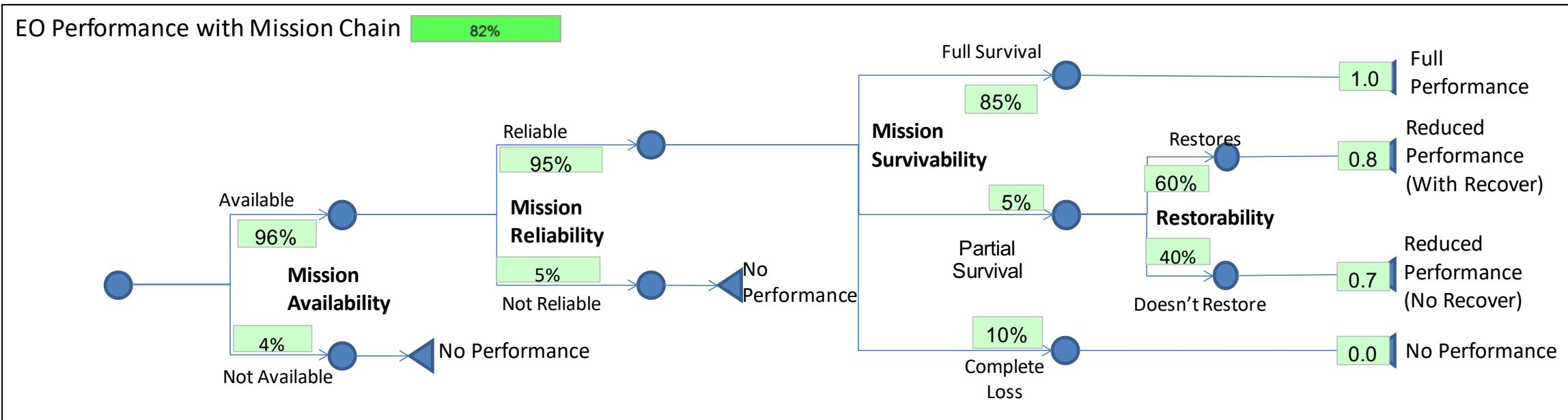
$$\text{Life Cycle Cost } (\$/K 2013)$$

$$= \text{TotalHardwareCost} * \text{NumberOfSystems}$$

$$+ (\text{Unit Level Manpower Costs} + \text{Unit Operations Costs} + \text{Maintenance Cost} + \text{Sustaining Support Cost} + \text{Indirect Support Cost}) * \text{Service Life}$$

To explore the lifecycle of the we have expanded the cost model to a lifecycle cost model.

Incorporating Mission Resilience in Performance Measures



Small, C., Buchanan, R., Cilli, M., Parnell, G., Pohl, E., Wade, Z., "A UAV Case Study with Set-based Design," 28th Annual INCOSE International Symposium, 7-12 July 2018, Washington, DC.

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Using probability trees, we have incorporated mission resilience into endurance and probability of detection performance calculations.

Perfect Options



Analyzing the effect of perfect ilities provides insight into resilience response decisions.

Uncertainty

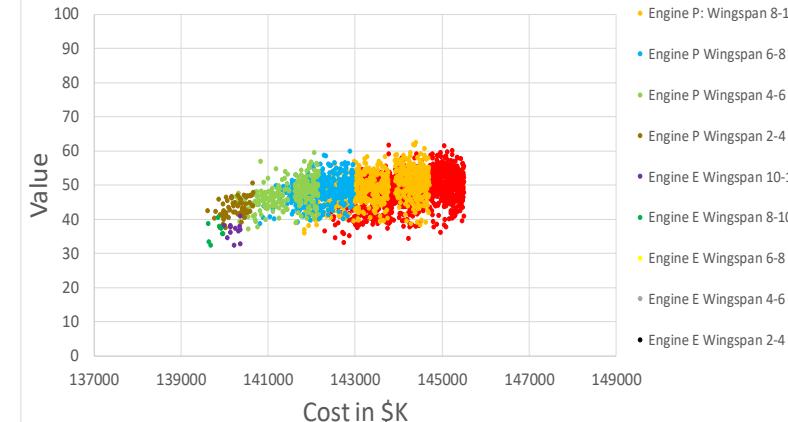


- In this research uncertainty has been incorporated into three areas:
 - Performance
 - Through physics based models
 - Through theilities
 - Cost model
 - In each of the cost types
 - Preferences
 - In the un-normalized elicited swing weights

