



33rd Annual **INCOSSE**
international symposium

hybrid event

Honolulu, HI, USA
July 15 - 20, 2023



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German Aerospace Center (DLR), Hamburg, Germany

The AGILE4.0 Project: MBSE to Support Cyber-Physical Collaborative Aircraft Development

15-20 July - 2023

www.incose.org/symp2023 #INCOSSEIS

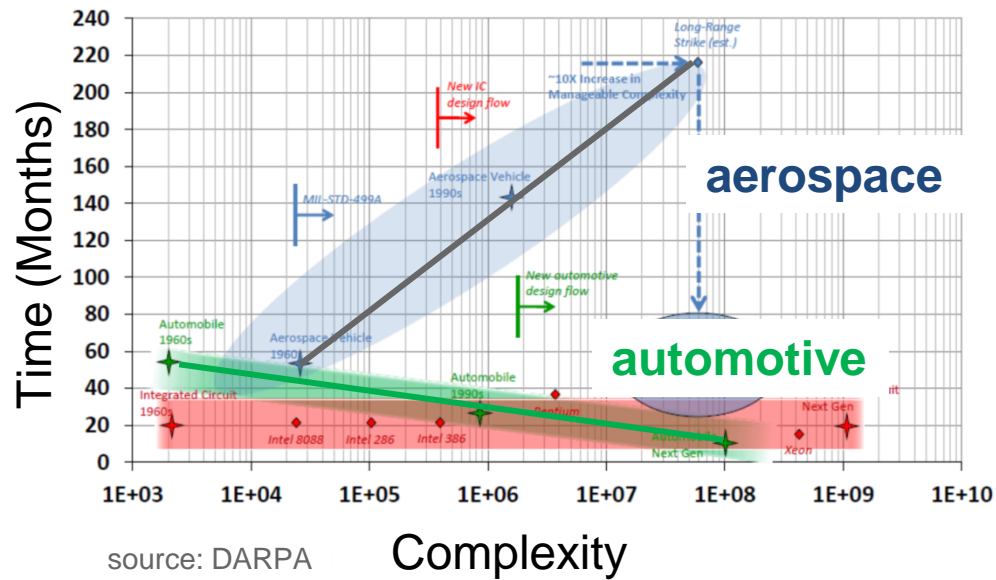


Development Challenges for Complex Aerospace Systems

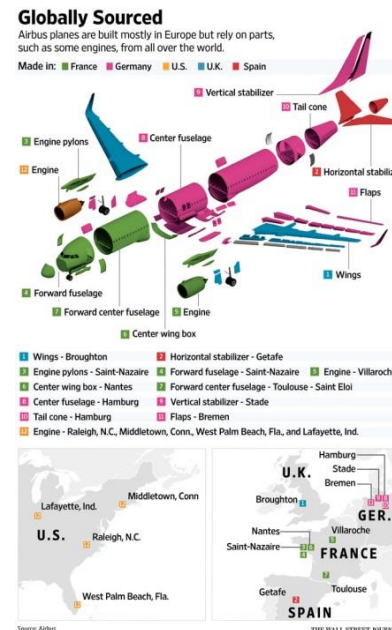
- Increasing **complexity** and development **time** → **integration** of large number of **parts** and **sub-processes**
- **Cross-organizational** development → distributed and **heterogeneous** knowledge and expertise
- Increasing **interactions** → systems operating in **system of systems** scenarios

Target: Managing complexity and accelerate the development time

Increasing development times



Distributed supply chain



source: Airbus

System of Systems



source: DLR

AGILE4.0

Towards cyber-physical collaborative aircraft development

AGILE 4.0 project logo and a small table of project milestones.

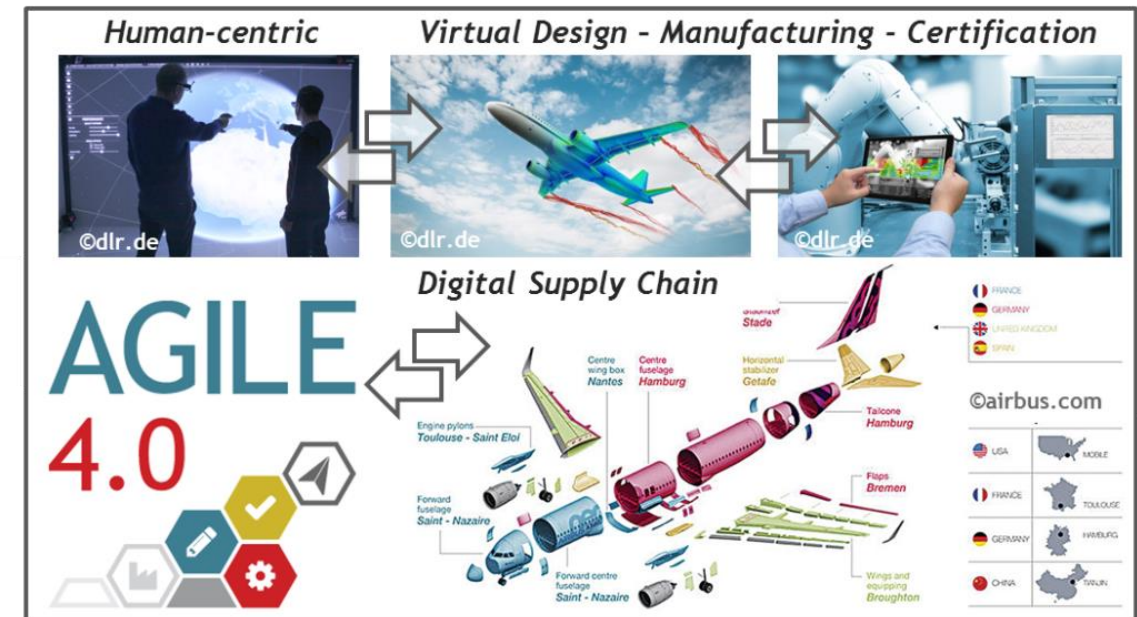
AGILE 4.0 project ambition:

“The high-level objective of AGILE4.0 is to bring significant reductions in aircraft development costs and time-to-market through the implementation of an integrated cyber-physical **aeronautical supply chain**, thereby increasing the competitiveness of the European aircraft industry, from integrators and high-tiers suppliers to SMEs, leading to **innovative and more sustainable aircraft products**”



EU funded H2020 project: **September 2019 – February 2023**

- 16 International Partners (EU, Brazil, Canada)
- Coordinated by DLR Hamburg

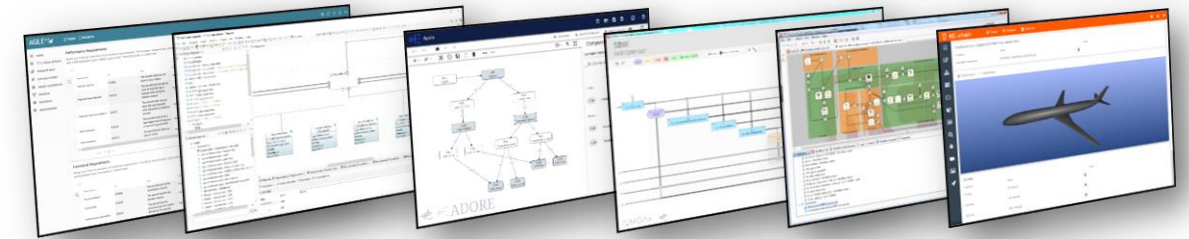
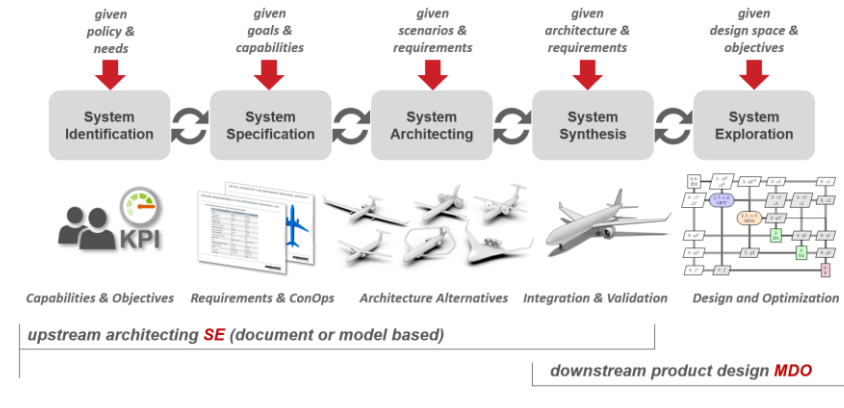


Towards the next generation **MBSE-MDO** accelerating the development of complex systems

Contents

1. AGILE 4.0 MBSE-MDAO Development Framework

- Upstream architecting phases
- Downstream product design phases



2. Implementation in a collaborative environment

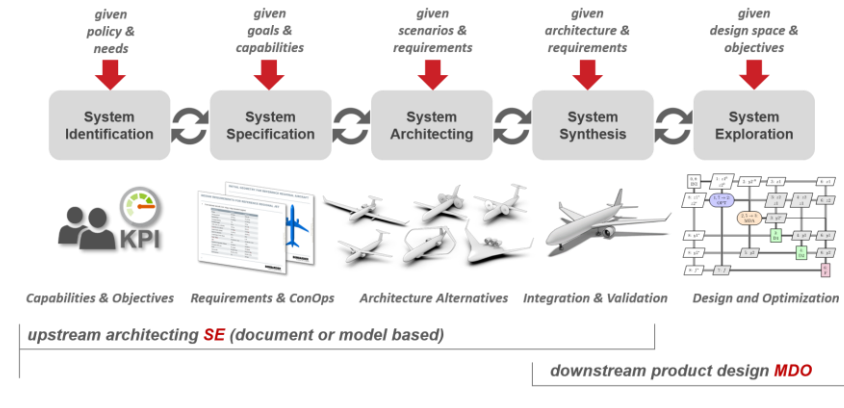
3. Demonstration: Design of a Business Jet Family



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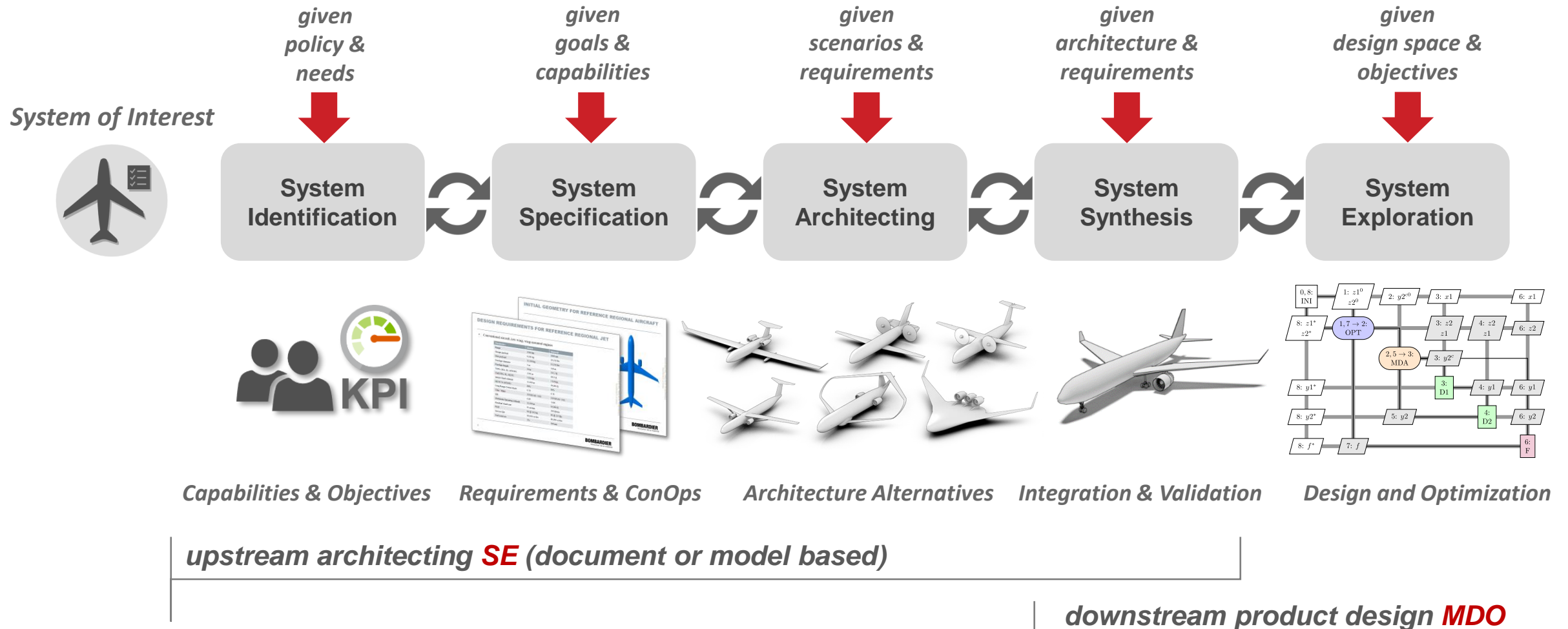
1. AGILE 4.0 MBSE-MDAO Development Framework

- Upstream architecting phases
- Downstream product design phases



3. Demonstration: Design of a Business Jet Family

AGILE4.0 MBSE-MDAO Development Framework



Requirement Types & Structure

Statements in natural language → Patterns: each type of requirement has a standard grammar, a set of mandatory and [optional elements] that ensure verifiability related to the type (*Carson, 2015*)

Examples:

- **Functional**
The **SYSTEM** shall [exhibit] **FUNCTION** [while in **CONDITION**]
- **Performance**
The **SYSTEM** shall **FUNCTION** with **PERFORMANCE** [and **TIMING** upon **EVENT TRIGGER**] while in **CONDITION**
- **Design (constraint)**
The **SYSTEM** shall [exhibit] **DESIGN CONSTRAINTS** [in accordance with **PERFORMANCE** while in **CONDITION**]
- **Environmental**
The **SYSTEM** shall [exhibit] **CHARACTERISTIC** during/after exposure to **ENVIRONMENT** [for **EXPOSURE DURATION**]
- **Suitability**
The **SYSTEM** shall exhibit **CHARACTERISTIC** with **PERFORMANCE** while **CONDITION** [for **CONDITION DURATION**]

The **aircraft** shall **provide propulsive power** [**during the entire mission**]

The **aircraft** shall **fly** at minimum Mach 0.8 **during cruise**

The **aircraft** shall have **technologies with maturity TRL 9** [**in 2020**]

The **aircraft** shall **be maneuverable** in case of icing conditions [**for the entire flight**]

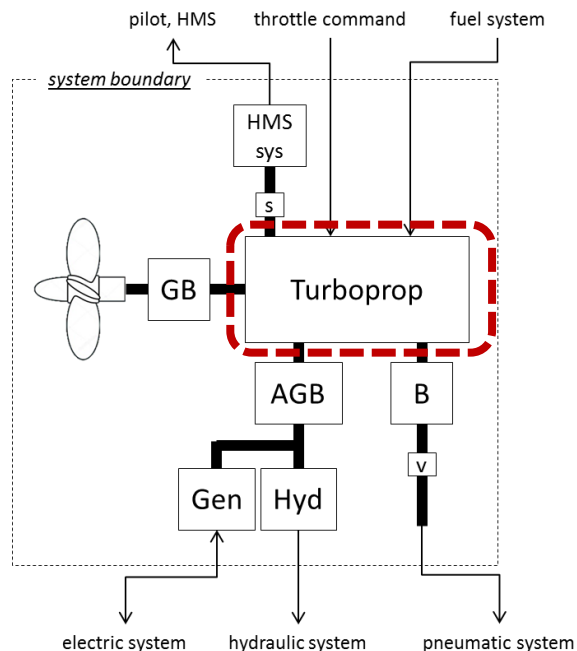
The **aircraft** shall have a **steady gradient of climb** of minimum 2.4% in case of **one-engine-inoperative**

Source: Boggero et al. "An MBSE Architectural Framework for the Agile Definition of System Stakeholders, Needs and Requirements", AIAA 2021

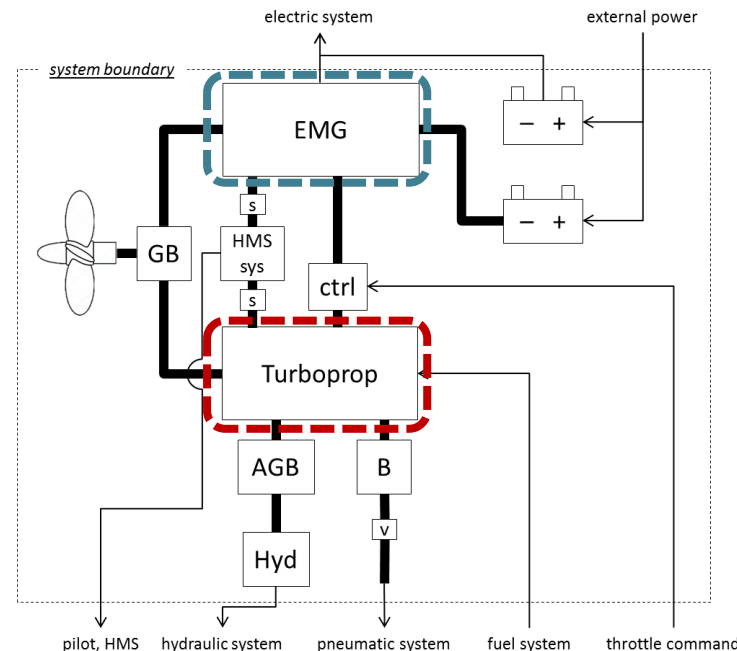
Generation and Modeling of System Architectures

System Architecture = formal **description** of a system. Representation of entities [...] that describes behaviors and relationships amongst the entities¹.

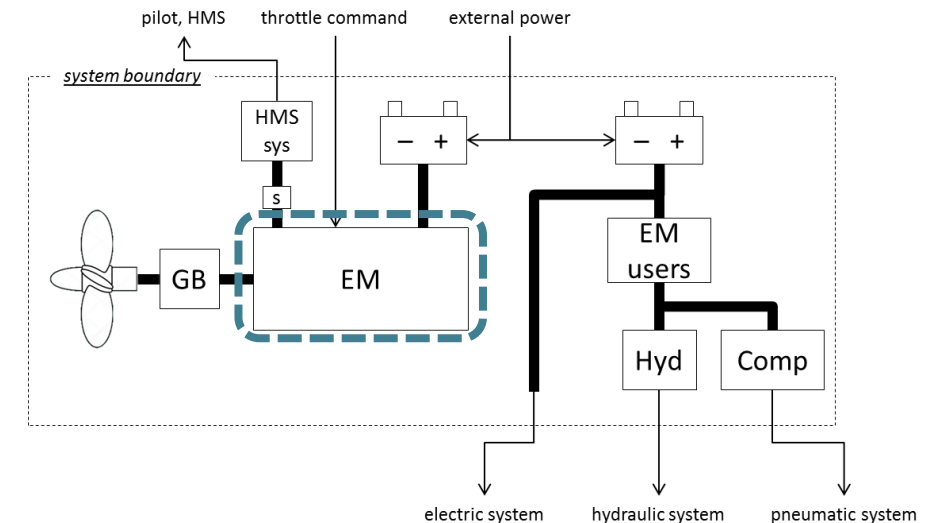
Top Function: “To provide propulsive power”



Conventional architecture



Hybrid-electric architecture



Electric architecture

many more other architectures....