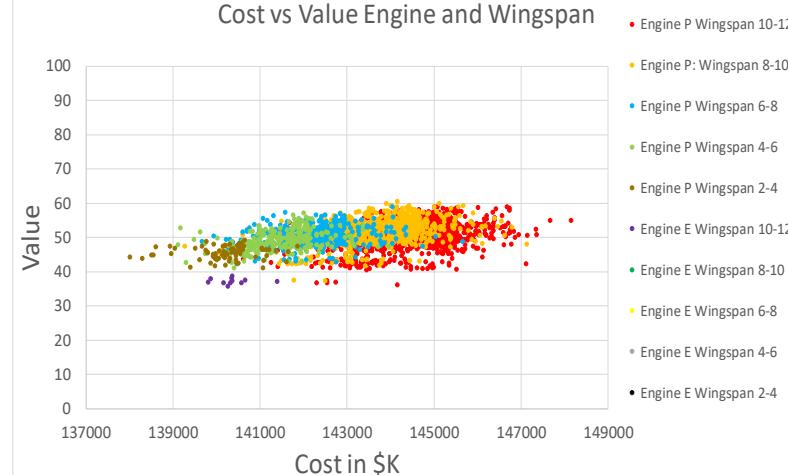
Uncertainty



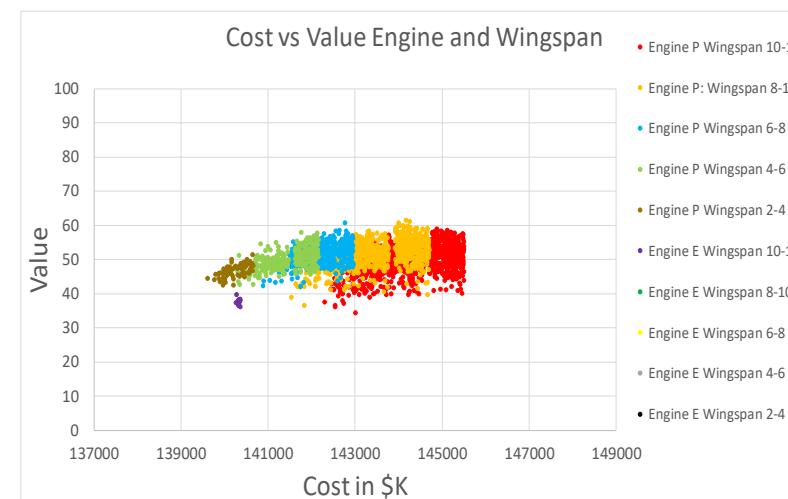
Cost vs Value Engine and Wingspan



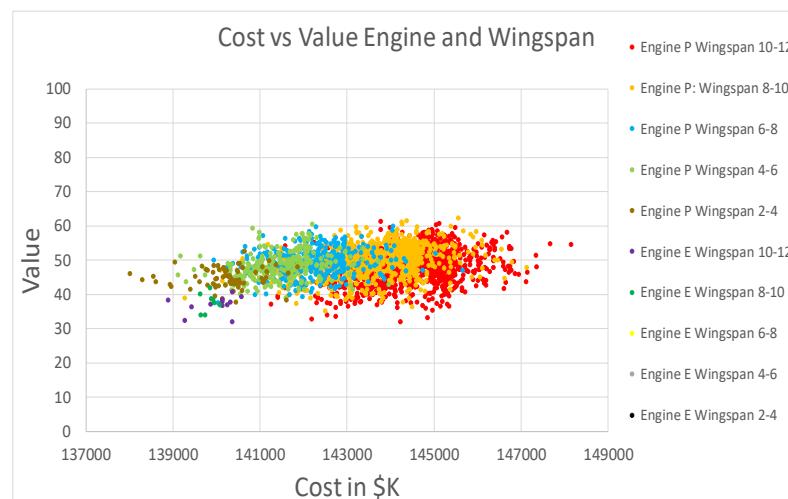
Cost vs Value Engine and Wingspan



Cost Uncertainty



Performance Uncertainty



Preference Uncertainty

Small, C., Demonstrating Set-Based Design Techniques: A UAV Case study, Master's Thesis, Industrial Engineering, University of Arkansas, 2018

All Uncertainty

- Overall incorporating uncertainty increases the overlap in sets, making them harder to distinguish like real world systems that can have wide ranges of uncertain performance and cost.
- Some sets are impacted by different types of uncertainty more than others