¹ MIT Definition

Function-based System Architecting Process

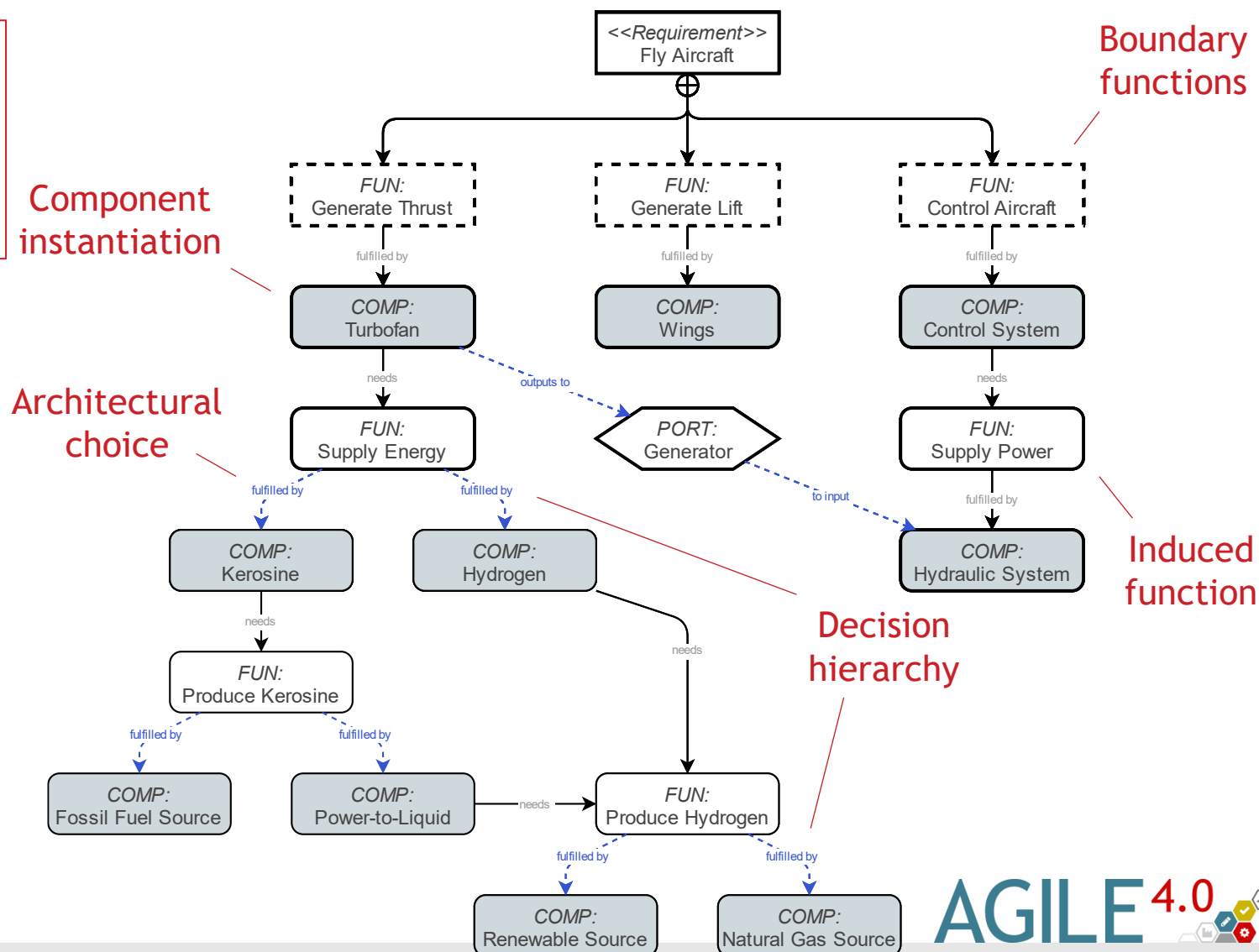
More details later today at 16:15:

Bussemaker et al.: Function-Based Architecture Optimization: An Application to Hybrid-Electric Propulsion Systems
Session 2.3.1 Architecture Analysis

- Collect functional requirements for specific level of elaboration
- Identify boundary functions
- Allocate boundary functions to components
- Identify induced functions and fulfill these
- Model additional architectural choices

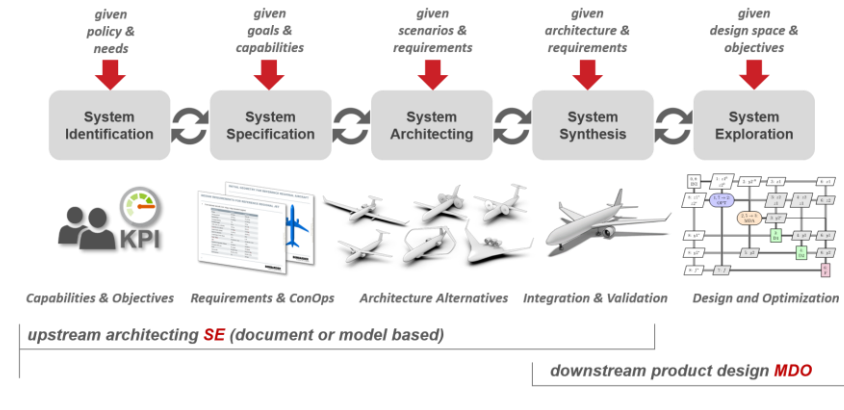
Function-based architecting benefits:

- Natural transition from problem to solution
- Less prone to solution bias
- Directly traceable to requirements

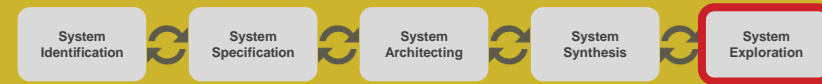


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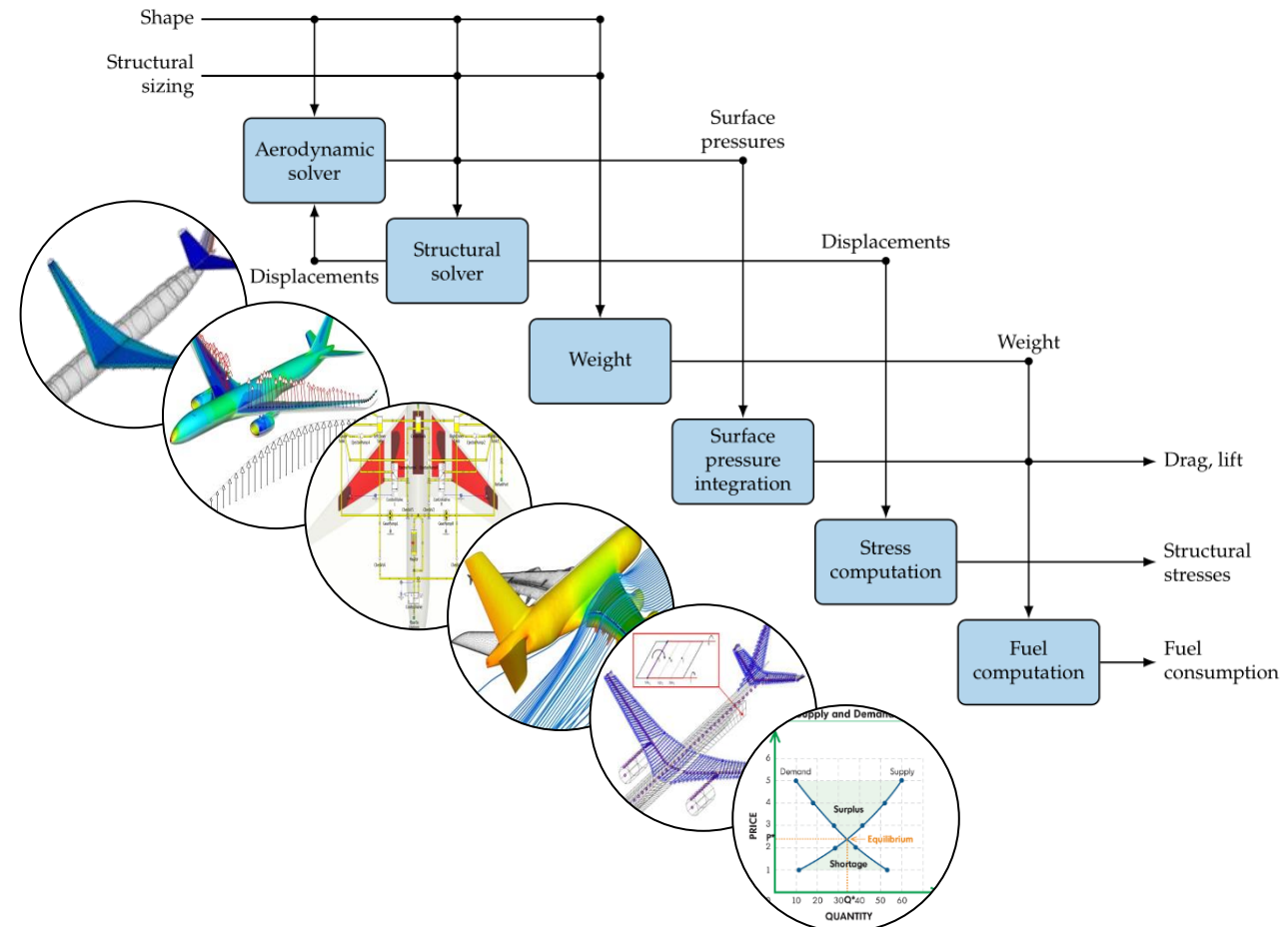
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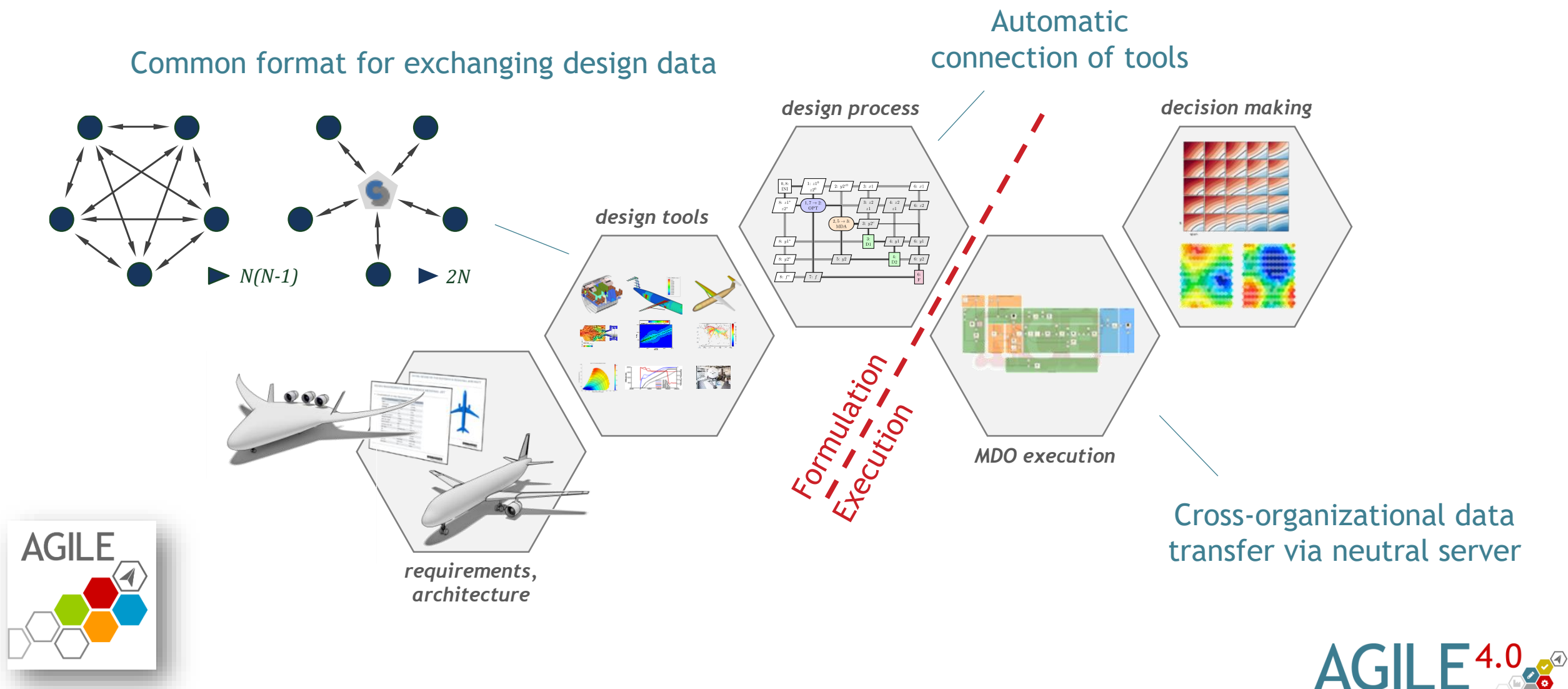
Multidisciplinary Design Analysis & Optimization (MDAO)



- Different system elements *influence performance* in different ways
- Engineering disciplines analyze the system from *different perspectives*
- **MDAO: numerical coupling of disciplinary analysis tools**
- Essential to enable *system-level* analysis
- How to deploy cross-organizational MDAO?



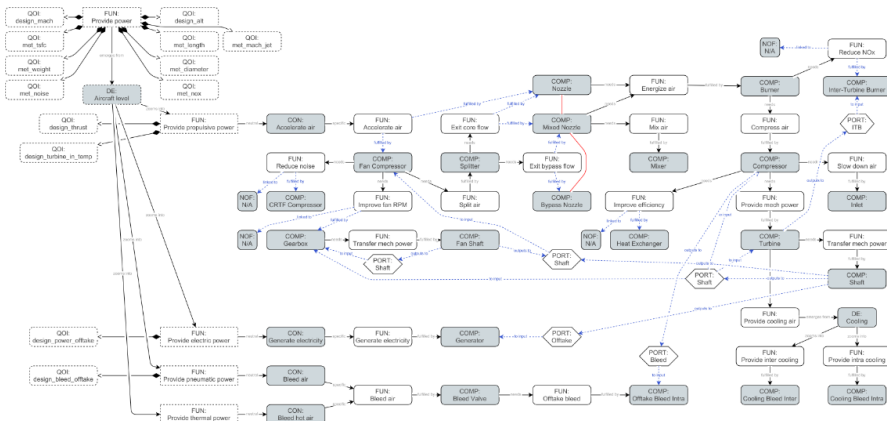
Collaborative MDAO Process



Connecting MBSE and MDAO

MBSE Architecture Model

- Logical description of the SOI
- Described in terms of functions, components, QOIs, etc.
- Link to upstream requirements



Define Design Problem



Sync Architectures

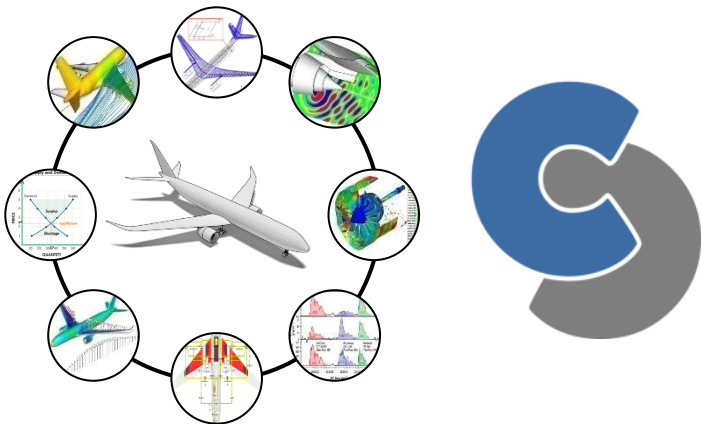


Performance Data

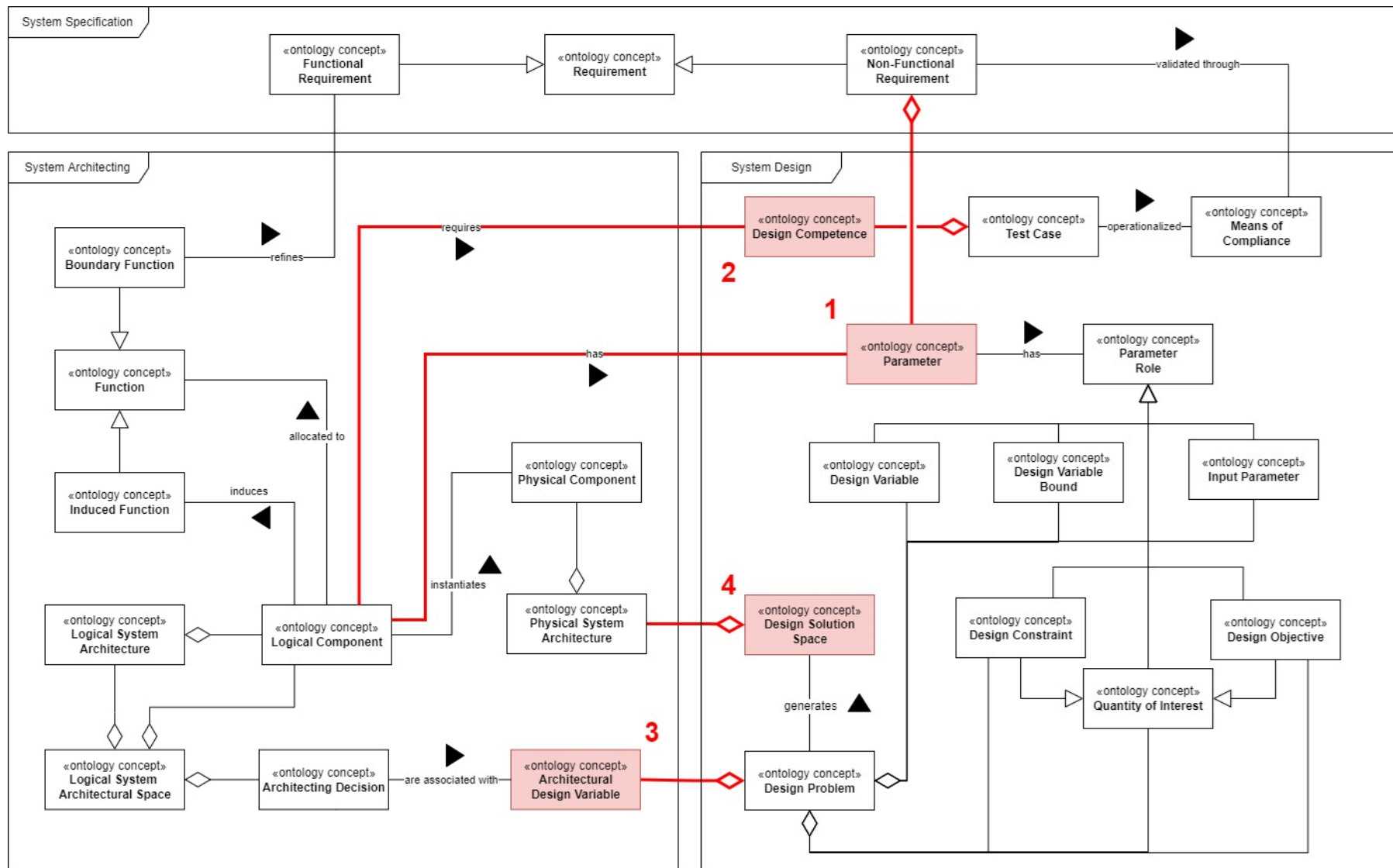


MDAO Product Model

- Physical description of the SOI
- Described using central data schema, in terms of geometry, performance data, etc.
- Connected to disciplinary tools



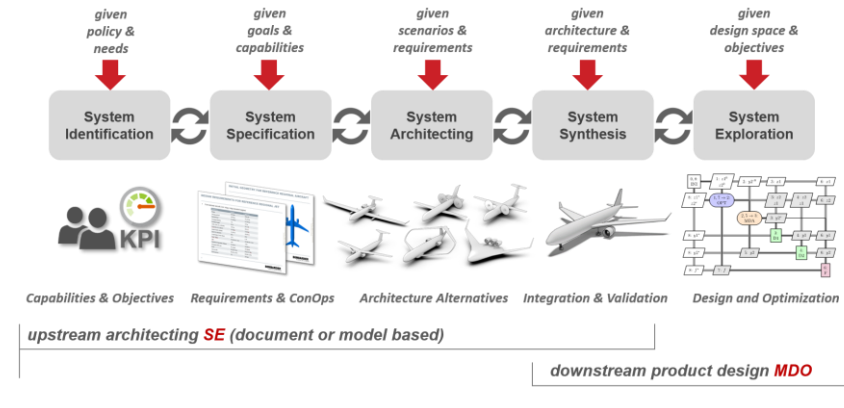
MBSE-MDAO Connection Ontology



- Design parameters are specified from:
 - Architecture components
 - Non-functional requirements
- Design competences (tools) are selected from:
 - The need to evaluate component influence
 - Specification in test cases
- Design variables are additionally defined from architectural choices
- Design solutions represent physical architectures