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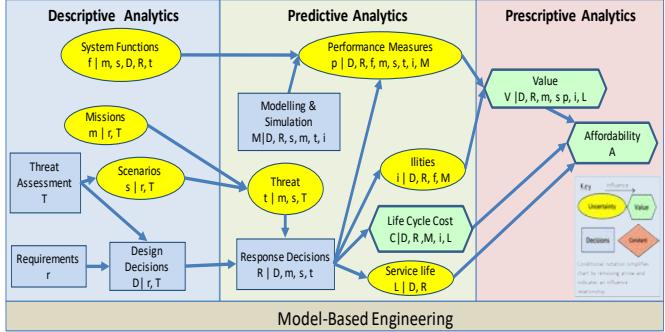
Contributions



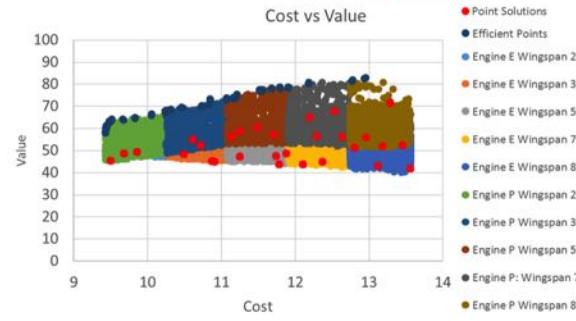
Incorporating ERS into AoAs



Trade-off Analytics Framework

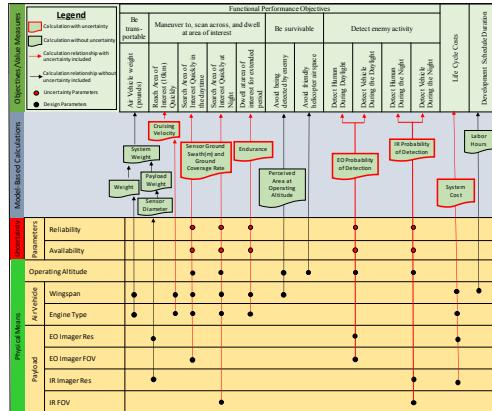


PBD vs. SBD



Due in large part to this research, the creator of the UAV case study has begun to incorporate major portions of this methodology and Set-Based Design within his systems engineering trade-off analysis for the ARDEC. (Cilli, 2018)

Implemented Trade-off analytics framework and SBD



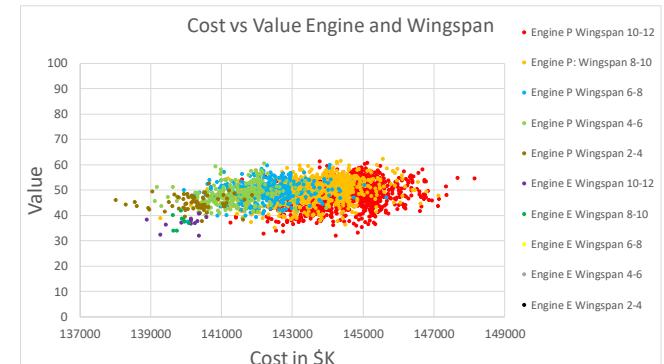
Methodology is robust to changes

Design Choice	Options
Engine	Discrete Choice: • Electric Continuous choice: • 2 ft. to 12 ft.
Wingspan	Continuous choice: • 300 m. to 1000 m.
Operating Altitude	Discrete Choice: • 1000 Pixels X 1000 Pixels • 1200 Pixels X 1200 Pixels • 1400 Pixels X 1400 Pixels • 1600 Pixels X 1600 Pixels • 1800 Pixels X 1800 Pixels
Electro-Optical (EO) Sensor Resolution	Discrete Choice: • 200 Pixels X 200 Pixels • 400 Pixels X 400 Pixels • 600 Pixels X 600 Pixels • 800 Pixels X 800 Pixels
EO Sensor Field of View	Discrete Choice: • 15 Degrees • 30 Degrees • 45 Degrees • 60 Degrees • 75 Degrees • 90 Degrees
Infrared (IR) Sensor Resolution	Discrete Choice: • 200 Pixels X 200 Pixels • 400 Pixels X 400 Pixels • 600 Pixels X 600 Pixels • 800 Pixels X 800 Pixels
IR Sensor Field of View	Discrete Choice: • 15 Degrees • 30 Degrees • 45 Degrees • 60 Degrees • 75 Degrees • 90 Degrees

Perfect Options



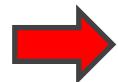
Effects of uncertainty on SBD



Overview



- Introduction
 - ERS
 - Trade-off Analytics Framework
 - Set Based Design
- Initial UAV Case Study
- UAV Tradespace Tool
 - Base model
 - Perfect Options
 - Uncertainty
- Conclusions
- Future Research