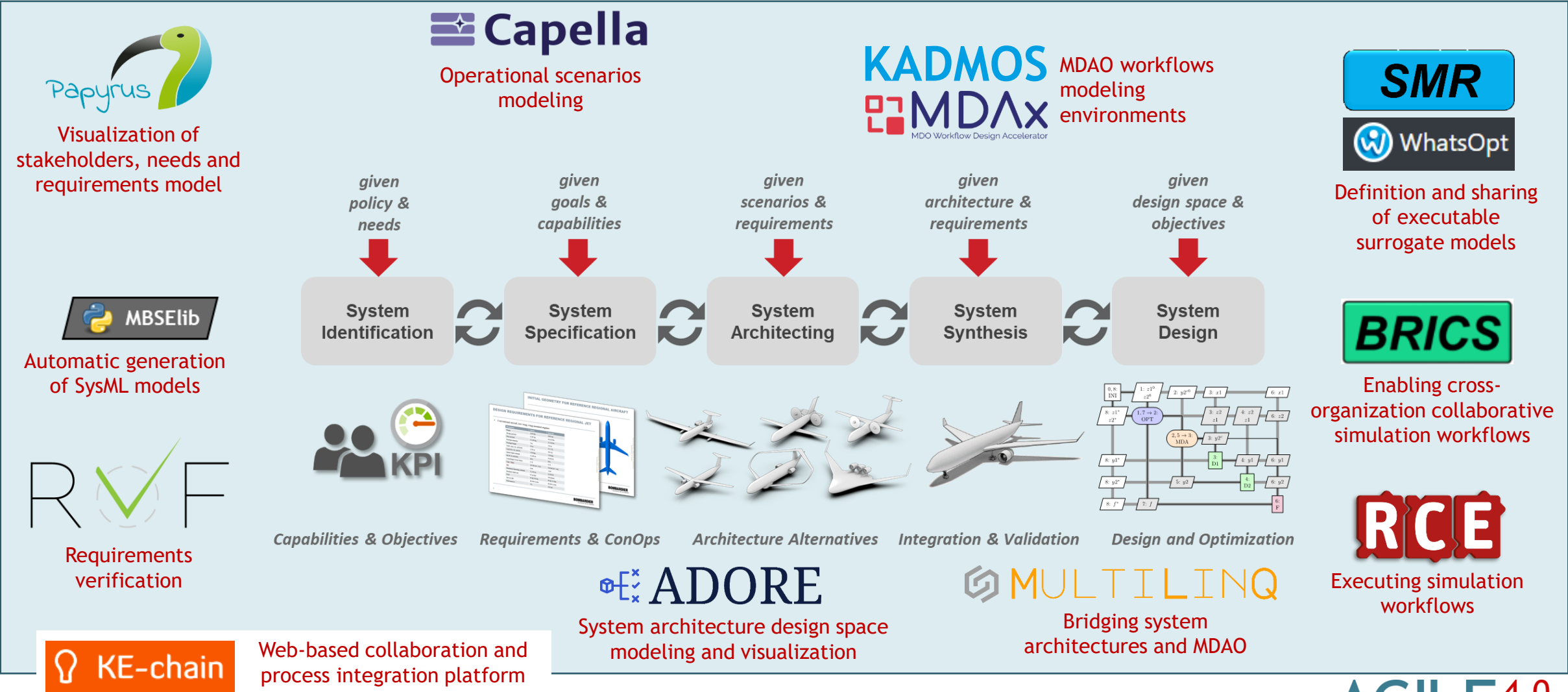
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The Operational Collaborative Environment

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ADORE: Architecture Design Space Editor

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Architecture design space modeler

- Define functions, components, connections
- Identify architectural choices
- Define input parameters and metrics

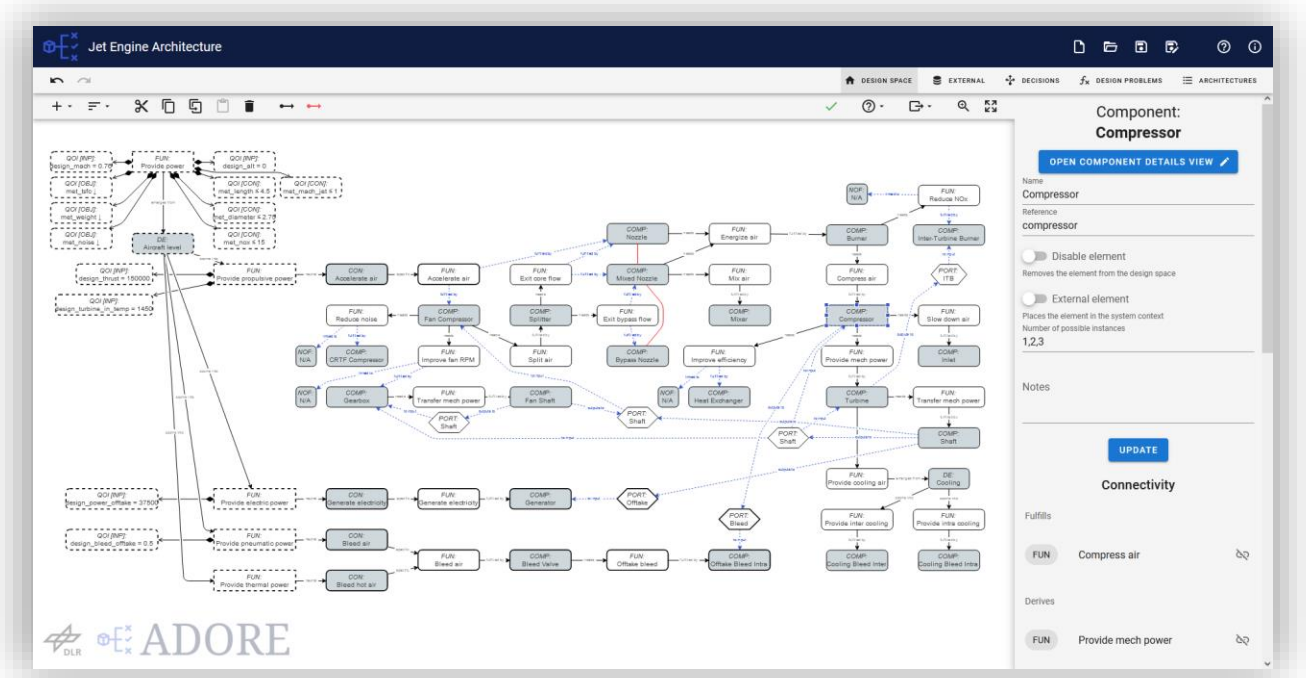
Architecture generator

- Make architectural choices to create architectures

Architecture optimization framework

- Define design variables, objectives, constraints
- Connect to optimization algorithms

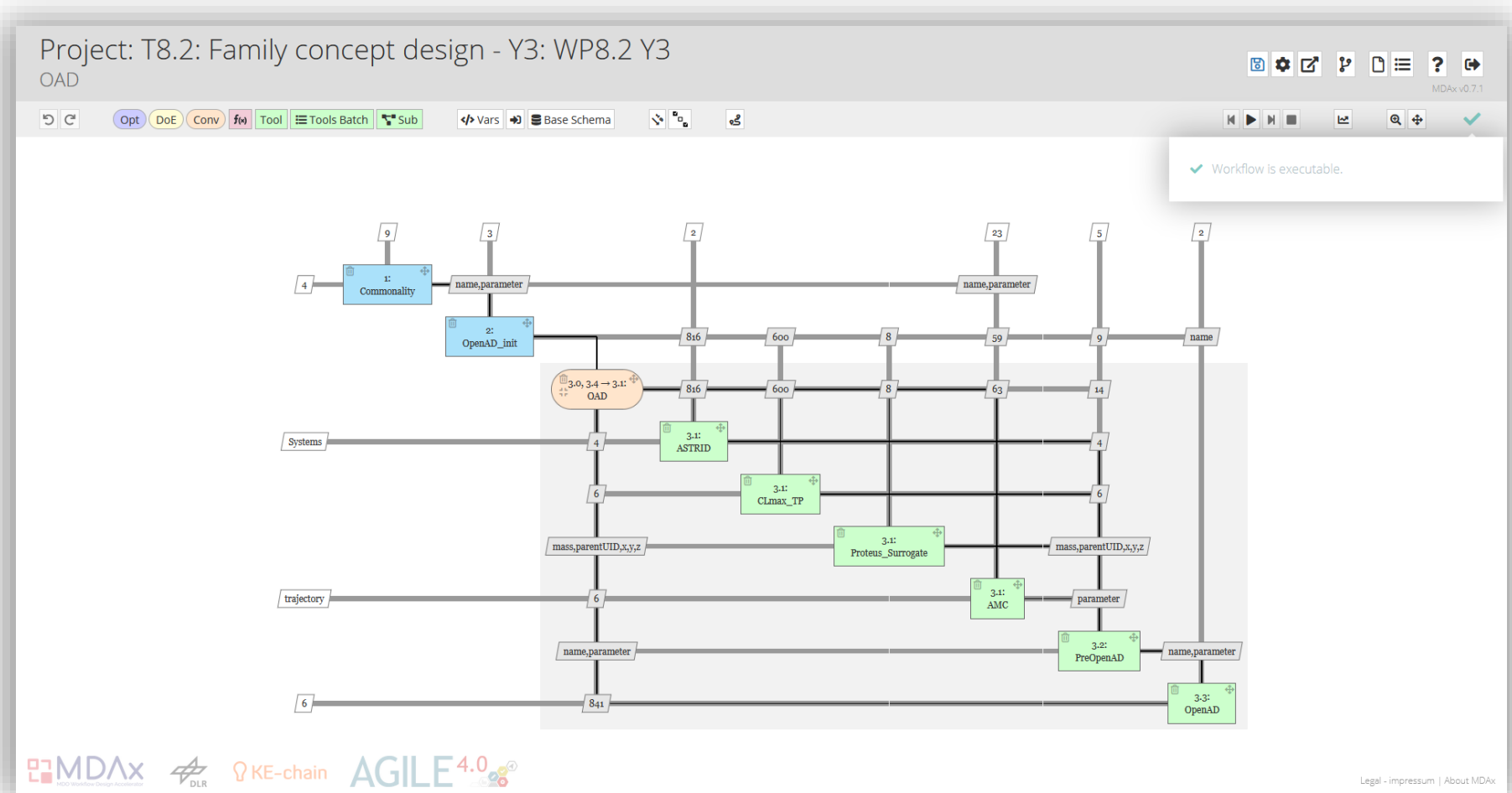
More details later today at 16:15:
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Session 2.3.1 Architecture Analysis



Collaborative MDAO Workflow Modeling

agile4.eu/technologies

- Import disciplinary tool I/O definitions
- Automatically discover data connections
- Interactively add MDAO elements like converger, optimizer, etc.
- One-click export to execution environment



Architecture-to-MDAO with MultiLinQ

1. Map architecture elements to CDS nodes

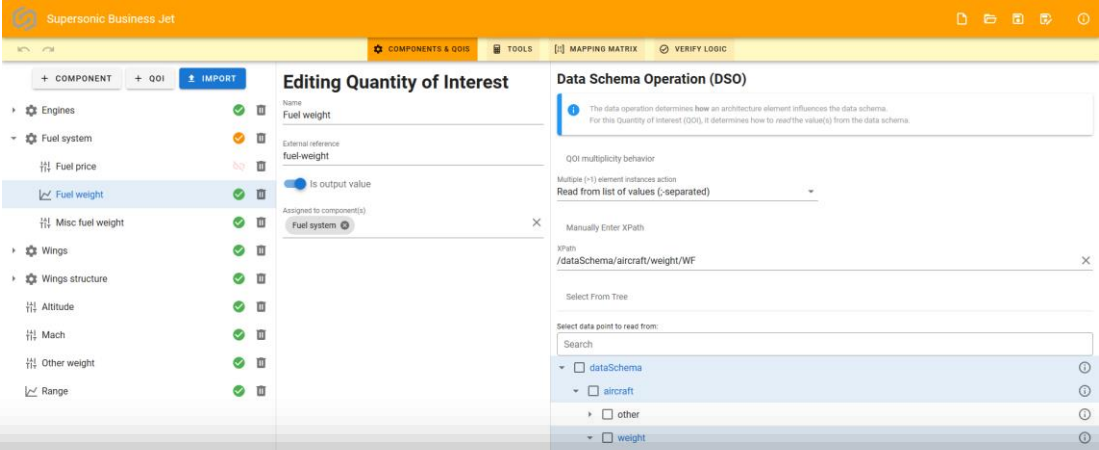


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  </mWingStructure>
</mWingsStructure>
```

QOI:
Wing Weight: 1229.61

2. Support selection of MDAO disciplines

Using the *Component-Tool* matrix



Components		QOIs	Tools												
			02_CPACs_Update	03_CATIA_Update	04_Surface_Mesh	05_Volmesh_Deform	06_Struct_Mdl	08_TAU	10_Nastran	11_HyperSizer	12_Struct_Mdl_Upd	14_Nastran_Deform	15_FlowSimulator	16_Obj	18_FlowSim_TAU
Aircraft			✓		✓	✓	✓		✓				✓	✓	✓
Aircraft	Flight cond						✓							✓	
Wing		✓					✓	✓	✓		✓			✓	
Wing	DV	✓													
Wing	Aero forces						✓								
Wing	Deformation										✓				
Wing	L/D						✓							✓	
Wing	Loads							✓							
Wing	Mass								✓						
Wing	Thickness								✓						
	Objective												✓		✓

Requirements Verification Framework

- Goals:

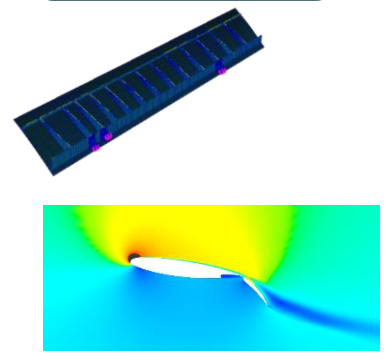
R ✓ F

 - to enable automatic verification
 - to integrate requirements in MDAO

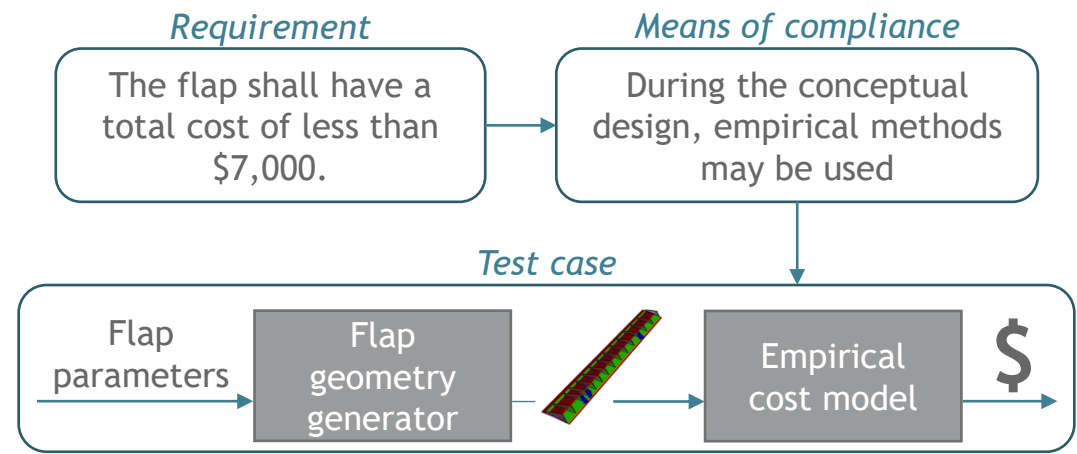
5-step approach:



- Functional
The SYSTEM shall [exhibit] FUNCTION [while in CONDITION]
- Performance
The SYSTEM shall FUNCTION with PERFORMANCE [and TIMING upon EVENT TRIGGER] while in CONDITION
- Design (constraint)
The SYSTEM shall [exhibit] DESIGN CONSTRAINTS [in accordance with PERFORMANCE while in CONDITION]
- Environmental
The SYSTEM shall [exhibit] CHARACTERISTIC during/after exposure to ENVIRONMENT [for EXPOSURE DURATION]
- Suitability
The SYSTEM shall exhibit CHARACTERISTIC with PERFORMANCE while CONDITION [for CONDITION DURATION]