Future Work



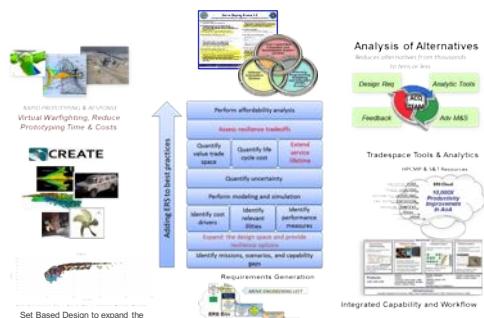
- There are four major areas for future research:
 - In this study the sets drivers were determined by using a heuristic method by looking at the impact of design decisions on the cost and value tradespace. To increase the feasibility of set-based design methodology, a repeatable, mathematical method of defining set-drivers needs to be developed.
 - The ability to generate the efficient frontier needs to be validated through comparison with genetic algorithms.
 - The resilience options research needs to be expanded to include explicit resilience options as well as the cost of resilience options.
 - In support of the ERS research effort at ERDC this tradespace tool will be implemented in an online trade-off analytics tool (TradeBuilder).

Questions

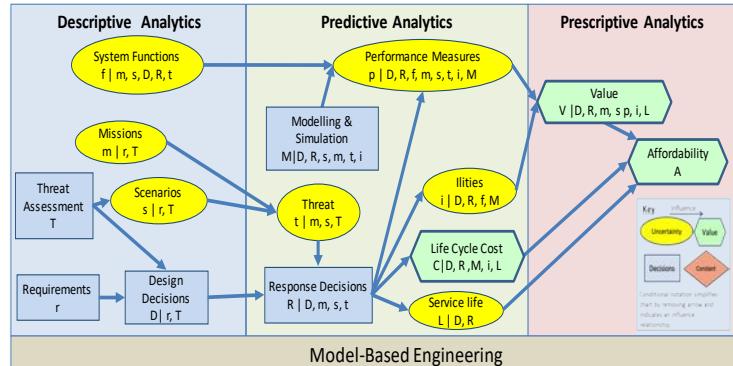


My thesis provides a case study using ERS/ARDEC UAV data that demonstrates the potential of Set-Based Design trade-off analytics in system decision making for Engineered Resilient Systems.

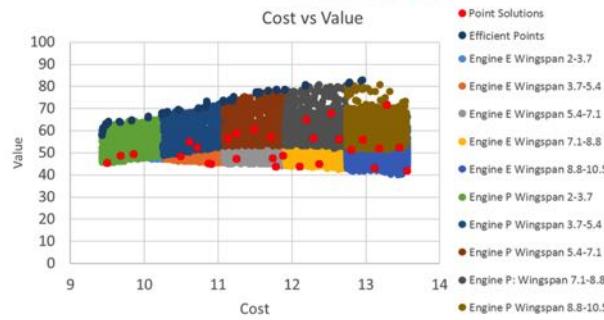
Incorporating ERS into AoAs



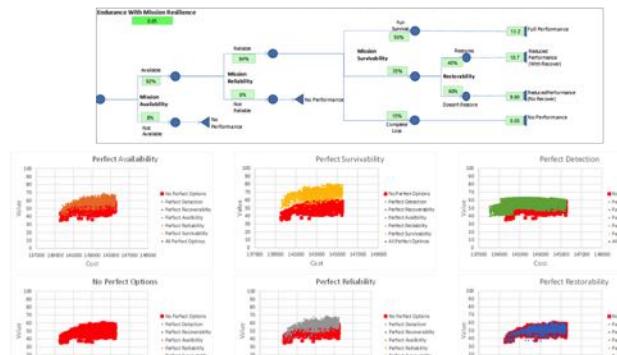
Trade-off Analytics Framework



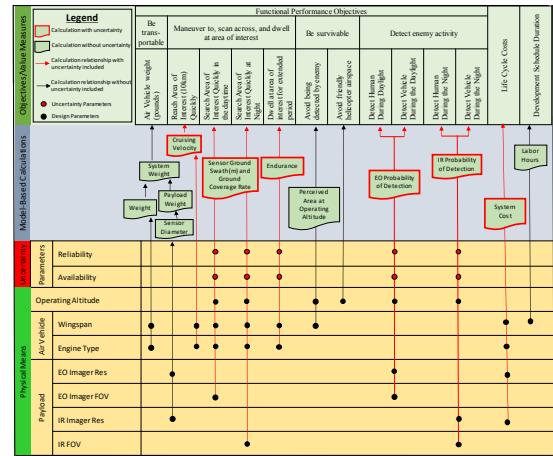
PBD vs. SBD



Perfect Options



Implementing Trade-off analytics framework and SBD in UAV Tradespace Tool



Effects of uncertainty on SBD