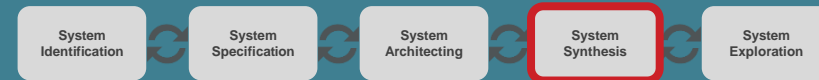


Objective
Constraint
Design variable

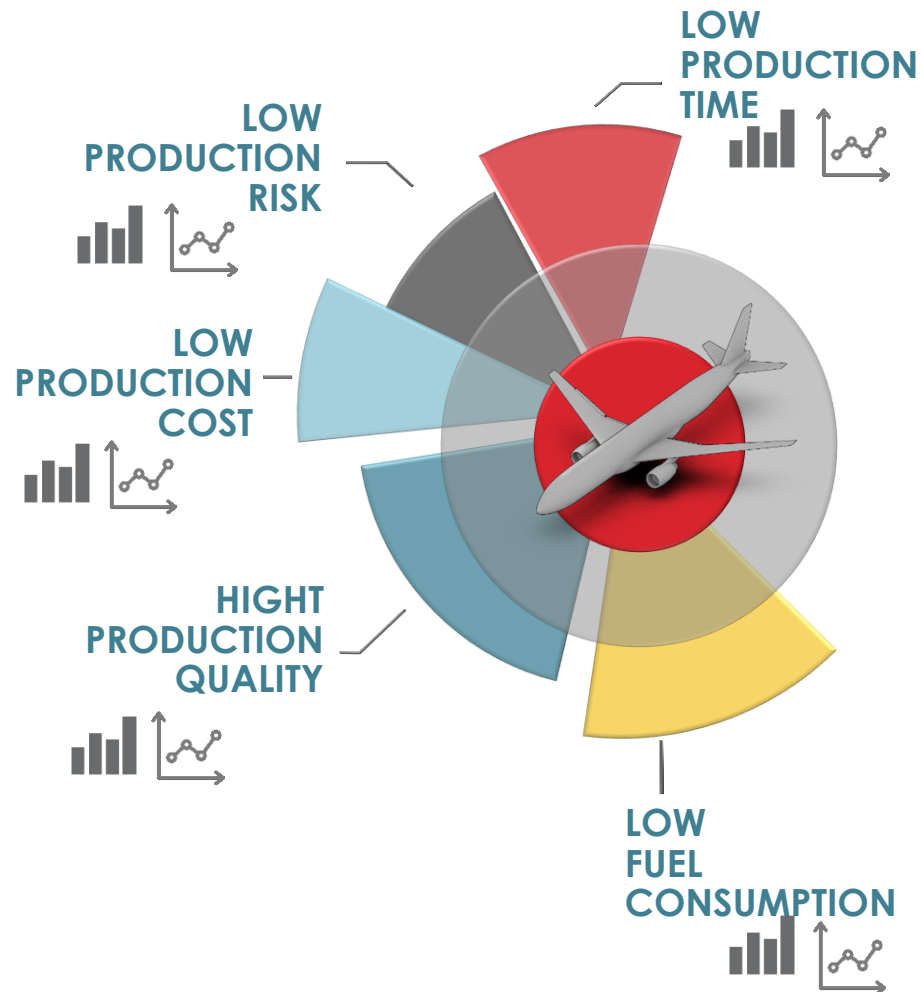


Bruggeman: From Requirements to Product: Digitalization of the Aircraft Design Process
Monday poster sessions

Connect Stakeholder Expectations to MDAO with VALORISE



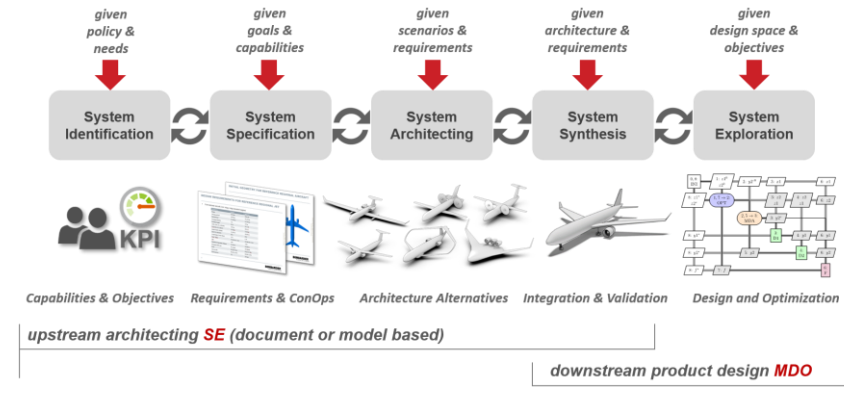
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Tuesday at 13:30:
Donelli et al.: Value-driven SE Approach addressing Manufacturing, Supply-Chain and Aircraft Design in the Decision-Making Process
Session 5.4.1 Supply Chain

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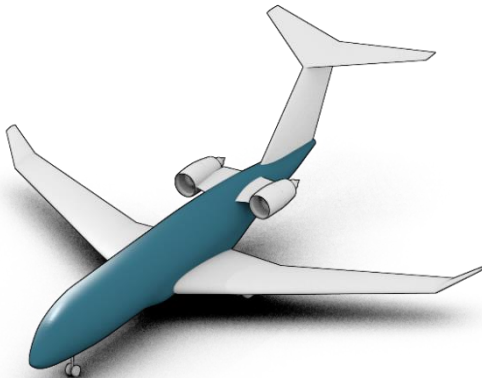
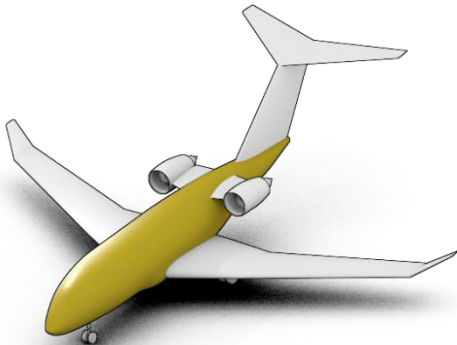
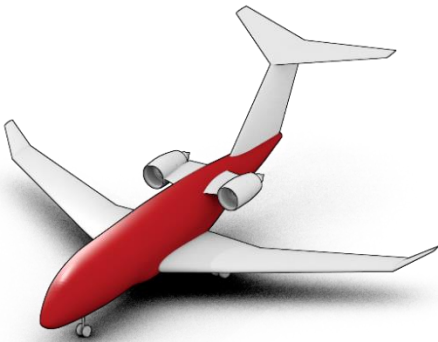
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Business Jet Family Design Case

- Tradeoff between OEM NRC and DOC
- Sharing more components:
 - Decreasing investment costs
 - Increasing operating costs

Parameter	Unit	Aircraft Values		
Long-range cruise speed	M	0.85		
High-speed cruise speed	M	0.90		
Design payload	Pax	8		
Design range	nm	2500	3000	3500
Cabin length	ft	25	25	35
Balanced field length	ft	5000		
Landing field length	ft	2500		
Initial cruise altitude	ft	41000		

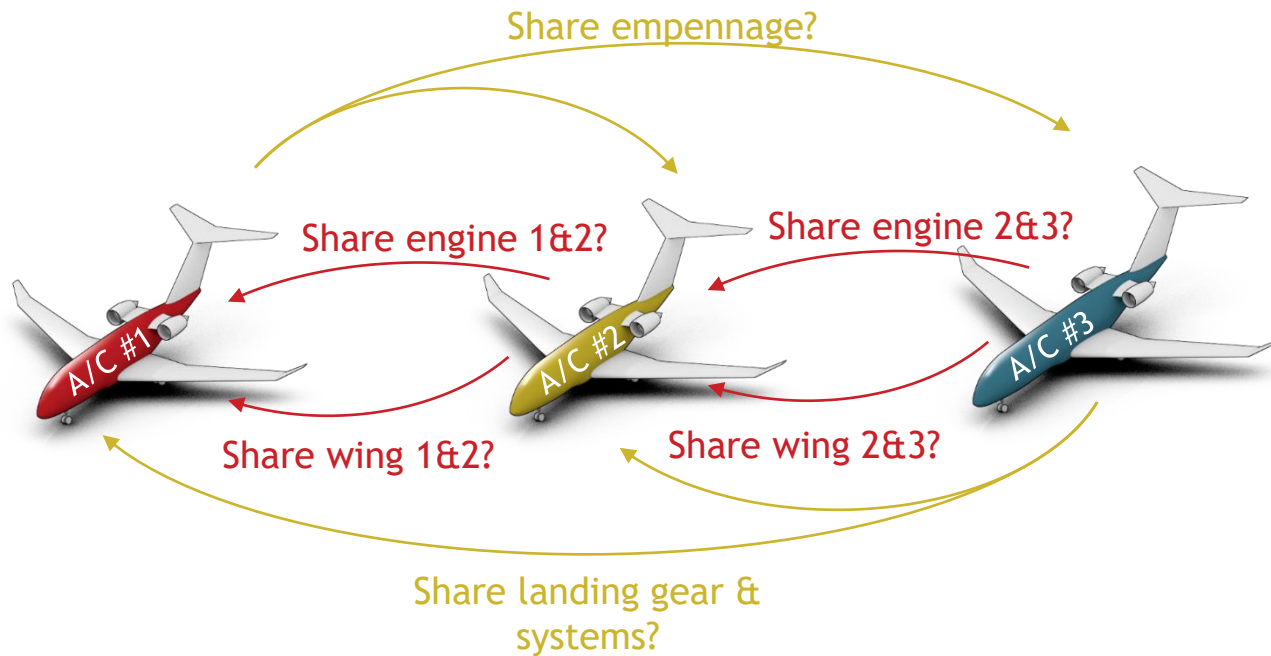


Commonality Sharing Decisions

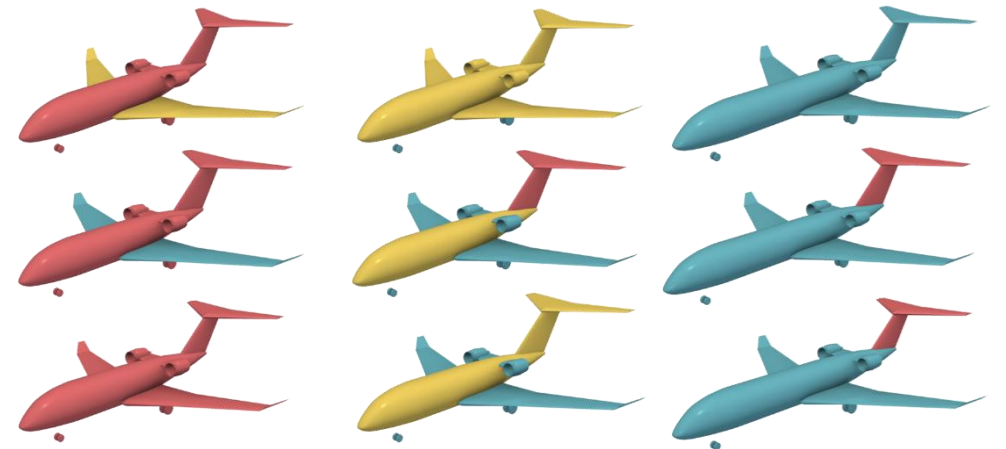
No sharing: high performance



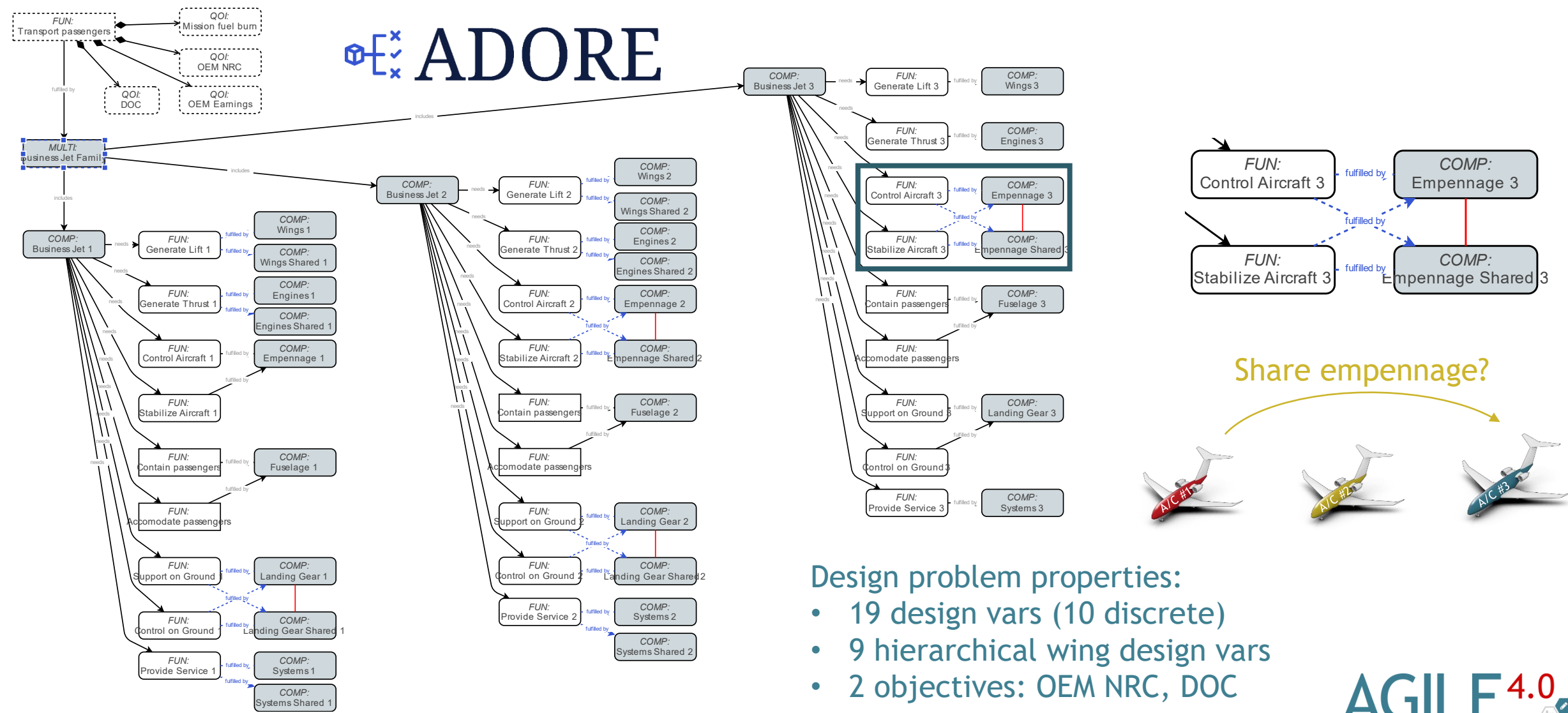
Max sharing: low investment costs



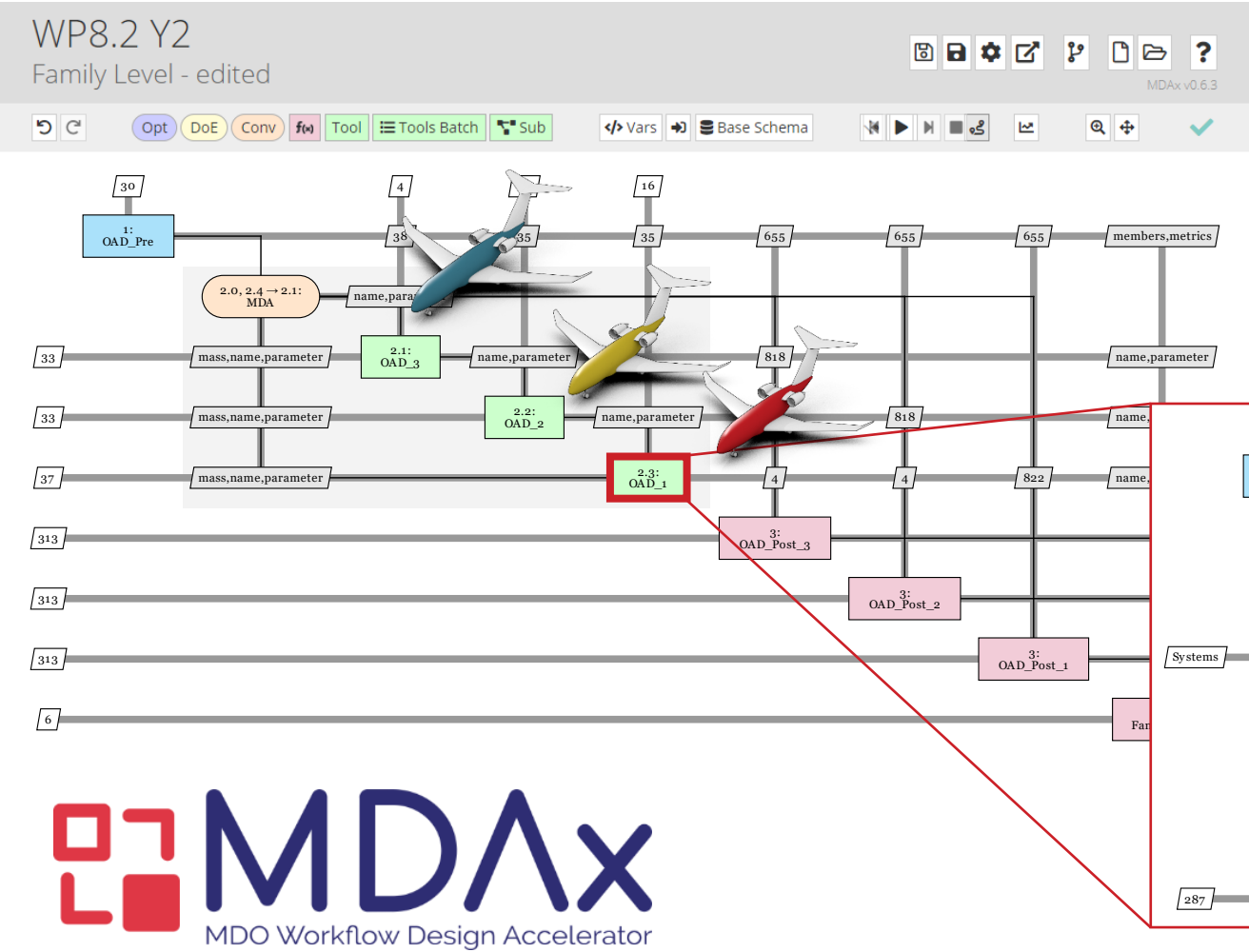
1024 possible family architectures



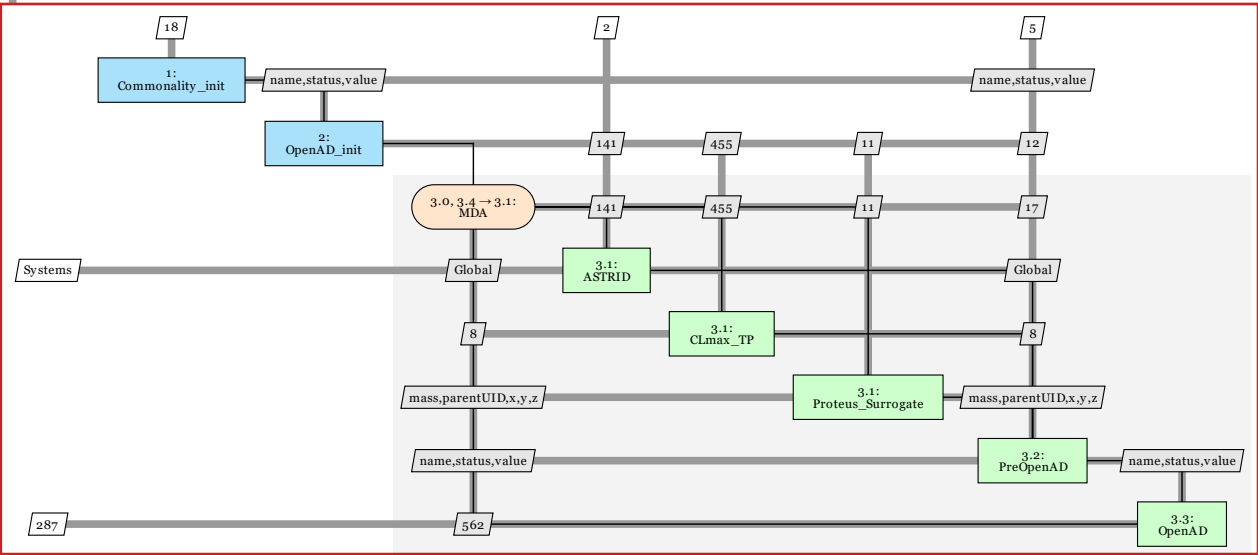
Architecture Design Space Model



Multilevel MDAO Workflow

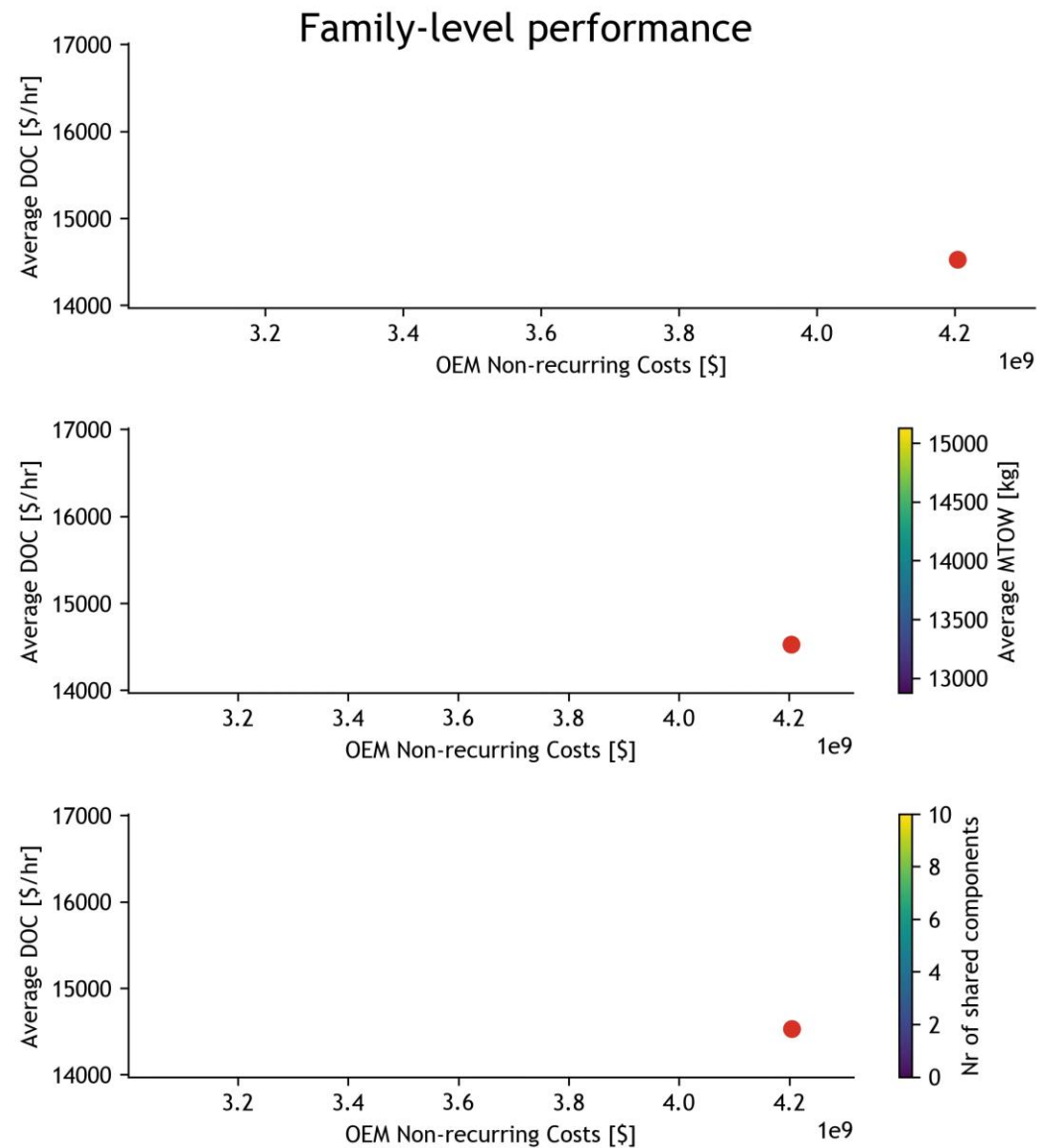


- One family-level workflow to design the aircraft family
- Three aircraft-level workflows to design each aircraft
- Automatically generated from I/O definitions





Business Jet Family Results



Business Jet Family #1 (DOE)

Design variables

	AC 1	AC 2	AC 3
Wing sized for	AC 1	AC 2	AC 3
Engine sized for	AC 1	AC 2	AC 3
Ldg gear sized for	AC 1	AC 2	AC 3
Systems sized for	AC 1	AC 2	AC 3
Empennage sized for	AC 1	AC 2	AC 3
Wing sweep (LE) [deg]	36.0	30.1	36.6
Rear spar position [%]	77.0	81.5	81.6
Thickness ratio [%]	8.5	10.1	7.2

Output metrics

	AC 1	AC 2	AC 3
MTOW [kg]	12000	13060	14940
Block fuel [kg]	3580	4340	5140
DOC [\$/hr]	12700	14680	16200
Wing area [m2]	48.9	48.8	59.6
Rated thrust [kN]	20.6	22.4	25.6
Mean DOC [\$/hr]	14520		
OEM NRC [M\$]	4200		

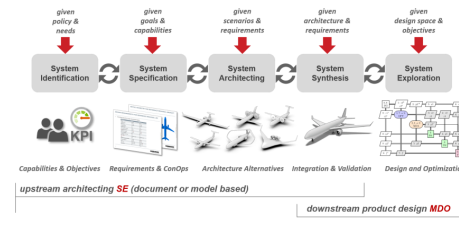
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Publications, technologies, use cases:
www.agile4.eu



Monday poster sessions:

Bruggeman et al.: From Requirements to Product: Digitalization of the Aircraft Design Process

Monday 16:15, session 2.3.1:

Bussemaker et al.: Function-Based Architecture Optimization: An Application to Hybrid-Electric Propulsion Systems

Tuesday 13:30, session 5.4.1:

Donelli et al.: Value-driven SE Approach addressing Manufacturing, Supply-Chain and Aircraft Design in the Decision-Making Process

Thursday 11:30, session 11.5.3:

Merola et al.: Value-driven Optimization Campaign Addressing Manufacturing, Supply Chain and Overall Aircraft Design Domains in the Early Development Stage